COMPANY PROFILE 1915-1960

FAIREY















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FAIREY

Company Profile 1915-1960

'FLYING-BOATS, SEAPLANES, AEROPLANES AND AMPHIBIANS'

Fairev

NE OF THE MANY ingredients required for creating a great aircraft company, is the ability to diversify and this was just one of the strengths that would keep the Fairey Aviation Company in business from 1915 through to 1960. Like so many other aircraft manufacturers of the day, it found its feet by taking on sub-contract work and, while this was being carried out, it began to build on its own aircraft portfolio. Fairey did not just sit back and produce aircraft in line with specifications; they designed new features which would be incorporated in all aircraft in the future.

While many companies folded during the post-First World War period and, following a small order for the Fairey III, the company's future was made secure during the inter-war period. This culminated in the excellent IIIF which was not declared obsolete until 1940!

While aircraft for the Fleet Air Arm would be Fairey's main focus, the company branched out, briefly, into bomber design by introducing the advanced (at the time) Hendon in 1930. The Hendon was the first British cantilever heavy bomber which, for some bizarre reason, initially missed out to the antiquated Heyford biplane bomber and was only, much later, ordered in limited numbers, by which time it had become obsolete.

Fairey would also achieve record breaking success in the 1930s when one of two Long-range Monoplanes built, captured the long distance record in 1933. Post war record breaking would continue with the outstanding, F.D.2 which claimed the world air speed record in March 1956.

However, the company's greatest, and most surprising success story came about, again in the 1930s, when the ubiquitous Swordfish, affectionately known as the 'Stringbag' entered production in 1936. By the beginning of the Second World War, the

Swordfish, on the surface, would appear to be completely obsolete and by this time, its replacement, the Albacore, was already under production. But this basic torpedo bombing biplane proved to be very effective against enemy warships during the early part of the war, achieving great success against the Italian fleet at Taranto and crippling the Bismarck to such an extent, that its demise quickly followed. Later withdrawn from the torpedo role, the Swordfish achieved further success while operating from MAC ships for convoy protection duties and by the end of the war, 21 U-boats had been claimed as sunk. This was a remarkable achievement for an aircraft that was considered by many to be completely unsuited to modern warfare, while its ultimate replacement, the Barracuda, did not achieve the same glittering war record. Post Second World War production saw the naval theme continue with Firefly, which would see action in Korea and finally the Gannet, which continued to serve the Royal Navy well into the 1970s.

The company's venture into rotary wing aircraft would eventually become its undoing, despite huge technical achievements being achieved in a very short space of time. Soaked up by Westlands in 1960, the legacy of the company's final aircraft, the Rotodyne, still lives on today and only now, in the 21st Century, is the world really ready for such a fantastic machine.

Martyn Chorlton, Editor November 2012

What could have been? The sole Fairey Rotodyne, XE521, now referred to as the Westland Rotodyne, pictured at the SBAC on September 7, 1961. Flight via Aeroplane

Acknowledgements
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COVER CAPTIONS The Royal Navy Historic Flight's Swordfish I, W5856; the world's oldest surviving example. *Aeroplane* From left to right; Fairey IIIF, Firefly T.1 and F.D.2. *All Aeroplane*

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FAIREY

The first production Fulmar, N1854 (G-AIBE) pictured during the late 1950s still being flown by Fairey as a company communications 'hack'. The aircraft is preserved at the FAA Museum at Yeovilton. *Aeroplane*

64-65	SEAFOX	96-97	FIREFLY 6
66-67	P.4/34	98-99	FIREFLY TT FAMILY
68-69	ALBACORE	100-101	GANNET
70	FC.1	102-103	FIREFLY 7
71	PRIMER	104-105	F.D.1
72-73	FULMAR I AND II	106-107	FIREFLY U.8 AND U.9
74-75	BARRACUDA I	108-109	GANNET AS.1
76-77	SWORDFISH II, III AND IV	110-111	JET GYRODYNE
78-79	FIREFLY I	112-113	GANNET T.2
80-81	BARRACUDA II	114-115	FD.2
82-83	BARRACUDA III	116	ULTRA-LIGHT HELICOPTER
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THE FAIREY STORY

A COMPANY THAT CAME ABOUT BECAUSE OF THE DEMAND FOR AIRCRAFT DURING THE FIRST WORLD WAR, IT BECAME A VICTIM OF ITS OWN ADVANCED THINKING DURING THE COLD WAR



(above) The prototype Fairey F.D.2, WG774, with Peter Twiss at the controls, captured en route to the 1958 SBAC show at Farnborough. *Aeroplane* (right) Charles Richard Fairey (the tall man!) pictured with friends and his prize winning model lying at his feet

The father of the company

The son of a timber merchant, Charles Richard Fairey was born on May 5, 1887, appropriately at Hendon in Middlesex, although aviation at this iconic location was still many years away. While aged only eleven, Charles lost his father, who had died whilst trying to repay his creditors after, through no fault of his own, his business failed. This not only deeply affected the whole family, but plans to send Charles to the highly respected middle class Merchant Taylor's School at Northwood had to be shelved. Charles spent the next four years in a mainstream school instead but, thanks to a family friend, he left school at 15 to take up employment with the Jandus Electric Company of Holloway who were specialists at producing arc lamps.

Even whilst in full employment, Charles studied four subjects for five nights of the week to City and Guilds standard, in order to qualify as an electrical engineer at the Finsbury Technical College. The Principal of the College was none



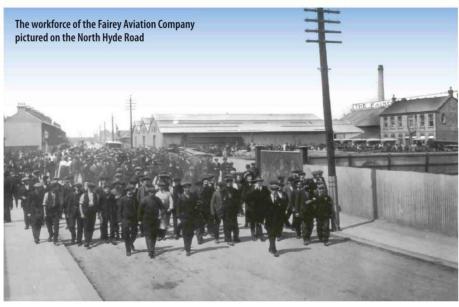
other than Silvanus Phillips Thompson FRS, one of the country's leading electrical engineers whose knowledge and experience would have helped Charles no end. Incredibly, he still managed to find time for one of his secret passions; building flying model aircraft.

While gaining knowledge, Charles was already making a name for himself at Jandus and was given responsibility for installing

electric lights in the docks and warehouses at Heysham in Lancashire. Then, for some unknown reason, the rising star was sacked just a few months later but within no time, he landed himself a reasonably well-paid job as the assistant manager and analytical chemist at Finchley power station. Already armed with a huge amount of knowledge, Charles made up his wages by lecturing at the Finchley Technical

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College and Tottenham Polytechnic.

By 1909, his mother had re-married and, along with his new half-brother, Geoffrey William Hall*, Charles and his family were able to move into a larger house. Charles was now able to spend more time on his model aircraft which were rapidly increasing in sophistication. One of these designs was a canard monoplane with a pair of oppositerotating pusher propellers, which neutralised the torque. He entered the model in several competitions during The Kite and Model

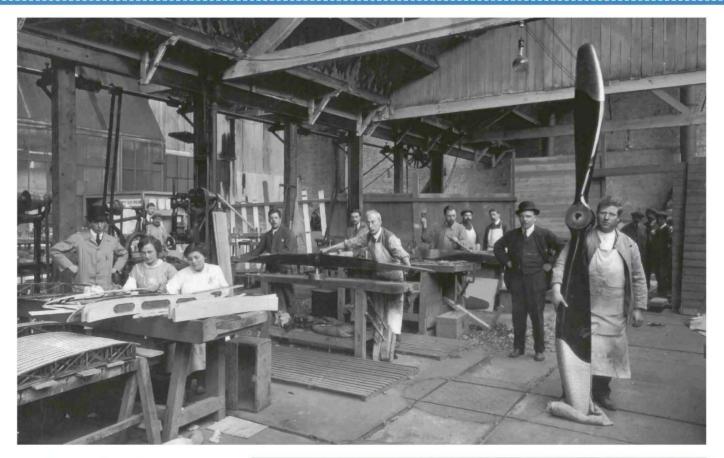
Aeroplane Association event held on June 4, 1910. His canard model not only won the Challenge Cup but also first prize for steering, long-distance flight and stability in flight competitions. The same aircraft also won him a silver cup for the best model at The Aero Models Association on August 13.

These model flying successes were a major turning point in Charles' life and, seizing the moment, he penned a letter to Benjamin Varlars, the manager of the large A W Gamage & Co. department store in Holborn, who invited

(above) The Erecting Shop at the Clayton Road factory in 1917, pictured during the production of 100 Sopwith 1½ Strutters

him to demonstrate the model. The demonstration was carried out in Hyde Park and the subsequent deal earned Charles approximately £300 for the rights to the model plus a royalty on every sale. Charles found a company in Balham to produce the aircraft at a cost price of 5s 6d each. Over 200 models had been delivered to A W Gamage by November 1910, each complete with their own case and instruction book.

Up to this point, Charles' only contact with a full size aircraft was in helping to build a monoplane for Everett Edgcumbe & Co. of Colindale in a nearby field which was destined to become Hendon aerodrome. However, this was soon set to change when, unbeknown to him at the time, his model aircraft was found to have infringed several patents held by a Capt John William Dunne who had been working for many years on the science of automaticallystable sweptwing tailless aircraft. Dunne's company was the Blair Atholl Syndicate which was based at Eastchurch on the Isle of Sheppey and was, at the time owned by Sir Francis McClean and was also the home of the Royal Aero Club. Charles offered to pay Dunne a licence fee for each model produced but instead his only requirement was that the statement 'Licensed under JW Dunne's Patents' should be clearly displayed on every instruction booklet. Charles must have made quite an

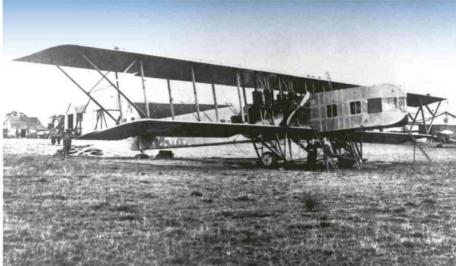


(above) The Propeller Shop at Clayton Road, circa 1917, showing production of fixed pitch, twin-blade propellers from start to finish (right) The incredible Kennedy Giant which was so large, it was assembled out in the open at Northolt

impression on Dunne who offered him a job as an engineer for Blair Atholl.

Charles was now faced with a tough decision. His position at the Finchley Power Station was secure while Aviation during this period of time was not anticipated to have a serious future. His family were against the move but Charles ignored all advice and took up the engineer's position with Dunne in 1911. This period of his life taught him a great deal about aircraft and everyone who knew anything about flying during these early years either passed through Eastchurch or worked there in some capacity. He learned a great deal about stress calculations from Harris Booth for example who was working for the National Physical Laboratory. It was at Eastchurch where Charles also met three men who were destined to work for his own company in the future; Vincent Nicholl, Maurice Wright and F GT Dawson. They would become a test pilot, director and financier respectively for Fairey in the future.

The biggest aircraft operators at Eastchurch were the Short Brothers and Charles was most influenced by Horace Short. By late 1912, Charles was surprised to be invited to join Short Brothers as their Chief Stressman, a position he quickly moved on from to later become works manger and finally chief engineer. Just prior to the beginning of the First World War, Short Brothers moved to their new factory on the



banks of the River Medway at Rochester. This was a move that Charles was not happy about and, along with Nicholl, Wright and Dawson, decided to join the RNAS (Royal Naval Air Service). All but Charles were successful, he also later tried to join the RFC (Royal Flying Corps) as well, the excuses given for rejection were both on medical grounds and the fact that he was too valuable to lose. It was at this point that Charles Richard Fairey, aged 28, decided that it was about time, he established his own aircraft manufacturing company.

*Geoffrey Hall joined Fairey in 1932 and, following service with the experimental and guided weapons department, joined the board as director of engineering in 1949. He went on to become chairman and managing director.

Fairey Aviation Company Limited is born

Following Fairey's failures to join the RNAS and the RFC, he was told by the Director of the Air Department of the Admiralty, Cdr (later Rear Admiral Sir) Murray F Sueter that he would be more usefully employed in designing and building aircraft. Fairey did not hesitate but replied to Sueter's suggestion that he would be happy to do just that, if another aircraft manufacturer would give him a sub-contract. Such was the demand for aircraft at the time, with the full approval of Horace and Oswald Short, a sub-contract for a dozen Short Type 827 seaplanes was issued to Fairey to build.

Fairey now needed financial backing to establish his own company which was duly

THE FAIREY STORY





provided by his friend, F G T Dawson, with further funds promised by Vincent Nicholl and Maurice Wright. A single room was rented at 175 Piccadilly, London as an office and drawing room and the company's first employee, ex-Shorts' Chief Draughtsman, A C Barlow, was taken on. With a capital of £35,000 made up of 25,500 preference shares of £1 each and a further 200,000 deferred shares valued at one shilling each, the Fairey Aviation Company Limited was officially registered in July 1915 with C R Fairey as the only name published as the 'first director'.

The struggle now began to get the company off the ground and Fairey was faced with the challenge of finding a factory to build the 827s, manufacturing equipment, the staff

to build them and finally a seaplane station to re-assemble the machines and test fly from them. All this had to be done in a hurry if Fairey was going to show The Admiralty that he was up to the task and, equally importantly, everything had to be in place before the company's working capital of £15,000 ran out. A suitable building was eventually found in Clayton Road, Hayes, half of which was being used by the Army Motor Lorries Company (AMLC) for reconditioning work. Conveniently, many of the AMLC workers were skilled Belgian refugees and several were made available to Fairey to help build the 827s.

With some time to spare, Fairey began production of its very first aircraft contract as a sub-tenant in the Clayton Road factory and

(above) Fairey IIIBs provide the backdrop for this photograph taken in the wing covering end of the Erecting Shop at Clayton Road (left) An example of a Felixstowe F.3 which was rebuilt by Fairey at Hamble in 1920. The aircraft is pictured at Funchal on March 22, 1921 after its first flight from Madeira to Lisbon. Via Stuart Leslie and Martyn Chorlton

further expansion took place when a flying field was purchased at Harlington, close to the GWR railway line to Reading. A hangar was erected here for 827 assembly; this site was destined to become the main Fairey factory which was erected along the North Hyde Road during 1917 and 1918. With regard to test flying the 827 seaplanes, a site was offered to Fairey by The Admiralty at Hamble Spit where the Hamble River flows into Southampton Water. Workshops were erected at Hamble on four foot high concrete stilts because of the marshy ground and the fact that it flooded. A wooden slipway was also built. The site was managed by Brice G Slater and all of the twelve Short 827s were test flown from here by Australian-born pilot Sydney Pickles.

The contracts come in

By mid-1916, the last Short 827 had been delivered and a further contract to build 100 Sopwith 1½ Strutters for the RNAS had been won. By October, work began at the Clayton Road factory and, to cope, the facilities at Harlington Road were expanded to include a large two-storey building for experimental and development work and to accommodate a much larger drawing office.

Fairey obviously needed more space at Clayton Road which the AMLC were reluctant



(above) Fairey IIIF Mk IIIs under production at Hayes, circa 1926

(right) A close up of one of the more than 600 IIIFs built by Fairey, pictured in the main Erecting Shop at Hayes during the late 1920s

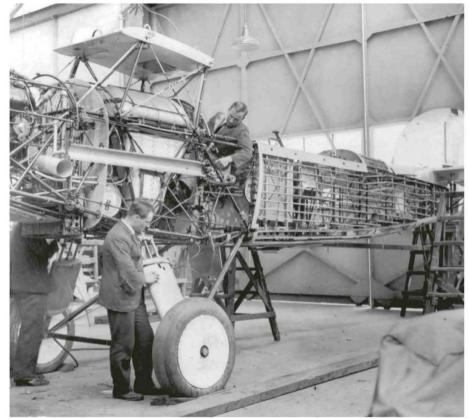
to give up but thanks to the company's solicitor, C O Crisp, control of the factory was prioritised for aircraft production rather than lorry refurbishment.

Within the space of twelve months, the $1\frac{1}{2}$ Strutter contract had been completed. The last aircraft was flown out of a field at Kingsbury, west of Watling Street, in September 1917 as by this time, the Harlington field was too small. Other Strutters are believed to have been flown out of the fledgling Northolt aerodrome as well; an airfield that would witness many Fairey first flights in the future.

Now on a roll, Fairey also received another contract to build 100 Sopwith Camels but like so many other contracts issued towards the end of the First World War, the work was cancelled before a single aircraft was built.

Early designs and the Fairey patented camber gear

In the meantime, Fairey was beginning to find its feet with its own aircraft designs, under the firm hand of the head of the design section, F Duncanson. The first was the F.2 twin-engined



THE FAIREY STORY



fighter and this was followed by the more successful Campania series of seaplanes. Both types were heavily influenced by Charles Fairey's experience of working at Short Brothers. However, Fairey's own unique innovations were incorporated into later aircraft such as the Hamble Baby and the experimental N.9 and N.10 seaplanes.

The most significant invention to come out of the Fairey Aviation Company was the camber gear. This was effectively the first use of a flap fitted to the trailing edge of a wing which could increase the amount of lift available at take-off and landing. The gear was first fitted to the Hamble Baby and continued to feature on Fairey type's right into the 1930s. The camber was the first of many Fairey patented designs which was applied for on May 19, 1916 and then issued the number 132,541.

The Kennedy Giant

Not included in the main aircraft chapters was Fairey's involvement in the ambitious Kennedy Giant bomber designed by C J H Mackenzie-Kennedy. The giant 142ft wing-span multiple-bay biplane was powered by four 200hp Salmson nine-cylinder water-cooled radial engines which proved to be woefully underpowered for the 20,000lb plus aircraft. The aircraft was heavily influenced by aircraft designs of Igor Sikorsky, with whom

Mackenzie-Kennedy had worked with in Russia.

Fairey were involved, to a limited extent, in the design of the bomber but were mainly responsible for the Giant's construction which had to be assembled in the open at Northolt because of its size. Serialled No.2337, the Giant was ready for test flying by Lt Frank T Courtney in late 1917. It was obvious from the start that the bomber would not fly far with the Salmson engines but as long as it left the ground Fairey would be paid for its work. Courtney managed to literally 'hop' the Giant off the ground and the lack of power, combined with a lack of tailplane response, convinced him to pursue no further with flight testing the aircraft. Inferior to the Handley Page O/400, the Kennedy Giant languished in the corner of Northolt aerodrome for many years before being scrapped in the early 1920s.

Expansion

During the production of the Campania, space at the Clayton Road factory became very limited because of the dimensions of the large flying-boat and, as a result, the majority are believed to have gone through final assembly at the Hamble site. As mentioned earlier, the facilities at Harlington were expanded as a result, thanks in part to a loan from the Government of £20,000. Work on the expanding factory began in late 1917 and was



being occupied by the spring of 1918. Part of the expansion included a 90ft wide, 24ft high erecting shop, complete with bay, machine and fitting shops.

The rapid run-down in demand after the First World War hurt many aircraft

FAIREY

(right) An incredible mix of Fairey aircraft types under the same roof including various marks of Firefly, three Spearfish prototypes and Fulmar G-AIBE, the company communications hack in the foreground. *Aeroplane*

manufacturers, including Fairey, which only produced 20 aircraft in 1919. The majority of these were Fairey IIICs, which were ordered for the North Russian Expeditionary Force.

Survival plans to keep the company going were quickly drawn up because of the lack of aircraft work. One of these was to form Fairey and Charles Ltd, a motor-body manufacturer with a capital sum of £50,000, in the summer of 1919. However, the new company was destined to be short-lived thanks to an order for 50 Fairey IIIDs which saw the aviation manufacturer back in business in a healthy way. While it would not be in the most comfortable of positions until the late 1920s, orders for the Fawn, Flycatcher and IIIF, saw the company through the decade.

With this in mind, Fairey managed to build a healthy enough bank balance to enable the company to pursue several private ventures. One of these was the risky plan to licence build the Curtiss D-12 engine, which cost the company £20,000. The result was the excellent Fox light bomber which only served in very limited numbers with the RAF. The Fox's design would make quite an impression on the way British aircraft were designed but would not, in the short term, benefit the company that introduced it. Fairey was also frowned upon for the way it promoted American aviation in Britain and one highranking industrial leader of the day described Charles Fairey as being 'the representative of the Curtiss Company'. As a result, Charles was temporarily disillusioned with the aviation business and, for a period of time, concentrated on his other passion in life which was designing and racing his own yachts, a sport he proved to be very successful in.

Growing in strength

1929 was a big year for Fairey, especially with



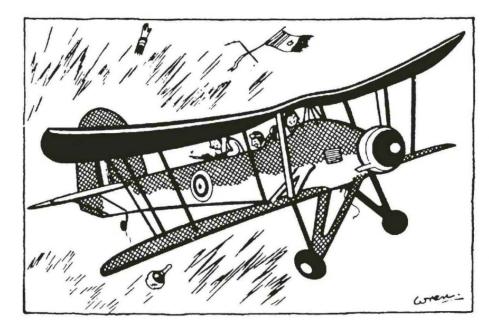
regard to its value which was strong enough to begin approaching outside shareholders. With assets valued at £615,486, the Fairey Aviation Company Limited was registered as a public company on March 5, 1929. The same year, the company was asked not only to vacate their original factory at Clayton Road but also Northolt Aerodrome from which it had been flight testing aircraft since 1917. It was a blessing in disguise for the latter, as Fairey was rapidly outgrowing the hangar space allocated for its use. Northolt's replacement was a 150 acre site at Harmondsworth in Middlesex which was bought for £15,000. From the summer of 1930, Harmondsworth became Fairey's main flight test centre and would remain so until 1944 when the site was requisitioned by the Air

Ministry despite plans to turn the airfield into another large extension of the Hayes factory. While concrete runways and massed expansion was carried out, the military never fully took over, instead the airfield evolved into Heathrow Airport. Fairey were offered Heston as an alternative in early 1945 but this proved unsuitable because of air traffic control issues and instead opted for White Waltham which was taken over in November 1947. Heston continued to serve as the point of departure for newly built aircraft which were fully flight tested at White Waltham. Remarkably, the compensation that Fairey was owed for the Air Ministry's requisition of Harmondsworth was not fully settled until 1964, by which time the company had already moved on from aircraft manufacturing. Harmondsworth's 'billiard table' surface made it ideal for flight testing and the first prototype to make its maiden flight from here was the Hendon on November 25, 1930.

Enter the Swordfish, just!

Only those people who actually worked for Fairey and the many civil servants involved in the specification for the Fairey Swordfish ever knew how close the type came to being scrubbed. The loss of the TSR.I in September 1933, luckily for Fairey, took place eight months before the company's main competitor for the specification, the Blackburn Shark, first flew. The aircraft's replacement, the TSR.II was designed and manufactured in an incredibly short period of time. Cutting every conceivable works procedure, the components were manufactured wherever space could be found within the Hayes factory walls. In the space of

(left) One of Fairey's greatest and possibly, most surprising success stories, the Fairey Swordfish. *Via Aeroplane*



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seven months, the TSR.II prototype, later to be named Swordfish was completed but Fairey was left to wait until April 1935 when the first production order was placed.

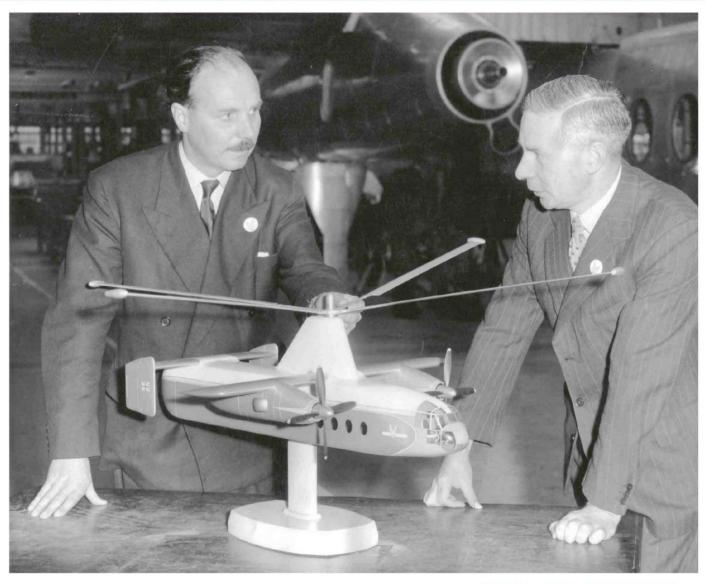
When another large order for the Battle was received it was time to expand again and, in late 1935, Fairey took over the ex-Willys-Overland Crossley car factory at Heaton Chapel in Stockport. Hayes was expanded again on October 1938 when a brand new research department was opened by Air Minister, Sir Kingsley Wood. It was the country's first private venture research centre and the well-equipped facility included a large wind-tunnel which had a test chamber 22ft long and a working section 12ft high and 10ft wide.

(above) The sprawl of the main Fairey factory at Hayes, looking west as it appeared in June 1946. Fairey Surveys via Martyn Chorlton (left) The imposing main headquarters building of the Fairey Aircraft Company Limited at Hayes, during the late 1940s. Via Martyn Chorlton

The vast Fairey 'empire'

By the beginning of the Second World War, Fairey were responsible for 25 factories and workshops of their own and many more which were operated by a host of sub-contractors of varying sizes. By now the Fairey Group was not only producing aircraft but also providing spares, aircraft repair, production and repair of propellers, production of standardised components for the aviation industry as a whole, production of special machinery and experimental work across the board.

New aircraft were built at Hayes and Heaton Chapel which had satellite factories spread across the north-west. Aircraft were assembled and flight tested at Heathrow (ex-Harmondsworth) in the south and from Ringway in the north. The Swordfish, Albacore and Firefly were all built at Hayes, the Swordfish until Blackburn took over in 1940. At Heaton Chapel, the Battle, Fulmar and Barracuda poured out by the hundred while a shadow factory at Errwood Park carried out the sub-contract production of the



(above) Dr G S Hislop (left), Chief Designer of Fairey
Helicopters and Mr R L Lickley, Chief Engineer of Fairey
Aviation with a model of the new Fairey Rotodyne on August
30, 1956. International News Photos via Martyn Chorlton
(below) The Fairey Rotodyne flight test crew after
carrying out a demonstration in June 1958. From left to
right; W R Gellatly, J G P Morton, E J Blackburn and
Blower. Via Martyn Chorlton



Beaufighter and Halifax.

During the war years' Charles Fairey was ordered to work for the British Air Commission in the USA; an appointment he was criticised for accepting despite having no choice at the time. However, in hindsight, Charles' work in the USA was far more important than the future of his own company at the time. The problem with his departure across the pond in August 1940 was that Charles had not left anyone behind with complete control over the company and even up to this point had devoted much of his time working for the Ministry of Aircraft Production (MAP) under the control of Lord Beaverbrook.

With Charles away, Fairey's aircraft production at Hayes inevitably did not run as smoothly as hoped. For example, pending the taking over of Swordfish production by Blackburn and frustratingly slow progress with Albacore production (in 1942 Hayes was brimming with Albacores when it should have been producing Fireflies), the Firefly was delayed by more than a year. It was a similar story at Heaton Chapel, where Barracuda production was delayed until 1942, although this was more due to the demand for the Rolls-Royce engine during the early part of the war.

A healthy post-war order book

Fairey came out of the Second World War in a remarkably healthy condition with regard to aircraft production. Large orders for the Firefly and later the Gannet were already in place, not to mention a considerable amount of sub-contract work at Heaton Chapel.

The company now looked at rotary wing aircraft which, in hindsight, may have been a mistake considering the technical infancy of such machines during the post-Second World War period. Fairey's contribution to rotarywing was very technically progressive, yet ultimately, it was also a failure.

Fairey's solution, to advance the concept, was to combine the best features of the helicopter and the autogyro. This began with the unimaginatively named Gyrodyne in 1946, followed by the Jet Gyrodyne which ultimately led to the amazing Rotodyne. In November 1955, Sir Charles Richard Fairey chaired what was to be his last annual general meeting. He announced that the company had still managed to achieve a profit of £2 million despite the massive development costs that the Gannet, Gyrodyne and F.D.1 had cost Fairey. Despite this, it was obvious that a stark period lay

THE FAIREY STORY



Andy Hay/www.flyingart.co.uk

ahead for Fairey and the aircraft industry as a whole. In April 1957, the issue of the White Paper on Defence confirmed the aircraft industry's fears by declaring that all the future costs of aircraft development should be paid for by the industry itself, rather than the government.

The future rapidly turned bleak as the Gannet came to the end of its production run, the Rotodyne had a doubtful future and hopes for a spin-off development supersonic fighter version of the Fairey F.D.2 came to nothing. Fairey, like so many other companies had pinned its hope on a potential for the Operational Requirement 339 which would become the TSR.2. Even at an early stage, Fairey thought that the most likely candidates to win this work would be English Electric or

Vickers-Armstrongs but a potential merger with either was also in the offing. Another more plausible merger at the time were the rotary-wing divisions of Bristol and Westland which did ultimately take place, but with Westland becoming the controlling company.

On March 31, 1959, Fairey Aviation Company Limited, renamed the Fairey Company Limited, became a holding company for a host of subsidiaries. The same day, Fairey Aviation Ltd was formed to nurture aircraft and manufacturing work across the country. Work being carried out by Fairey was only on the Gannet and Rotodyne by this time while operations at Heaton Chapel, now referred to as the Stockport Aviation Company, continued on limited sub-contracts. The latter was then taken over by Fairey Engineering

which was secured as a subsidiary of the holding company. At the same time, the following were also created; Fairey Hydraulics, Fairey Surveys and several other non-aviation companies within Britain.

By 1960, Fairey had left the aviation industry but, in August 1972, the opportunity arose to acquire the entire share capital of Britten-Norman aircraft based at Bembridge on the Isle of Wight. When the Fairey Aviation Group took over, Britten-Norman had 354 Islanders in the order book. This work would be shared between the Bembridge factory and Fairey SA, in Belgium, who would build the Trislander. Unfortunately, the Fairey Group as a whole hit troubled waters in 1977 and was forced into liquidation, Fairey Britten-Norman was sold to Pilatus in 1978.



F.16, F.17 AND F.22 CAMPANIA



Carrier operations begin

DEVELOPMENT

Destined to be the first Fairey aircraft to be built in quantity, this two-seat patrol seaplane was specifically designed to operate from one particular naval vessel, the purpose-built HM Seaplane Carrier Campania, from which the aircraft would gain its name. The ex-Cunard liner was built in 1893 and purchased by the Admiralty in October 1914. Its conversion to a seaplane carrier with a 120ft (later modified to 200ft) flying deck on an extended foredeck, was completed in April 1915 by Cammel Laird.

DESIGN

An unequal-span two-bay biplane, the Campania, was made of wood with a fabric covering. The wings, with ailerons on fitted to the upper, could be folded for shipboard stowage, both being hinged at the point they met the narrow fuselage. The main floats were pontoontype, attached to the undercarriage cross-bars by four bungee shock-absorbers while wing-tip floats were fitted directly to underside of the outer wings. A larger tail-float was also fitted complete with a water rudder.

200 serials were allocated for Campania production, from which 100 aircraft were ordered and 62 were built. The aircraft was built in three main versions, the main difference being the powerplant.

OPERATIONAL SERVICE

The prototype, F.16, N1000, powered by an Eagle IV engine, made its first flight from Hamble on February 16, 1917 in the hands of Sydney Pickles. Aircraft number two was designated as the F.17 and serialled N1001 and powered by a 275hp Eagle V and was effectively the second prototype. Both these aircraft later served with the RNAS, N1000 carrying out a record breaking flight from the Isle of Grain to Scapa Flow. The pilot, Lt M E A Wright, later Sqn Ldr Maurice Wright AFC was destined to become a director with the Fairey Aircraft Company.

The main variant was the F.22, of which 62 were built, 50 of these (N1000 to N1009 and N2360 to N2399) were built at Fairey's Hayes factory while the remaining twelve (N1840 to N1851) were constructed by Barclay, Curle and Co. based in Clydeside.

Operational service not only involved serving with the Campania but also the light-carrier Nairana and Pegasus. Service with the Nairana included duty as part of the British North Russian Expeditionary Force in 1919. Shore service saw Campanias also operating with 240 and 210 TDS (Training Depot Stations) from Calshot and 241 TDS at Bembridge and Portland.

>> 1915

Fairey Aviation Co. Ltd. formed

LOADED WEIGHT: (F.16) 5,252lb; (F.17) 5,530lb; (F.22) 5,329lb

MAX SPEED (sea level):

CLIMB (to 2,000ft): (F.16)

5 min 20 sec; (F.17) 5 min

SERVICE CEILING: (F.16)

7,300ft; (F.17 & F.22) 6,000ft

(F.16) 6 hr 30 min; (F.17) 5

hr; (F.22) 4 hr 30 min

(F.16 & F.17) 89 mph;

35 sec; (F.22) 7 min

(F.22) 85 mph

ENDURANCE:

>>> FEB 16, 1917

F.16 makes first flight from Hamble

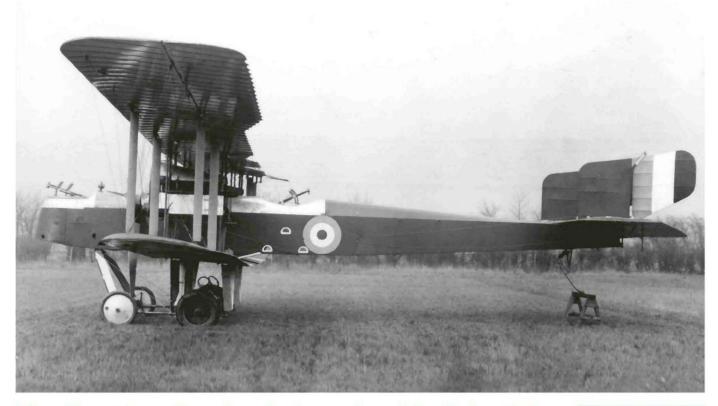
>>> 1919

Campania withdrawn from service

The sole Fairey F.2 was the first aircraft to be designed and built by the fledgling Fairey Company.

The giant 'fighter' is pictured outside the original Harlington 'assembly' shop which would make way for the beginnings of the giant Fairey factory at Hayes.

Via Martyn Chorlton



The first aircraft to be designed and built by Fairey

DEVELOPMENT

The F.2 was not the first aircraft to be built by Fairey, as it had already been carrying out subcontract work building Short Type 827s and was in the process on constructing Sopwith 1½-Strutters as well. Ordered by the Admiralty, the F.2 was a large twin-engine, three-seat land-plane which was intended for use as a long-range fighter, a general-purpose aircraft or even a bomber.

HISTORY & DESIGN

Design work began on the F.2 in November 1915 and several preliminary versions of the aircraft would have also been designed prior to the F.2 appearing in its final form. However, no evidence exists as to how the aircraft evolved. Three aircraft, in two versions, were initially planned and allocated the serial range No.3702 to 3705. The first version was to be tractor powered and the second pusher while two of the aircraft were to be powered by 200hp Brotherhood Ltd engines (believed to be Green engines built under licence). Fairey planned for the three aircraft to be the F.1 (3702), F.2 (3704) and the F.3 (3705), however, neither 3702 or 3705 reached a particularly advanced stage of design.

A three-bay biplane, the F.2, was fitted with a heavy-duty four-wheel undercarriage to make night landings safer and reduce the risk of the aircraft nosing over. The wings could be folded outboard of the engines, the latter being 'handed' or, in other words, opposite rotating so as to counteract torque reaction

and swing. The tailplane had twin fins and rudders.

What is known about the development of the F.2 is that the engines were initially designed in tandem, buried in the fuselage driving outboard mounted propellers via a chain-and-sprocket system. This complex idea was later changed to a more conventional arrangement. The wing-mounted design and the relatively unknown Brotherhood powerplants were changed in favour of a pair of 190hp Rolls-Royce Falcon engines. It was this design that advanced to the prototype stage.

OPERATIONAL SERVICE

The sole prototype was constructed in a wooden shed at Harlington, near the same site that would see the company's complex factory at Hayes, completed just 18 months later. It is believed that the F.2 No.3704 first flew, albeit in very short hops from a nearby field, in the spring of 1917. However, the first official flight was carried out by Sydney Pickles from Northolt, where it was transported by road and then re-assembled on May 17, 1917. Despite the war being far from over, Admiralty interest had by then been lost and the day of the large, slow, multi-seat, multi-engined fighter was over. It is possible that the F.2 could have useful against the Zeppelin and possible deep-penetration operations as well.

PRODUCTION

Only a single aircraft, No. 3704, was built.

TECHNICAL DATA F.2

ENGINE: Two 190hp
Rolls-Royce Falcon
12-cylinder vee liquid-cooled
WING SPAN: 77ft
LENGTH: 40ft 6in
HEIGHT: 13ft 6in
WING AREA (Total):
814 sq ft
LOADED WEIGHT: 4,880lb
MAX SPEED: 93 mph at
sea level
LANDING SPEED: 38 mph
CLIMB: 5,000ft in 6 min
ENDURANCE: 3 hr 30 min

>>> NOV, 1915

Design work begins

MAY 17, 1917First official'

First 'official' flight of the F.2



ENGINE: One 200hp Rolls-Royce Falcon I twelve-cylinder vee liquid-cooled; later One 250hp Sunbeam Maori II twelve-cylinder vee liquid-cooled WING SPAN: 50ft LENGTH: 35ft 6in **HEIGHT:** 13ft WING AREA (Total): 456 sq ft EMPTY WEIGHT: 2,699lb LOADED WEIGHT: 3,812lb MAX SPEED: 90 mph at sea level CLIMB: 2,000ft in 4 min

ENDURANCE: 5 hr 15 min ARMAMENT: One Lewis machine-gun on a Scarff ring for observer/gunner in rear cockpit

10 sec

>>> JUL 5, 1917
First flight of the N.9

MAY 1920
On civilian register as G-EAAJ

>>> JUN 12, 1928 Scrapped after accident

The beginning of a long and successful road

DEVELOPMENT

It was from a pair of experimental seaplanes, unimaginatively referred to as N.9 and N.10 that the long and enduring series of Fairey III aircraft evolved from. Since 1916, there was renewed enthusiasm for aircraft carrying out spotter duties for the Royal Navy especially since the US Navy had, by this time, three cruisers fitted with catapults. This type of launching had already been studied by the Admiralty before the war began but had unwisely been put on hold. Tenders were put out for an aircraft catapult in 1916, capable of launching an aircraft weighing up to 2½ tons, propelling it at 60 mph in a space of no more than 60ft and all without breaching an acceleration of 2.5 g. N.9 would later be tested on a catapult designed and be built by another great aircraft manufacturer, Sir W G Armstrong Whitworth.

DESIGN

The N.9 or F.127, was a single-bay, folding-wing biplane, the upper mainplane being of considerably longer span, almost giving the aircraft a sesquiplane appearance. The aircraft was built to Admiralty specification N.2(a) which called for a two-seat aircraft, specifically for operations from seaplane carriers. It was destined never to be used for such a purpose, spending the bulk of its existence, performing experimental work with a prototype catapult.

Power was provided by a 200hp Falcon I engine, giving N.9 a maximum sea-level performance of 90mph. N.9 was also fitted with camber-changing gear and the flaps along the entire length of the lower wing and between the centre-section of the upper wing as far as the ailerons.

OPERATIONAL

N.9 first flew on July 5, 1917 and, as mentioned earlier, spent a great deal of time being using on catapult trials. Much of this was with the appropriately known catapult vessel HMS *Slinger* which joined N.9 at the MEAD (Marine Experimental Aircraft Depot) at Grain for test from June 1918. The trials were under the command of Lt Col H R Busteed who also carried the majority of flying in N.9. Launches were carried out from HMS *Slinger* both at anchor and under steam, and were the first of their kind to be carried out in Britain using a seaplane. Earlier trials with another catapult were carried out at Hendon using a landplane. Despite the advances being made, it would not be until October 1925 that the first service aircraft was launched from a Royal Navy warship's catapult.

Purchased back from the Admiralty by Fairey in 1919, N.9 was modified with a 250hp Maori II engine and equal span wings. It is believed that N.9 was being prepared to be an entrant for the race to carry out the first non-stop crossing of the Atlantic, a challenge for which the *Daily Mail* was offering a prize of £10,000.

Instead, N.9 became one of the earliest aircraft to be civil-registered as K-103 (the future 'G' prefix came later, all civilian aircraft were initially registered as 'K', beginning with K-100) and later G-EAAJ. By May 1920, N.9 had been sold to the Norwegian Navy where it served until purchased by Bjorne Neilson and registered as N-20 in 1927. The long serving aircraft suffered an irreparable accident on June 12, 1928 and was scrapped in February the following year.



The historically important progenitor of a series

DEVELOPMENT

The second of two aircraft designed to meet Admiralty specification N.2(a) was, like the N.9, only referred to by its serial N.10 and/or by the Fairey construction number, F.128. This aircraft was an alternative carrier-based seaplane designed by F Duncanson which would later be designated as the Fairey III.

DESIGN

N.10 was an equal-span two-bay biplane of slightly bigger proportions and weight to the N.9. It was fitted with folding wings and a more powerful 260hp Sunbeam Maori engine. The fuselage was the same as N.9, the only difference being a larger fin while, like its stablemate, a full-span variable-camber gear was also fitted to the lower wing and ailerons fitted to the upper. Cooling, like N.9, was via radiators mounted either side of the engine but when N.10 was later converted to a landplane configuration, a radiator was mounted in front of the engine.

SERVICE

N.10 was first flown on September 14, 1917 by Lt Cdr Vincent Nicholl, DSO, DFC from the Isle of Grain after being delivered there on August 31. Following a host of different trials and military modifications, N.10, just like N.9, was bought back from the Admiralty by Fairey.

N.10 acquired the civil registration, G-EALQ, in May 1919 and, in September, made its first public appearance at Bournemouth for the Schneider Trophy air race. Now referred to as the Fairey III, the aircraft had changed a great deal, appearing as a single-seat single-bay with its span reduced to 28ft and engine replaced by a 450hp Napier Lion. While the Fairey III did not win the trophy, it was by far the most robust entrant and the only aircraft to return to its moorings under its own power at the end of the race.

G-EALQ also appeared at an Air Ministry commercial aircraft competition held at Felixstowe and Martlesham Heath during August and September 1920. The Fairey III was the only float-plane to enter and for the competition was returned to its original equal-span two-biplane design, but with a two-seat, side-by-side passenger cockpit aft of the pilot. The floats, ingeniously, were combined with a retractable, wheeled undercarriage making the aircraft a useful amphibian. A total of £16,000 was for the taking at the competition which was won by the Vickers Viking III, with the Supermarine Seagull second and the Fairey III winning £2,000 for third.

G-EALQ was used by Fairey as a communications aircraft until late 1922, mainly operating from Hamble.

TECHNICAL DATA N.10/III

ENGINE: (N.10) One 260hp Sunbeam Maori II twelve-cylinder vee liquid-cooled; (III) Once 450hp Napier Lion twelve-cylinder liquid-cooled WING SPAN: (N.10) 46ft 2in; (III) 28ft LENGTH: 36ft HEIGHT: 11ft 10in WING AREA (Total): 542 sq ft EMPTY WEIGHT: (Seaplane) 2,970lb; (Amphibian) 3,771lb LOADED WEIGHT: 4,159lb MAX SPEED: (Seaplane) 104 mph at sea level; (Amphibian) 118 mph at sea level LANDING SPEED: (Amphibian) 54 mph CLIMB: 2,000ft in 3 min 45 sec SERVICE CEILING: 14,000ft ENDURANCE: 4 hr 30 min

» SEP 14, 1917

First flight of the N.10

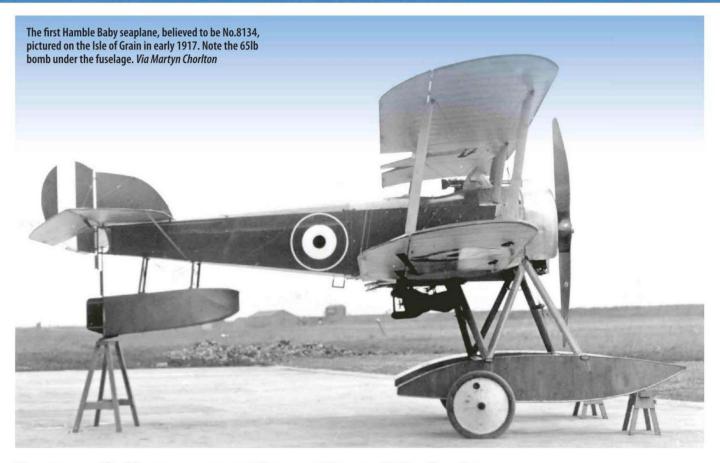
» MAY 1919

Registered as G-EALQ

>> 1922

N.10 retired

HAMBLE BABY



Successfully resurrecting a Sopwith design

- >>> MAY 19, 1916
 Patent applied for camber gear
- >>> DEC 19, 1916 Full specification of camber gear
- >>> EARLY 1917
 First flight by
 Maurice Wright
- >>> JUN 7, 1918
 Enters service with
 253 Squadron
- >>> DEC 1918

 Hamble Baby retired from RAF service

DEVELOPMENT

The Hamble Baby, from a design perspective, was one of the most important aircraft built by Fairey prior to the arrival of the Fox, but the aircraft originated from another manufacturer. The aircraft was a redesigned version of the Sopwith Baby, which had its roots in the 1914 Schneider Trophy winning Tabloid.

Fairey had built several Sopwith Baby floatplanes but one particular aircraft, No.8134, arrived at the Hamble works on October 23, 1916 for repair. Fairey not only repaired the Baby but also took the opportunity to install several modifications, the most obvious being the Fairey Camber Changing Gear. The gear, which was patented by Fairey in late 1916, would be incorporated into the majority of future designs, right up to the arrival of the Swordfish.

DESIGN

The camber gear which set the Hamble Baby apart from the original was a system of half span trailing edge flaps which could operate in unison to increase lift or differentially to control roll or lateral movement. Prior to the introduction of the gear, the early Tabloid aircraft had a tendency to suffer float failure during take-off because of the weight demands being placed upon it. The increased lift achieved by the camber helped the Baby and the combination of the Clerget engine gave the aircraft the capability to carry an offensive war load of two 65lb bombs and a single synchronised Lewis machine-gun.

Originally designed as a floatplane, a large proportion of all Babies built were landplane variants which were designated as Hamble Baby Converts and fitted with a wide-track wheeled undercarriage.

OPERATIONAL SERVICE

The Hamble Baby seaplane did not enter service until May 1918, initially joining 403 (Seaplane) Flight at Seaton Carew, 406 (Seaplane) Flight at Westgate, 412 (Seaplane) Flight at Bembridge, 450 (Baby Seaplane) Flight at Dundee, 359 (Flying Boat) Flt at Otranto and 441 (Seaplane) Flight at St Maria Di Leuca. All but 403 Flight, which only used the Hamble until August 1918, were incorporated into new RAF squadrons. 253 Squadron, formed on June 7, 1918 at Bembridge, was the first RAF squadron to receive the type, followed by 219 Squadron at Westgate in July, 229 and 249 Squadrons in August and finally, 263 Squadron at Otranto which had incorporate both 359 and 441 Flights.

The Hamble Baby floatplane's service career was short and, by the end of 1918, they had been removed from both flight and squadron service. The Hamble Baby Convert served with 481 and 483 (Fighter) Flights at Andrano and also 225 Squadron which was formed at Alimini on April 1, 1918 not to mention several RNAS training units prior to this.

Hamble Babies were generally employed on antisubmarine and attack duties and, while being more common in British waters, the type also served across the Mediterranean, notably from the seaplane carrier, HMS *Empress*, from which two of them attacked Turkish installations in Palestine.

PRODUCTION

Priced at £2,000 each (not including armament and instruments), 180 Hamble Babies were built, of which only 50 (N1320-N1339 & N1450-N1479) were built by Fairey. 130 Babies (N1190-N1219 & N1960-N2059) were built by Parnall of which 74 (N1986-N2059) of them were built as Hamble Bay Converts.



TECHNICAL DATA - HAMBLE BABY

ENGINE: One 110hp (possibly only fitted to the first ten produced by Fairey) or 130hp Clerget rotary

WING SPAN: 27ft 9in

LENGTH: 23ft 4in

HEIGHT: 9ft 6in

WING AREA (Total):

302 sq ft

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MAX SPEED:

90 mph at 2,000ft

EMPTY WEIGHT:

LOADED WEIGHT:

CLIMB: 2,000ft in 5 min 30 sec

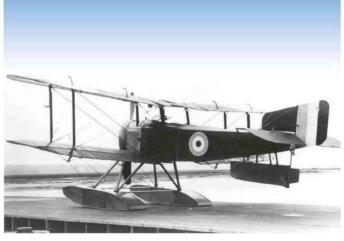
SERVICE CEILING:

7,600ft

1,386lb

1,946lb

ENDURANCE: 2 hr



Fairey-built Hamble Baby, N1457, construction number F. 157, pictured during flight testing in early 1917. *Via Martyn Chorlton*



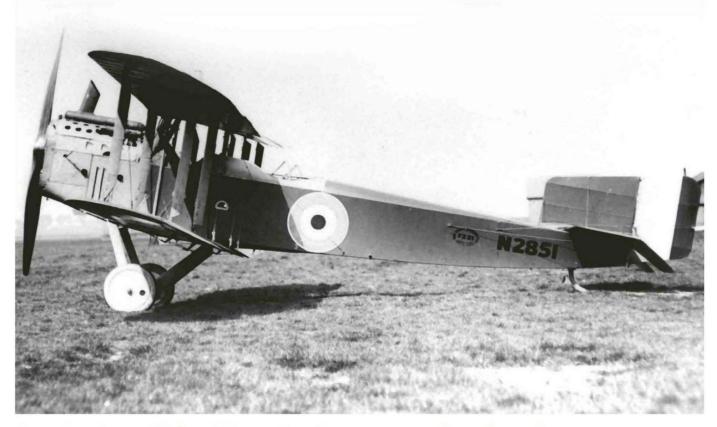
N1452 displays c/n 'F.152' on the side of the fuselage and Fairey Aviation Co. Ltd name emblazoned on the large rear float. Note also a synchronised Lewis machine-gun for the pilot, breaking with the tradition of using a Vickers in this position. Via Martyn Chorlton



74 Hamble Baby Converts were built by Parnall & Sons of Bristol. The majority of them were used by the RNAS for training purposes although some saw service with the RAF and 225 Squadron. Unusually, this aircraft is minus skids. *Via Martyn Chorlton*

IIIA AND IIIB

The second production Fairey IIIA, N2851 (c/n F.221). Via Martyn Chorlton



Beginning of the III-series journey on land and at sea

- >>> JUN 6, 1918
 First flight of IIIA
- » AUG 8, 1918
 First flight of IIIB
- >>> OCT 1918
 IIIB enters RAF
 service
- >>> NOV 1918
 IIIA enters RAF
 service
- >>> 1919
 IIIA declared obsolete
- >>> FEB 7, 1920
 IIIB retired
 from RAF

DEVELOPMENT

By late 1917, the N.10, which was actually the prototype of the III series of aircraft, was converted into a landplane with a traditional V-strut undercarriage and designated as the IIIA. An order for 50 IIIA two-seat bombers for the RNAS saw the aircraft enter production only months before the service was merged with the RFC to become the RAF.

DESIGN

Using the same Maori II engine as the N.10, the IIIA was a better performer, mainly because of the lack of the drag inducing floats. Armament was a single Lewis machine gun on a Scarff ring for the observer, while bombs could be carried on racks under the fuselage. The first IIIA, N2850, made its maiden flight in the hands of Lt Col G L P Henderson from Northolt on June 6, 1918.

The IIIB was designed for bombing duties in response to Admiralty N.2(b) requirements and while the fuselage was the same as the IIIA, the wing, fin and rudder were of a greater area and the floats were larger than the N.10s. The wings folded and the upper had a visible over-hang where the ailerons were attached. These extended wingtips were braced by large king-posts directly above the inter-plane struts. The aircraft also employed the same camber changing gear as the Hamble Baby.

Armament was the same as the IIIA although a bomb load of up to 600lb could be carried in tubular containers under the fuselage. Vincent Nicholl made the first flight of IIIB, N2246, from Hamble on August 8, 1918.

OPERATIONAL SERVICE

Intended as a replacement for the Sopwith 1½ Strutter, by the time the first IIIAs were due to enter service, the First World War had come to an end and, by 1919, the type had already been declared obsolete. The IIIA did join 258 Squadron at Luce Bay and 272 Squadron at Machrihanish in November 1918 but, by March 1919, both units had been disbanded.

With regard to IIIB, only 25 were built, because Fairey had the foresight to convert those on order into IIICs. The IIIB only served with two RAF units, namely 219 Squadron at Westgate and 230 Squadron at Felixstowe from October 1918; the latter until March 1919. The IIIB saw out its days with 219 Squadron until the unit was disbanded on February 7, 1920.

PRODUCTION

50 IIIAs were built, from N2850 to N2899 while 60 IIIBs were ordered but this was reduced to just 25 in the serial range N2230 to N2254.



TECHNICAL DATA - HAMBLE BABY

ENGINE: One 260hp Sunbeam Maori II twelvecylinder vee liquid cooled

WING SPAN: (IIIA) 46ft 2in; (IIIB) 62ft 9in

LENGTH: (IIIA) 31ft; (IIIB) 37ft 1in

HEIGHT: (IIIA) 10ft 8in; (IIIB) 14ft

WING AREA (Total): (IIIA) 542 sq ft; (IIIB) 616 sq ft

EMPTY WEIGHT: (IIIA) 2,532lb; (IIIB) 3,258lb

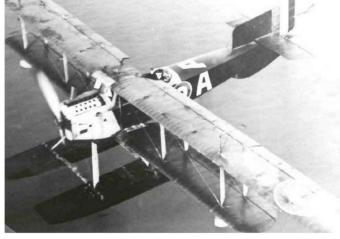
LOADED WEIGHT: (IIIA) 3,694lb; (IIIB) 4,892lb

MAX SPEED: (IIIA) 109 mph at sea level; (IIIB) 95 mph at 2,000ft

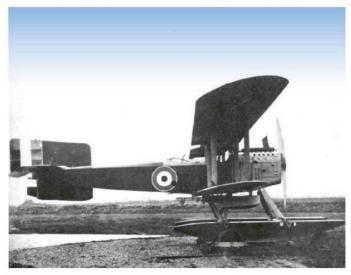
CLIMB: (IIIA) 5,000ft in 7 min 5 sec; (IIIB) 2,000ft in 4 min 10 sec

SERVICE CEILING: (IIIA) 15,000ft; (IIIB) 10,300ft

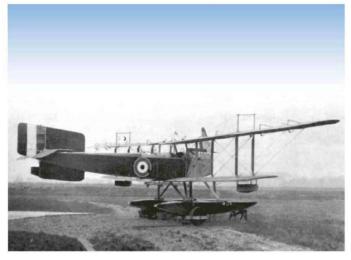
ENDURANCE: (IIIA & IIIB) 4 hr 30 min



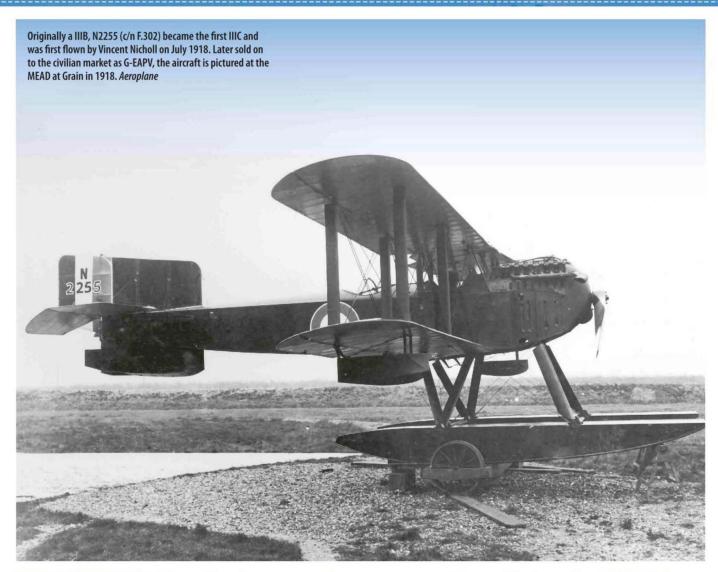
The Fairey IIIB served with 219 Squadron, stationed at Westgate, Kent from October 1918 to February 1920; longer than any other unit. *Via Martyn Chorlton*



The Fairey IIIB only served with two operational RAF units, 219 Squadron at Westgate and 230 Squadron at Felixstowe. *Via Martyn Chorlton*



The prototype IIIB seaplane, during flight trials at Grain in 1918. Note the extended upper wing and the additional bracing provided by the large king-posts. *Via Martyn Chorlton*



TECHNICAL DATA IIIC

ENGINE: One 375hp
Rolls-Royce Eagle VIII
WING SPAN: 46ft 1in
LENGTH: 36ft
HEIGHT: 12ft 2in
WING AREA (Total):
542 sq ft
EMPTY WEIGHT: 3,392lb
LOADED WEIGHT: 4,800lb
MAX SPEED: 110.5 mph
at 2,000ft
CLIMB: 6,500ft in 9 min 30 sec
SERVICE CEILING: 15,000ft
ENDURANCE: 5 hr 30 min

>>> JUL 1918 IIIC first flight

>>> NOV 1918

Enters service with 229 & 230 Sqn

>>> JUN 1919

Action in Northern Russia

The first general-purpose aircraft for the RNAS

DEVELOPMENT

Combining the ability to bomb as per the IIIB and the capability to be used in the reconnaissance role as per the IIIA, the IIIC was one of the best seaplanes designed during the First World War. Unfortunately, the aircraft arrived too late to see action, entering service in November 1918.

DESIGN

The IIIC combined the main features of its two predecessors having the tail and equal-span wings of the III and IIIA and the seaplane floats of the IIIB. It was the engine, an Eagle VIII, which made the biggest difference to the performance of the IIIC. Not only did the Eagle have an excellent power to weight ratio, the engine was also reliable. The additional power meant that much larger fuel tanks could be installed with a capacity of 120 gallons which, potentially, gave the IIIC an endurance of approximately 5½ hours.

Armament was a single synchronised forward-firing Vickers machine-gun for the pilot and a Scarff ringmounted Lewis for the observer. Bombs could be carried on under fuselage racks.

OPERATIONAL SERVICE

Fairey IIIC, N2255, was flown for the first time from

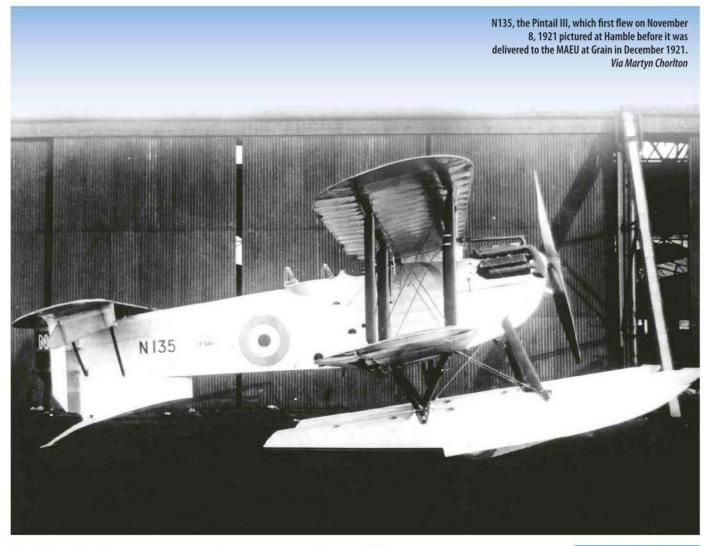
Hamble by Vincent Nicholl in July 1918. However, it was not until September 1918 that the first IIIC arrived at the MEAD (Marine Experimental Aircraft Depot) on the Isle of Grain for testing.

By November, the type had entered service with 229 Squadron at Great Yarmouth and 230 Squadron at Felixstowe, just missing out on seeing active service during the First World War. Both units dispensed with their IIICs in March 1919.

The IIIC did see action in 1919 when at least seven aircraft were embarked on HMS *Pegasus* as part of the North Russian Expeditionary Force based at Archangel. Several IIICs unsuccessfully took part in an attack on four Russian naval vessels in June although better results were achieved during a second raid on rail targets. FIt Lt L Massey Hilton, who was destined to become a director at Fairey, was awarded the DFC for his actions during this short campaign.

PRODUCTION

36 Fairey IIICs were built, N2246, N2255 to N2259 and N9230 to N9259. Four IIICs joined the civilian registry; N9253 became G-EBDI, N2876 to G-EADZ (later G-EAMY), N2255 to G-EAPV and N9256 to G-EARS. The latter was later shipped to Canada via the Aircraft Disposal Co. and re-registered as G-CYCF.



Fairey's first new post-war aircraft

DEVELOPMENT

The Pintail was a fighter-reconnaissance seaplane designed to Specification XXI, issued on May 20, 1919 for a two-seat amphibian which could operate from, land, water or an aircraft carrier. Three prototypes were built, designated from Mk I to III, each only being separated by different amphibious gear and varying lengths of fuselage.

DESIGN

Originally named the Type XXI, after the specification, the Pintail made its first public appearance at the Olympia exhibition in July 1920. Powered by a 475hp Napier Lion V, the aircraft was designed for a crew of two and was armed with a forward fixed Vickers gun, operated by the pilot, and a Lewis gun mounted on a Scarff ring for the observer.

The fuselage was made up in three parts, consisting of the powerplant, a centre-section made of steel which carried the wings and undercarriage and the rear fuselage, which included the rear cockpit. Both the front and rear sections of the fuselage could be removed from the central unit for major servicing, repair or for ease of transportation.

SERVICE

Pintail I, N133, first flew on July 7, 1920 in the hands of

Vincent Nicholl and was originally fitted with amphibious gear where the wheels retracted into the floats. Prone to letting in water, the gear was redesigned so that the wheels were mounted on a swinging frame between the floats as per Fairey III N.10.

The second Pintail, Mk II N134, first took to the air from Hamble on May 25, 1921, again by Nicholl, although freelance test pilot Norman Macmillan is credited with carrying out the type's flight test programme. The Mk II had a longer fuselage and the wheels were, this time, fitted outboard of the floats. The amphibious gear was the main difference introduced by Pintail III, N135, once again first flown by Nicholl on November 8, 1921. N135 had its main wheels fixed into the floats, protruding just enough to allow for land operations and faired sufficiently not to disrupt a landing on water.

The Pintail was never adopted by the FAA or RAF but a small export order of three Pintail IVs from Japan in August 1923 was the type's only success. The Mk IV had its upper wing raised by nine inches, an issue with the original aircraft, the first being delivered to the Imperial Japanese Navy in August 1924.

PRODUCTION

Six aircraft, one Mk I, II and III (N133-N135) and three Mk IVs (F.478-F.180).

TECHNICAL DATA PINTAIL III

ENGINE: One 475hp Napier Lion V twelve-cylinder broad-arrow liquid-cooled WING SPAN: 40ft; (folded) 15ft 8in LENGTH: 32ft 3in HEIGHT: 11ft WING AREA: 400 sq ft LOADED WEIGHT: 4,700lb MAX SPEED: 125 mph at 2,000ft CLIMB: 5,000ft in 5.13 min SERVICE CEILING: 15,000ft ENDURANCE: 5 hr 30 min

>>> JULY 7, 1920
Pintail I, N133
first flight

>>> MAY 25, 1921 Pintail II, N134 first flight

» NOV 8, 1921
Pintail III, N135
first flight

IIID (AKA'IIIC (IMPROVED)')

Fairey IIID Mk II, N9575, one of a batch of twelve aircraft ordered in February 1923 to Contract No.375547/22. The aircraft is pictured over a cloud-covered Malta during its service with 481 Flight, based at Kalafrana. *Via Martyn Chorlton*



The successful family line continues

- >>> AUG 1920
 First flight of the 'Improved IIIC'
- >>> AUG 12, 1921
 Deliveries to
 Australian Navy
- >>> 1924 441 & 444 Flights received IIID
- >>> NOV 1925
 Cape Flight
 formed at Northolt
- >>> MAR-JUN, 1926 11,000 mile record flight
- >>> 1930 IIID retired from FAA

DEVELOPMENT

The IIID ranks as one of the most successful aircraft ever produced by Fairey. This straight-forward, robust and reliable aircraft was a common sight during the 1920s, serving in larger numbers with the FAA and RAF as well as achieving several export orders. Built to Air Ministry Specification 38/22, production continued from 1921 to 1925 by which time, 207 IIIDs had been delivered to the FAA and RAF.

The IIID's greatest strength was its incredible versatility, being able to operate as a landplane from aircraft carriers or shore bases. It could also be launched from a catapult on a warship as a seaplane, and be operated as a general-purpose aircraft, bomber or spotter-reconnaissance with either wheels or floats.

DESIGN

The IIID was a folding, equal-span, two-bay biplane made of wood which incorporated several new design features over its predecessor the IIIC. One of these features was an oleo-pneumatic undercarriage, making the IID the first landplane to use such a system. Another novel feature was that the entire forward fuselage, including the engine and its steel bearers, could be removed as a single separate unit.

In FAA service, the IIID was generally fitted out as a three-seater on floats, although seven were built as trainers or target-tugs in a two-seat configuration with dual-controls. General-purpose variants were armed with a forward fixed Vickers machine-gun, a Scarff ring

mounted Lewis in the rear and a bomb-load of up to 400lb

OPERATIONAL SERVICE

The prototype IIID, N9450, seaplane was first flown by Vincent Nicholl from Hamble in August 1920, who one year later also first flew the landplane version. A batch of 50 was initially ordered by the Air Ministry (N9450-N9499) all powered the Eagle VIII engine. An order for 50 more followed not long after and then another for twelve (N9657-N9578) for the RAF, powered by the 450hp Napier Lion II.

441 and 444 Flights were the first units to receive the IIID in 1924, the same year in which the RAF's carrier-based branch became the FAA. It was with the RAF that the IIID achieved national fame when the first official long-distance formation flight was formed at Northolt in November 1925. The Cape Flight, led by Wg Cdr C W H Pulford, set out in IIID S1102-S1105 on March 1, 1926 from Heliopolis bound for Cape Town. On April 19, all four IIIDs arrived safely in South Africa before beginning the long flight back to England. The quartet staged back through Greece, Italy and France, arriving at Lee-on-Solent (floats had been fitted at Aboukir) on June 21, 1926, having completed a remarkable flight of 13,901 miles without a single mechanical failure or incident.

PRODUCTION

207 built for the FAA and RAF and 20 export (Australia, Portugal and the Netherlands).



TECHNICAL DATA - IIID SEAPLANE, LANDPLANE &'TRANSATLANTIC LOAD CARRIER'(TLC)

ENGINE: One 365hp Rolls-Royce Eagle VIII & IX twelve-cylinder vee liquid-cooled; One 450hp Napier Lion II/IIB/V & VA twelve-cylinder broad-arrow liquid-cooled

WING SPAN: 46ft 11/4in; (TLC) 62ft

LENGTH: (Sea & TLC) 36ft 1in;

(Land) 31ft 5in

HEIGHT: (Sea) 13ft; (Land) 12ft

WING AREA (Total): 500 sq ft

EMPTY WEIGHT (Equipped):

(Sea) 3,990lb; (Land) 3,430lb

LOADED WEIGHT: (Sea & Land)

5,050lb; (TLC) 7,250lb

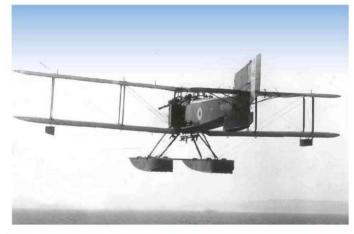
MAX SPEED: (Sea (Lion IIB)) 117 mph; (Sea (Eagle IX) 101 mph; (Land (Lion IIB)) 120 mph; (Land (Eagle IX)) 110 mph; (TLC) 95 mph

CLIMB: (Sea (Lion IIB)) 10,000ft in 12 min 30 sec; (Sea (Eagle IX)) 10,000ft in 25 min; (Land (Lion IIB)) 5,000ft in 4 min 50 sec; (Land (Eagle IX)) 5,000ft in 10 min; (TLC) 350 ft/min

SERVICE CEILING: (Sea (Lion IIB)) 19,500ft; (Sea & Land (Eagle IX) 16,500ft; (Land (Lion IIB)) 20,000ft ENDURANCE: (Sea) 6 hr 30 min;

(Land) 6 hr

RANGE: (TLC) 1,500 miles



Fairey IIID Mk I, N9469, during trials with HMS Argus in August 1922. The aircraft was launched as a seaplane from the carrier's deck from a launching trolley fitted with guide arms running along a monorail. Via Martyn Chorlton



Fairey IIID, ex-N9630, was one of only two to join the civilian register. Fitted with an Eagle IX engine, the seaplane was modified in 1924 for the Real Daylight Balata Estates, British Guiana, as an ambulance. Modifications included a hinged rear deck, port holes and room for a single patient and attendant. Via Martyn Chorlton



Originally built for service with the Cape Flight, IIID Mk III, S1108, is pictured during its service with 445 Flight, HMS Resolution, over the Mediterranean. Via Martyn Chorlton

FLYCATCHER I & II

For almost two decades, a replica of a Fairey Flycatcher (S1287) graced our skies until it was delivered to the FAA Museum at Yeovilton by John Fairey on June 5, 1996. Aeroplane



An aircraft only the designer or its pilot could love

- >>> FEB 1923
 Deck handling
 trials, HMS Argus
- >>> APR 1923
 Enters service
 with 402 Flight
- >>> FEB 19, 1924
 First production
 aircraft flies
- >>> AUG 16, 1927 Flycatcher II makes first flight
- JUN 20, 1930 Last aircraft (S1418) delivered
- >>> JUN 1934
 Retired from
 FAA service

DEVELOPMENT

The Flycatcher was a typical naval aircraft design that sacrificed its appearance for performance, yet despite this, the small, traditional fighter looked the part and, from its introduction in 1923, loyally served the FAA into the mid-1930s.

DESIGN

Designed and built to Air Ministry Specification 6/22 (Deck-landing single-seat fighter to replace the Nightjar), the prototype Flycatcher, N163, was first flown from Hamble by Vincent Nicholl on November 28, 1922.

The Flycatcher could operate as a fighter with a standard undercarriage or be fitted with amphibious landing gear (floats with an integrated wheel protruding by a few inches). The latter option allowed the Flycatcher to land and take-off with the floats still in place. General construction of the fuselage was of wood and metal, while the wings were solely made of wood with all surfaces covered in fabric.

Not the most attractive of aircraft, the Flycatcher had single-bay staggered wings, the upper having a dihedral of five degrees, while the fuselage gave the impression of curving upwards towards the squared off tail unit. Complete with camber gear installed, the Flycatcher was very popular with its pilots and ground crew alike. Easy to fly, very manoeuvrable, the Flycatcher could do no wrong.

The Flycatcher II, of which only one aircraft, N126, was built, first flew on October 3, 1926. Resembling the Firefly I more than its namesake, the aircraft was designed to Specification N.21/26 for a new deck-landing fighter for

the FAA. Written off in an accident in May 1929, none of the aircraft entered for the new specification were accepted by the Air Ministry.

OPERATIONAL SERVICE

From its entry into service, the Flycatcher was the FAA's only standard fighter until the Hawker Nimrod arrived in 1932. It served on every Royal Navy aircraft carrier in land and floatplane configuration, shore bases, platforms from turrets of capital ships and from the fore-decks of HMS *Courageous, Furious and Glorious*.

402 Flight was the first unit to receive the Flycatcher at Leuchars in April 1923 before embarking on HMS *Eagle*. By September 1924, every FAA fighter flight had re-equipped with the Flycatcher, including a pair of new flights and at the type's peak of service in 1930, the fighter was the only type on the strength of eight flights.

It was not until 1934 that the Flycatcher was superseded by Hawker Osprey and Nimrod. The former replaced the Fairey fighter on catapult flights on board warships. 403 Flight, which operated five Flycatchers on the China Station as part of the 5th Cruiser Squadron and 406 Flight with the East Indies Squadron, were the last to see service when they were replaced by Ospreys in June 1934.

PRODUCTION

195 Flycatchers were built between 1922 and 1930 including four prototypes, N163-N165 and N216 followed by production batches, serialled N9611-N9619, N9655-N9680, N9854-N9895, N9902-N9965, S1060-S1073, S1273-S1297, S1409-S1418 and S1590.



TECHNICAL DATA FLYCATCHER I (LANDPLANE & SEAPLANE) & II

ENGINE: (I) One 400hp Jaguar II/IV or Jupiter IV; (II) 540hp Jaguar VIII (supercharged) or 480hp Mercury IIA

WING SPAN: (I Land & Sea) 29ft; (II) 35ft

LENGTH: (I Land) 23ft; (I Sea) 29ft; (II Land) 24ft 9in; (II Float) 28ft 9in

HEIGHT: (I Land) 12ft; (I Sea/amphibian) 13ft 4in; (II Land) 10ft 9in; (II Float) 12ft

WING AREA: (All) 288 sq ft

EMPTY WEIGHT: (I Land) 2,038lb; (I Sea) 2,571lb

LOADED WEIGHT: (I Land) 3,028lb; (I Sea) 3,531lb; (II Land) 3,266lb; (II Float) 3,667lb

MAX SPEED: (I Land) 134 mph at sea level; (I Sea) 130 mph at sea level; (II Land) 153 mph; (II Float) 144 mph

CLIMB: (I Land) 5,000ft in 3 min 41 sec; (I Sea) 5,000ft in 7 min 15 sec

SERVICE CEILING: (I Land) 20,600ft; (I Sea) 14,000ft

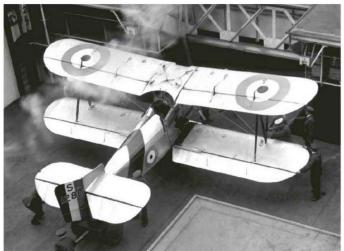
ENDURANCE: (I Land & Sea) 1.82 hr at max speed at 10,000ft



A Fairey Flycatcher taking off from the main flying deck of the aircraft carrier HMS *Glorious* not long after the vessel was commissioned in 1930. *Aeroplane*

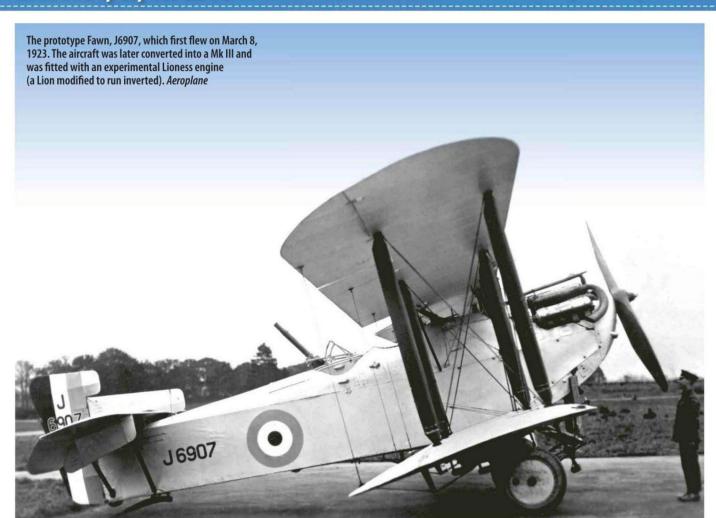


The sole Jaguar VIII-powered Flycatcher II, N216, which took part in trials from October 1926 to May 1929 when it crashed after take-off when the engine cut at Gosport. *Aeroplane*



Clearly the deck lift of HMS *Glorious* was designed for much bigger aircraft than the Flycatcher which had no need to be designed with folding-wings for carrier operations due to its compact proportions. *Aeroplane*

FAWN I, II, III & IV



The RAF's first post-war day bomber

- >>> MAR 1923

 First flight of prototype
- >>> JAN 24, 1924
 First production
 aircraft flies
- >>> MAR 1924
 Enters service
 with 12 Sqn
- >>> APR 1924
 Fawn joins 11 Sqn
- >>> 1925/26 Displays at Hendon
- >>> OCT 1929
 Retired from RAF

DEVELOPMENT

The Fawn was one of only a handful of new aircraft to enter RAF service during the austere and financially difficult 1920s. Many squadrons were still operating aircraft from the First World War so the arrival of the Fawn, which was the first of its kind to enter RAF service following the armistice, was something of anomaly.

DESIGN

The Fawn was effectively the landplane development of the Pintail, which was originally designed as an Army Co-operation and/or reconnaissance machine. Designed to Air Ministry Specification 5/21, the prototype, Fawn I, J6907, was first flown by Vincent Nicholl in March 1923. A two-seater, the Fawn, was armed with a single forward firing Vickers machine-gun and up to two Lewis machine-guns on the obligatory Scarff ring in the rear. A bomb load of up to 450lb, made up of four 112lb bombs or two 230lb, could be carried on under-wing racks.

The prototype had the same short fuselage as the Pintail but a pair of follow-up prototypes, designated as Mk IIs and serialled J6908 and J6909 had longer fuselages and better longitudinal stability. It was this mark which first flew in September 1923 that was chosen for production.

An order for 40 Mk IIs was placed by the Air Ministry to Specification 20/23 and the first production Fawn, J7182 first flew from Northolt on January 29, 1924. Once these

had been built, a follow-up order for 15 Mk IIIs was issued to Specification 1/25. This mark differed in having a 468hp Lion V engine. Another engine change resulted in the Fawn IV, although only six of the Lion VI-powered variant was ordered.

OPERATIONAL SERVICE

The Fawn entered service with 12 Squadron at Andover in March 1924, replacing its ageing DH.9As. 11 Squadron followed in April at Bircham Newton with 100 Squadron being the only other 'full-time' unit to fly the type at Eastchurch. The highlight of the Fawn's service career was performing at the 1925 and 1926 Hendon Air Displays as a 36-strong massed formation. 12 and 100 Squadrons took part in both events.

By late 1926, the Fawn was already being earmarked for replacement by the Horsley and, in 12 Squadron's case, the Fairey Fox. The Fawn continued to serve with fledgling auxiliary 'part-time' units of 503 (City of Lincoln) Squadron at Waddington from October 1926 and 602 (City of Glasgow) Squadron at Renfrew from September 1927. It was with the latter that the type was retired in October 1929.

PRODUCTION

74 Fawns built including six prototypes in the serial ranges: Mk I, J6907; Mk II, J6908, J6909, J7182-J7231; Mk III, J7768-J7779 & J7978-J7985; Mk IV, J7215, J7768, J7773, J7780 & J7781.



TECHNICAL DATA - FAWN I, II & III

ENGINE: (II) One 470hp Napier Lion II twelvecylinder broad-arrow liquid-cooled engine; (III) One 468hp Napier Lion V; (IV) One 520hp Lion VI

WING SPAN: 49ft 11in

WING AREA: 550 sq ft

LENGTH: 32ft 1in HEIGHT: 11ft 11in LOADED WEIGHT:

5,834lb

MAX SPEED: 114 mph

CLIMB: 5,000ft in 6 min

30 sec

SERVICE CEILING:

13,850ft

ABSOLUTE CEILING:

15,500ft

RANGE: 650 miles

ENDURANCE: 5 hr



The second Fawn built, J6908, was the working model for the first production Mk II with a longer fuselage, ventral fins deleted and the fin area increased. Aeroplane



100 Squadron Fawn I, J7192, at Spittlegate in December 1924. The aircraft only briefly served with the unit before being placed in storage at a HAD (Home Aircraft Depot). The aircraft also served with 503 Squadron between December 1927 and March 1928. *Via Martyn Chorlton*



Not the most attractive of Fairey designs, the Fawn's appearance was not improved by the 'official requirement' for a pair of externally mounted fuel tanks on the upper wing. *Aeroplane*



TECHNICAL DATA N.4

ENGINE: (Atalanta) Four 650hp Rolls-Royce Condor IA; (Titania) Four 650hp Condor III WING SPAN: 139ft LENGTH: 66ft WING AREA (Total): 2,900 sq ft LOADED WEIGHT: (Atalanta) 30,500lb; (Titania) 31,612lb MAX SPEED: 115 mph at sea level CLIMB: 5,000ft in 8 min SERVICE CEILING: 14,100ft ENDURANCE: (normal) 7 hr; (maximum) 9 hr

>>> 1919 First N.4 assembled

>>> JUL 4, 1923

Atalanta makes
first flight

>>> JUL 24, 1925

Titania makes
first flight

The world's largest flying-boats

DEVELOPMENT

In late 1917, the Admiralty issued the new Specification N.4 for a very large flying-boat for fleet co-operation and open-sea reconnaissance duties. The RNAS was already heavily reliant on large flying-boats for its open-sea operations and more than 200, the majority of them Felixstowe designs, were in service by the beginning of 1918.

Despite having no previous experience of building large flying-boats, especially 'very large' ones, Fairey won an order for three aircraft, two of them to be constructed by Fairey and one by the Phoenix Dynamo Manufacturing Co. of Bradford.

DESIGN

The management and design of the Fairey N.4 was controlled from Hayes but, even as the project was about to begin, Fairey had to sub-contract one of its flying-boats to Dick Kerr and Co, at Lytham St Annes because of a lack of floor space.

The hull of the N.4 was a Linton Hope design and these were constructed separately by boat-builders; one by May, Harden and May, one by the Gosport Aviation Co. and the other by Fyffes on the Clyde. The project was rapidly spreading itself around the country, with six major companies involved, none of them located near each other and this was destined to cause a great deal of logistical problems.

The N.4 was the largest flying-boat in the world at the time, the giant having a gross weight of over 30,000lb. However, when the Armistice came and went, the problems of scattered production, assembly and transportation saw the project slow to a snail's pace and

the first aircraft, N119, named *Atalanta*, did not take to the air until the summer of 1923. The second aircraft, and the only other to fly was N129 *Titania* in 1925.

The N.4 was an un-equal span biplane powered by four Condor engines which were arranged, un-cowled, as tractor and pusher in joined pairs. The tailplane, which had three fins and rudders, was a biplane arrangement. Two pilots were accommodated in an open cockpit and positions for gunners in the nose and two apertures behind the main-plane were also provided.

SERVICE

N119 Atalanta, following its assembly at Lytham in 1919, was not dismantled until 1921 and transported by road to Grain. Re-assembled, the giant flying-boat made its maiden flight on July 4, 1923 and, after initial flight testing, was later transferred in early 1924 to Felixstowe for further trials with the MAEE.

The second N.4, N129 *Titania*, was assembled at Hamble, once the hull, built in Scotland, and the superstructure, built at Hayes, arrived in late 1921. Further modifications to the hull were needed and these were carried out at Hayes, the hull later returning to Hamble where N129 resided until 1923. Dismantled, the flying-boat was moved to Grain but the first flight did not take place until July 24, 1925 following another move to Felixstowe in June 1924. *Titania* is believed to have been operated solely by the MAEE until early 1929.

The third N.4, serialled N118 and named Atalanta II, was finally assembled at the Phoenix works in Bradford. Destined never to fly, N118 was later dismantled and transported to Grain where the hull was used for floatation testing.





The Fremantle was one of the largest singleengine floatplanes ever built and was originally intended for an attempt on the round-the world record which fell to the United States eight weeks before the Fairey aircraft made its maiden flight. Via Martyn Chorlton

An attempt to be the first 'round-the-world'

DEVELOPMENT

One of many aircraft ordered for 'one-off' production by the Air Ministry, the Fremantle came about from Specification 44/22 (the last of 1922) for a long-range reconnaissance aircraft. The specification failed to mention that the Fremantle was actually ordered for an attempt to be the first aircraft to circumnavigate the world. It was intended to be piloted by Capt R H McIntosh of Handley Page Transport and navigated by Capt F Tymms, the latter serving with the Civil Aviation Department of the Air Ministry, an organisation that had great influence on the specification.

DESIGN

The Fremantle was powered by a Condor III engine fitted with spur reduction gear which could raise the propeller hub, therefore giving the fixed-pitch wooden propeller greater tip clearance above the water. The clearance allowed the designer to incorporate shorter struts between the fuselage and the floats which had the knock on effect of improving aerodynamics and, in turn, improved performance.

The fuselage had a large internal cabin for the crew with large side windows and this was created using mahogany planking rather than obstructive crossed bracing. The cabin was large enough to allow crew to stand upright and was long enough to sleep in, had room for food and water storage and a navigators table.

The pilot's cockpit was aft of the all-metal engine bay, with the added protection of a fireproof bulkhead. The

pilot had a good, unobstructed view, aided by the two-bay biplane wings having a dihedral but no stagger. Good directional stability was provided by a large fixed fin and a big, air-balanced rudder giving outstanding control in the yaw. Aided by Fairey's flap gear and combined with an RAF 15 aerofoil, take-offs and landings were a delight.

The Fremantle was fitted with five floats, the two main ones constructed from mahogany planking for extra strength and to overcome the effects suffered by plywood floats in humid and hot conditions.

SERVICE

By the time the Fremantle was ready for its first flight in November 1924, the round-the-world had already been achieved by a trio of US Douglas World Cruisers. The Fremantle was first flown on November 28, 1924 by Norman Macmillan who carried out flight testing, although this did not include a period with the MAEE at Felixstowe, until June 16, 1925. During this period, Macmillan flew the aircraft for 11hrs 55mins and described the aircraft as, 'a very pleasant aircraft to handle, both on the water and in the air.'

Serialled as N173, the Fremantle joined the MAEE in November 1925 and remained until January 1926 when it was transferred to the RAE at Farnborough for radionavigation development work (presumably converted to a landplane). During 1925, the Fremantle was briefly registered as G-EBLZ for The Air Council, a registration it would have displayed if it had made the round-the-world attempt.

TECHNICAL DATA FREMANTLE

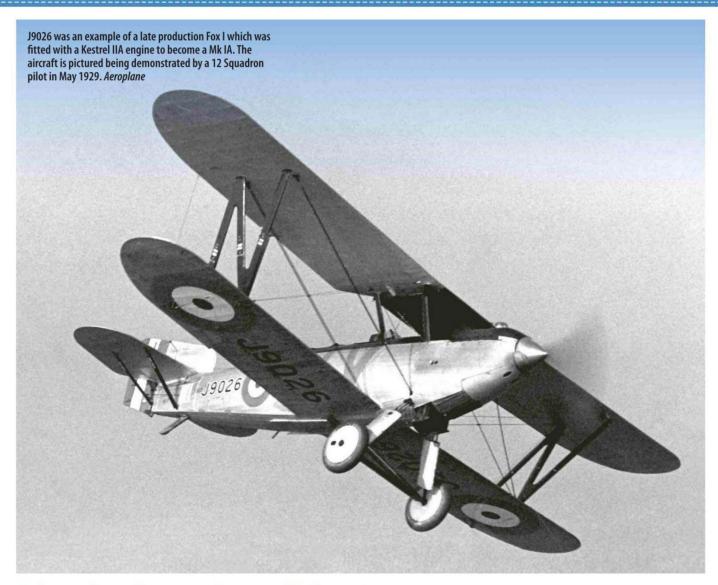
ENGINE: One 650hp Rolls-Royce Condor III twelve-cylinder vee liquid-cooled engine WING SPAN: 69ft 2in LENGTH: 53ft HEIGHT: 20ft 3in WING AREA: 1,095 sq ft LOADED WEIGHT: 12,550lb MAX SPEED: 108 mph CLIMB: 5,000ft in 25 min

» SEP 28, 1924

Round-the-world record achieved

>>> NOV 28, 1924
First flight of
Fremantle

>>> 1926 Transferred to RAE



A bomber faster than a fighter!

- >>> 1923
 C R Fairey visits the USA
- >>> JAN 2, 1925
 First flight of prototype
- >>> DEC 10, 1925

 First production aircraft flies
- >>> JUN 1926 Fox I joins 12 Squadron
- >>> JAN 1, 1929
 Fox IA joins 12
 Squadron
- >>> JAN 1931 Fox replaced by Hart

DEVELOPMENT

It seemed to be an entrenched tradition by the mid-1920s that bombers were slow, lumbering machines while fighters were nimble and fast. The illusion was impressively shattered when Fairey introduced its outstanding Fox in 1925, an aircraft that neither looked nor performed like any other in the RAF inventory, yet it was being presented as a bomber.

50mph faster than the Fawn, the Fox was more than capable of out-pacing the world's in-service fighters, let alone the RAF's. Yet, it was the financial constraints of the period which forced a restricted order of just 28 aircraft and only one squadron receiving the type. The virtually forgotten Fox certainly raised the performance bar for bombers many years before the highly-credited Hawker Hart.

DESIGN

The main reason for the Fox's excellent performance was its streamlined shape and clean lines which were attributed to the novel use of an American powerplant. C R Fairey was on a visit to the USA in 1923 when the Curtiss D-12 engine caught his eye. The engine had a low frontal area and could be cowled very smoothly. Fairey managed to acquire the engine for his new Fox which was not only aerodynamic around the engine

but also across the entire airframe.

OPERATIONAL SERVICE

The prototype Fox, only referred to by its construction number (c/n) F.573, first flew on January 3, 1925. By August, the aircraft was displayed to Air Chief Marshal Sir Hugh Trenchard, the Chief of Air Staff, who was so impressed he ordered enough Foxes to equip an entire squadron without delay. The first production Fox flew on December 10, 1925.

By June 1926, the type entered service with 12 Squadron, under the command of Sqn Ldr Gray, at Andover, replacing its Fawns. The squadron was destined to operate the Fox until they were replaced by the Hart in January 1931.

The Fox made a huge impression on the pilots of 12 Squadron and other units looked on with envy as they were usually out-performed and manoeuvred during air exercises. From this period onwards, 12 Squadron re-badged itself with the head of a fox and the motto, Leads the Field. The Fairey Fox certainly did.

PRODUCTION

28 Fairey Foxes built between 1924 and 1927 serialled; Mk I, J7941-J7958, J8423-J8427; Mk IA J9025-J9028 & J9515.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA - FOX I & IA

ENGINE: (I) One 480hp Curtiss D-12 (Felix); (IA) Rolls-Royce F.XIIA (later Kestrel IIA)

WING SPAN:

(Prototype) 33ft 6in; (Production) 37ft 8in

LENGTH: 28ft 3in

HEIGHT: 10ft 8in

WING AREA:

(Production) 324 sq ft

TARE WEIGHT: 2,609lb

LOADED WEIGHT: (I) 4,170lb; (IA) 4,640lb

MAX SPEED: (I) 156 mph at sea level; (IA) 160 mph at sea level

CLIMB: 2,000ft in 1.8 min

SERVICE CEILING:

17,000ft

RANGE: 500 miles



Fox Is, J7950 and J7954, in company, circa late 1926. J7954 was destroyed after catching fire and blowing up in a field in August 1928 while J7950 was sold to National Flying Services and registered as G-ACXO in 1934. Entered in the England to Australia air race in October 1934, the aircraft saw out its days in New Guinea during the late 1930s. *Via Martyn Chorlton*



Only five Foxes were built to Mk IA standard from new, the rest were converted from the original batch of Mk Is. This was J7945 which was delivered to 12 Squadron in July 1926, converted to a Mk IA and returned to the unit in March 1929. *Via Martyn Chorlton*



Fairey Fox I, J7948, pictured at the RAF Display on July 13, 1929. The aircraft, which remained a Mk I, served with 12 Squadron from July 1929 to June 1930. *Via Martyn Chorlton*



ENGINE: (I) One 400hp Armstrong Siddeley Jaguar IV fourteen-cylinder two-row radial; (II & III) One 425hp Bristol Jupiter nine-cylinder radial WING SPAN: (I) 39ft 10in; (II & III) 40ft 7in LENGTH: 29ft 6in WING AREA: 380 sq ft EMPTY WEIGHT: (III) 2,583lb LOADED WEIGHT: (III) 4,179lb MAX SPEED: (III) 132 mph at sea level CLIMB: (III) 5,000ft in 6 min 30 sec SERVICE CEILING: (III) 15,000ft

>>> JUN 5, 1925 Ferret I N190 first flight

MAY, 1927
Mk III tested by
A&AEE

Replacing the 'Bris-Fit' and the 'Nine-Ack'

DEVELOPMENT

Initially designed to FAA Specification 37/22, the next aircraft to be built by Fairey was the company's first all-metal aircraft.

The Ferret, which was effectively a development of the IIID and is often referred to as the missing 'IIIE', was not received with great enthusiasm by the FAA but a second Specification, 22/26, saw the aircraft being put forward as a DH.9A and F.2B replacement.

DESIGN

The all-metal, three-seater Ferret had a fuselage covered in sheet aluminium while the rest of airframe was covered in fabric. Although based on the IIID, the Ferret was smaller; its wing span was seven feet shorter and its weight was approximately 25% less.

The first of three prototypes was powered by a 400hp Jaguar IV engine while the next two aircraft had a 425hp Jupiter engine. The undercarriage had an oleo V-strut and the wings, both upper and lower, were staggered and sweptback while the tail surfaces remained the traditional Fairey design.

SERVICE

The first aircraft, Ferret I, N190, was first flown from Northolt by Norman Macmillan on June 5, 1925. Not long after, the aircraft was transferred to the A&AEE for flight trials in competition with the Hawker Hedgehog and Blackburn Airedale, both vying for a production from Specification 37/22. However, part way through the trials, 37/22 was abandoned and Specification 26/27 was introduced, which called for a general-purpose aircraft to replace the DH.9A.

The second Ferret, N191, was already close to completion when 37/22 was withdrawn but the third and final Ferret, N192, was at stage where it could be re-designed to compete for 26/27. The aircraft was now a two-seater and was competing against the Bristol Beaver, de Havilland DH.9J Stag, DH.65 Hound, IIIF, Gloster Goral, Vickers Vixen VI, Vickers Valiant and the Westland Wapiti. The Ferret reached the final stages of the flight trials in May 1927 and, until problems began occurring with the engine and propeller, it was on the verge of winning the competition outright. However, the Ferret failed but Fairey still won a major production for the IIIF



A fighter to look after the fox

DEVELOPMENT

It would not be unrealistic to surmise that the Fairey Firefly I was designed as an escort fighter for the Fox bomber. Both were creations of Marcel Lobelle and both were designed around the same Curtiss D-12 engine which was renamed Felix by Fairey. Both aircraft were made from wood and both were private ventures and, while only one Firefly I would be built, the Fox was ordered, albeit in small numbers.

DESIGN

The Firefly was conceived in March 1924, after Fairey gained a licence to build the D-12 engine in Britain. A request was made to the Air Ministry for its latest requirements for a single-seat fighter from whence the reply stated; seating for a single pilot, two forward-firing machine-guns with room for 1,200 rounds and fuel capacity large enough for at least two hours' endurance.

To back up this information, the Air Ministry also, generously, gave Fairey the specifications for the Napier Lion IV powered Gloster Gorcock and the future Rolls-Royce Condor IV powered Hawker Hornbill. Both aircraft would ultimately become nothing more than research aircraft.

SERVICE

The sole Fairey Firefly I (c/n F.572) was flown for the first time by Norman Macmillan from Northolt on November 9, 1925. Manufacturer's trials were completed on December 2 and, following flight testing by the A&AEE, the Air Ministry report was very positive. The view for the pilot was one particular feature of the Firefly, a very important factor for a fighter, that was commented on as being far superior to the visibility from a Gloster Gamecock or Armstrong Whitworth Siskin. However, the Air Ministry report was bluntly ended with the statement, 'not likely to buy more than one unless redesigned with a British engine'.

It was then proposed that the fighter be fitted with the 490hp Rolls-Royce F.10 (later named Kestrel) engine which, following Air Ministry calculations, should have given the Firefly an improved maximum speed and greater ceiling.

Fairey set about re-designing the Firefly to take the geared Rolls-Royce F.II which was much heavier and larger than the original American D-12 powerplant. As a result, on paper at least, Fairey predicted that a Rolls-Royce powered Firefly I would be a failure and the company withdrew the offer to the Air Ministry of its latest fighter design.

TECHNICAL DATA FIREFLY I

ENGINE: One 430hp Curtiss D-12C twelve-cylinder vee liquid-cooled engine WING SPAN: 31ft 6in LENGTH: 24ft 10in HEIGHT: 9ft 1in LOADED WEIGHT: 2,724lb MAX SPEED: 185 mph CLIMB: 5,000ft in 2.4 min

- >>> MAR 1924
 Firefly conceived
- >>> NOV 9, 1925
 First flight of Firefly
- >>> DEC 2, 1925

 Manufacturers

 trials complete



Most prolific, longest-serving aircraft of the inter-war period

- >>> 1924 Specification 19/24 issued
- >>> MAR 19, 1926
 IIIF makes
 maiden flight
- >>> MAY 1927
 FAA receives its
 first IIIF
- JAN 1928
 IIIF officially on RAF strength
- >>> 1935
 IIF retired from RAF
- >>> 1936
 IIIF retired from FAA

DEVELOPMENT

The last of the long line of III series aircraft, the IIIF, was produced in greater numbers than any other British operational aircraft from the end of the First World War to the mid-1930s (except the Hawker Hart family). A refined version of the IIID, over 600 were built.

DESIGN

Following the failure of Specification 37/22 for a three-seat deck landing reconnaissance aircraft to replace the Blackburn and Bison, a new Specification, 19/24, for a three-seat fleet reconnaissance aircraft for the FAA and a two-seat general-purpose aircraft for the RAF, was specifically written for the IIIF.

In late 1926, to early 1927, two prototypes, N198 and N225, were ordered for evaluation. N198 was first flown by Norman Macmillan on March 19, 1926. Constructed with a wood and metal composite fuselage and wooden wings, the first aircraft was flown as a landplane but, not long after, was sent to Hamble, fitted with a pair of Fairey metal floats and sent for further evaluation to the MAEE at Felixstowe.

OPERATIONAL SERVICE

The new IIIF was well received and, to hasten its entry into service, the last ten IIIDs built were converted to IIIF standard and shipped to Aboukir, Egypt for theatre trials with 5 and 60 Squadrons. These trials were complete by April 1927 and the first of many production orders

followed, the initial examples being IIIF Mk IV C/M (GPs). The majority went on to serve with 8 Squadron in Aden and 207 Squadron at Eastchurch.

The FAA received its aircraft in May 1927 when the IIIF Mk I entered service with 443 (Composite) Flight for active service on board HMS *Furious*. Service with Catapult Flights on board warships of the 2nd, 6th and 8th Cruiser Squadrons soon followed. By December 1927, the first Mk IVs had arrived with the RAF's first front-line unit, 47 (General Purpose) Squadron at Khartoum, while the FAA had established 440, 445, 446 (Fleet Spotter Reconnaissance) Flights on board HMS *Argus* and *Courageous*.

The majority of IIIFs supplied to the RAF were the Mk IV, although 61 Mk Is and IIIs were also transferred to the RAF over a period of time. The sub-variant of the Mk IV was designated C/M which indicated the aircraft was of composite metal and wood construction while others were designated as M/As, which was all-metal construction.

The IIIF was phased out of front-line RAF service in 1935 while the FAA continued to use the type operationally until the following year, although several examples remained on strength until 1941.

PRODUCTION

55 Mk Is were built, followed by 33 Mk IIs, 291 Mk IIIs and 243 Mk IVs. The Mk V variant became the Gordon while the Mk VI became the Seal.



Andy Hay/www.flyingart.co.uk

IIIF MK I LANDPLANE, MK II SEAPLANE, MK IIIM/B GP LANDPLANE, MK IIIM/B SEAPLANE

ENGINE: One Napier Lion VA, XI or XIA twelve-cylinder broad-arrow liquid-cooled engine

WING SPAN: 45ft 9in

LENGTH: (I) 33ft 10in; (II & III sea) 35ft 6in; (III land) 34ft; (IV) 36ft 9in

HEIGHT: (I) 11ft 3in; (II) 12ft 7in; (III land) 12ft 9in; (III sea) 14ft; (IV) 14ft 2in

WING AREA: (Prototype) 443 sq ft

LOADED WEIGHT:

(I) 5,120lb; (II) 5,300lb; (III land) 5,874lb; (III sea) 6,301lb; (IV) 6,041lb

MAX SPEED: (I) 150 mph; (II) 135 mph; (III land) 136 mph; (III sea); 130 mph; (IV) 120 mph at 10,000ft

CLIMB: (III land) 5,000ft in 5.56 min; (III sea) 5,000ft in 6.42 min;

SERVICE CEILING: (III sea)

20,000ft

ENDURANCE: (III sea) 3 to 4 hr



45 Squadron operated the IIIF Mk IV from August 1929 to December 1935 in both float and landplane versions. This is K1702 '6' at rest on the Nile at Khartoum in March 1932. *Aeroplane*



Fairey IIIF Mk I S1189 had a long, varied career which began at HAD Henlow in May 1927 and ended in 1941 with the Royal Hellenic Navy. The aircraft is pictured whilst serving with 443 Flight from HMS Furious in late 1928. Aeroplane



S1795 was one of a batch of 68 IIIF Mk IIIBs built to Contract No.110958/31 and first delivered to the FAA Pool at Gosport in December 1931. The aircraft was then transferred to the SoNC (School of Naval Co-Operation) at Lee-on-Solent and coded 'H' in August 1932. *Aeroplane*

FIREFLY II



The first all-metal Firefly

- >>> FEB 5, 1929
 First flight of
 Firefly II
- >>> JAN 6, 1930
 First flight of
 Firefly IIM
- » AUG 1930 Belgian Air Force place order
- >>> SEP 12, 1931
 Avions Fairey
 formed
- >>> FEB 26, 1932
 First Avions-built
 aircraft delivered
- >>> NOV 1942
 Type still on
 strength with BAF

DEVELOPMENT

Another private-venture designed by Marcel Lobelle, the Firefly II had little in common with its predecessor other than it being a single-seat biplane fighter. The Firefly II was already being prepared on the drawing board when Specification F.20/27 was issued, calling for a fast-climbing, single-seat interceptor fighter powered by a radial engine. The latter powerplant was soon dispensed with in favour of a liquid-cooled inline engine, which was in keeping with Lobelle's thinking around fighter design at the time.

DESIGN

The Mk II was structurally different from the Mk I and the wings were prominently staggered with 'N' shaped inter-plane struts while power was provided by a 480hp F.XIS engine. The Fairey II was also modified by replacing the retractable radiator and interconnected surface-cooling system with a fixed radiator under the fuselage between the undercarriage legs.

The construction of the fuselage, wings, tail, undercarriage and inter-plane struts were also changed and the wires that joined the pairs of ailerons were replaced by a single strut. Fuel capacity rose to 52 gallons, split between a 40 gallon main and a twelve gallon gravity tank. The open cockpit was heated to such a degree that the pilot could make notes with bare hands at 21,000ft; a height where the outside air temperature would be around -34°C.

SERVICE

First flown on February 5, 1929, the Firefly II (c/n F.1130) was entered into that year's 'interceptor' competition at the A&AEE, Martlesham Heath. Unfortunately for Fairey, the

Firefly II was marginally beaten by the Hawker Hornet (developed into the Fury and ordered for RAF in 1930). Despite having superior speed, the aircraft lost out because of heavy controls and the fact that it was made of wood.

Rebuilt with a metal fuselage and redesignated as the Firefly IIM, the fighter first flew in its new guise from Northolt on January 6, 1930. In June 1930, the Mk IIM was demonstrated at the RAF Display at Hendon and it was here that it caught the attention of several potential overseas customers, including the Belgians.

Registered as G-ABCN, the Firefly IIM was flown to Belgium by Chris Staniland on July 23, 1930 where it was demonstrated against the Avia B.H.33 and the Dewoitine D.27. The Firefly won the day and by late summer 1930, a contract was signed for an initial order of 25 aircraft. Built at Hayes but assembled in Belgium, a follow up order for 20 more Firefly IIMs soon followed which would be built by a new, associated company called Avions Fairey which was formed on September 12, 1931 at Gosselies near Charleroi.

The Firefly IIM joined the Belgian Air Force (BAF) in late 1931, the first unit being 1 Squadron, 1 Group at Schaffen. Other units followed and up to the German invasion in May 1940, 50 Fireflies were still in service. Several were also shipped to North Africa in June 1940 and several were still parked on the airfield at Oran's La Sénia in November 1942.

PRODUCTION

88 Firefly IIMs built; the first 25 (c/n F.1489 to F.1513) built at Hayes, the remainder by Avions Fairey totalling 62 (c/ns F.1651 to F.1670, F.1693 to F.1722, F.1928 to F.1933 and F.2032 to F.2037). One aircraft (F.1876) was built for sale in the Soviet Union.



TECHNICAL DATA - FIREFLY II & IIM (BELGIAN)

ENGINE: One 480hp Rolls-Royce F.XIS (later named Kestrel IIS) twelve-cylinder vee liquid-cooled engine

WING SPAN: (II) 30ft 8in; (IIM) 31ft 6in

LENGTH: 24ft 8in

HEIGHT: (II) 9ft 1in; (IIA) 9ft 4in

EMPTY WEIGHT: (Equipped) 2,387lb

LOADED WEIGHT: 3,285lb

MAX SPEED: 175 mph at sea level

CLIMB: 19,685ft in 10 min 55 sec

SERVICE CEILING: 30,840ft



Labelled as the F.1130 (MOD), this is the original Firefly II following modification to Mk IIM and registered as G-ABCN. The aircraft remained in use by Fairey until 1932. *Via Martyn Chorlton*



A show of strength by the Belgian Air Force as 2 Squadron 'Comete', 1 Group displays its 30 Firefly IIMs at Schaffen in 1932. *Via Martyn Chorlton*



F.1130 as the Firefly IIM pictured in June 1930 only weeks before it was successfully demonstrated displaying the serial G-ABCN. *Via Martyn Chorlton*

FIREFLY IIIM & IV



A naval Firefly to replace the Flycatcher

- MAY 17, 1929
 Mk III flies for first time
- >>> DEC 10, 1929
 Mk IIIM makes
 maiden flight
- >>> MAR 1, 1930
 Trials onboard
 Furious
- MK IIIM emerges as twin-float
- >>> NOV 24, 1933 Mk IV flight tested by Staniland
- 33 JUN 2, 1934 S1592 delivered to Boscombe

DEVELOPMENT

Another private venture, also by the name of Firefly, was running alongside the Mk II and Mk IIIM during 1929. This was being designed to meet the latest Royal Navy Specification, N.21/26, for a single-seat carrier/decklanding fighter for service with the FAA. Therefore, the Firefly III (c/n F.1137) was born which, on the surface, resembled the Mk II.

DESIGN

The Firefly III had a larger area wing with a span of 33ft 6in, exactly two feet longer than the Mk II. Powered by a Rolls-Royce F.XIMS (moderately supercharged) engine, it was first flown in this configuration by Norman Macmillan on May 17, 1929. As with the Mk II before though, the Firefly III was returned to the Hayes factory for a period of six months to re-emerge as the Mk IIIM on December 2. As well as many metal parts being incorporated, the Firefly IIIM was strengthened for catapult operations, fitted with floatation gear and wheel brakes and was capable of carrying a quartet of 20lb bombs. The engine was also replaced by a fully supercharged Rolls-Royce F.XIS.

SERVICE

The Firefly IIIM was first flown on December 10, 1929 by Macmillan and then delivered to Martlesham Heath on March 1, 1930 for its first batch of tests. By June, the Mk IIIM began carrier trials aboard HMS *Furious*. The Armstrong Whitworth Starling, Hawker Hornet and

Vickers 177 were also trialled at the same time, but eventually the Firefly IIIM lost out to Hawker again, when N.21/26 was won by the Hawker Norn, later re-named the Nimrod. In September 1930, the Mk IIIM displayed the registration G-ABFH but would later appear with the military serial S1592.

The Mk IIIM, S1592, was converted to a twin-float plane in early 1931 and, on March 12, was delivered to the MAEE at Felixstowe for flight testing. During August and September 1932, S1592, along with the Fairey Fleetwing, joined the RAF's High Speed Flight who were preparing for that year's Schneider Trophy competition. The aircraft was employed for floatplane training and for weather checks before returning to the MAEE for additional testing.

By October 1931, S1592, according to MAEE records, had completed 137 flying hours, 81 one of these were on floats. By 1932, the floats had been replaced by a neatly spatted undercarriage and, after a spell with the RAE, the sole Firefly IIIM was last recorded as joining the A&AEE in June 1934.

A pair of Belgian-built Mk IIMs were converted to Mk IV standard in 1933 by fitting a Hispano-Suiza twelve-cylinder engine in one and a Rolls-Royce Kestrel IIS in the other. One aircraft (A.F.5050 (AF = Avions Fairey)) was purchased back by the parent company and flight tested by Chris Staniland at Harmondsworth on November 24, 1933. After a short period of testing, it was apparent that the performance from either engine was not much better than the original Mk IIM.



TECHNICAL DATA - FIREFLY IIIM & IV

ENGINE: (IIIM); One Rolls-Royce F.XIMS and later one F.XIS (fully supercharged); (IV) One 785hp Hispano-Suiza 12Xbrs twelve-cylinder vee liquid-cooled engine and one Rolls-Royce Kestrel IIS

WING SPAN: 33ft 6in LENGTH: 25ft 4in HEIGHT: 9ft 10in EMPTY WEIGHT: (IV) 2,468lb

LOADED WEIGHT: (IIIM) 3,816lb; (IV) 3,405lb

MAX SPEED: (IIIM) 188 mph at sea level; (IV) 192 mph at sea level

CLIMB: (IIIM) 5,000ft in 2 min 48 sec; (IV) 13,120ft in 4 min 29 sec

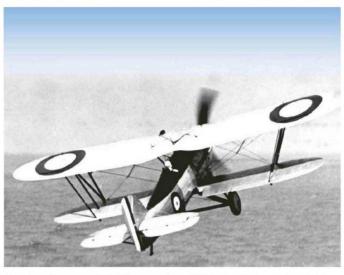
SERVICE CEILING: (IV) 29,520ft



The sole Fairey Firefly III naval fighter, pictured in its first form wearing its construction number F.1137 prior to the civilian serial, G-ABFH during and carrier trials and then the serial S1592. *Aeroplane*



S1592 pictured during its service with the RAF High Speed Flight off Calshot in preparation for the 1931 Schneider Trophy race. *Via Martyn Chorlton*



The Fairey IIIM pictured climbing away from HMS *Furious* during carrier trials in June 1930. *Via Martyn Chorlton*

LONG-RANGE MONOPLANE I & II



- J927 Draft Specification 33/37 accepted
- » NOV 14, 1928 J9479 made maiden flight
- Jones-Williams/ Jenkins flight to India
- DEC 16, 1929
 J9479 crashes
 near Tunis
- >>> JUN 30, 1931 K1991 made maiden flight
- >>> FEB 6-8, 1933 World Distance Record set at 5,309 miles

DEVELOPMENT

The Fairey Long-range Monoplane came about in 1927 because of the failure of the Hawker Horsley to capture the World's long-distance record for Britain. This expensive exercise was supported by the Air Ministry and, when the cost of the project was questioned in the Commons, supporters of the idea managed to veil the aircraft as a valuable asset in exploring long-distance flight. This was partly true and, if the aircraft was successful, it could potentially create both a great deal of publicity for the RAF and kudos for Britain.

DESIGN

One of the most attractive aircraft of its day, the story of the Long-range Monoplane began in December 1927 when the Air Ministry issued draft specification 33/27, which was received with great enthusiasm by Fairey's chief engineer, Maj T M Barlow. The specification was an open one and designs presented for the aircraft were as a biplane, high-wing and low-wing monoplane versions which were all studied. A high-wing cantilever arrangement was chosen following successful wind-tunnel testing. The need for a gravity feed fuel system to the engine from large fuel tanks, with a capacity of 1,000 gallons in the wing, helped to secure a high-wing design.

Power was provided by a Napier Lion XIA and flight instruments, as well as the standard array, included a rate-of-turn indicator, which was a very rare instrument in 1928.

SERVICE

The first of two Long-range Monoplanes, J9479, made its maiden flight from Northolt on November 14, 1928 in the hands of Sqn Ldr A G Jones-Williams and Flt Lt F V Major. After preliminary testing, the aircraft was flown to Cranwell, where a specially tuned Lion XIA was fitted in preparation for the aircraft's record breaking flight in the

spring of 1929. The task of fitting the new engine did not go smoothly and it was not until December 8 that J9479 was returned to Northolt for further flight testing and this resulted in further engine changes.

A 24-hour endurance flight which was initially planned to take place in December 1928, was actually carried out on March 22/23, 1929. During this flight it was calculated that J9479 could potentially fly 5,200 to 5,500 miles, despite a valve in the engine burning out.

By this time, the original idea of carrying out the first flight to South Africa was postponed but an alternative trip to India was feasible. On April 24, 1929, Sqn Ldr Jones-Williams and Flt Lt N H Jenkins took-off from Cranwell at 0957hrs and, 50hrs 37mins later, landed at Karachi on April 26. The intended destination was Bangalore which would have broken the distance record, but unexpectedly strong head-winds forced J9479 down early with just eight gallons of fuel remaining.

The original South African flight was attempted on December 16, 1929 when, once again, Jones-Williams and N H Jenkins departed Cranwell. Sadly, the aircraft hit high ground south of Tunis, killing both crew and destroying J9479.

In 1931, a second Long-range Monoplane was built, K1991, constructed to Air Ministry Specification 14/30. K1991 only differed from the original aircraft in having an automatic pilot, revised fuel system and spatted wheels. On October 27/28, in the hands of Sqn Ldr O R Gayford and Flt Lt D L G Bett, K1991 was flown from Cranwell to Abu Sueir, Egypt, in 31½ hours, a distance of 2.857 miles.

It was not until February 1933 that the hard work of Fairey and the RAF bore fruit. On February 6, Sqn Ldr Gayford and Flt Lt G E Nicholetts left Cranwell at 0710hrs and landed at Walvis Bay, south west Africa at 1635hrs on February 8. The distance of 5,309 miles in 57hrs 25mins captured the World's Long-distance Record for Britain.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA - MONOPLANE I & II

ENGINE: One 570hp Napier Lion XIA

WING SPAN: 82ft

LENGTH: 48ft 6in

HEIGHT: 12ft

WING AREA: 900 sq ft

LOADED WEIGHT:

(I) 16,000lb; (II) 17,500lb

ESIMATED STILL-AIR RANGE:

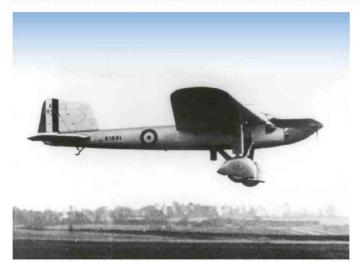
(I) 4,900 miles; (II) 5,550 miles

ACTUAL TAKE-OFF DISTANCE:

(I) 3,705ft; (II) 4,500ft



Sqn Ldr A G Jones-Williams and Flt Lt N H Jenkins pictured in front of J9479 at Karachi in April 1929; the distance covered was 4,130 miles in 50hr 37 min. Via Martyn Chorlton



K1991 sets out from Cranwell on October 27, 1931 on its proving flight to Abu Sueir in Egypt. The distance of 2,857 miles was covered in 31hrs. Via Martyn Chorlton



The second Long-range Monoplane, K1991, prior to its maiden flight on June 30, 1931. K1991 differed from its predecessor in several subtle ways, including a different fuel system, landing-wheel spats, and an automatic pilot by the name of 'George' a nickname that the device was stuck with for many years after. *Via Martyn Chorlton*

The only Fairey Fleetwing was N235 which is shown here in its original form in May 1929 with wire-connected ailerons and standard rudder without a horn-balance. *Aeroplane*



TECHNICAL DATA FLEETWING LANDPLANE & FLOATPLANE

ENGINE: One Rolls-Royce F.XI (Kestrel I) and later a Rolls-Royce Kestrel IIMS WING SPAN: 37ft LENGTH: (Land) 29ft 4in; (Sea) 32ft HEIGHT: (Land) 11ft 5in; (Sea) 12ft 6in WING AREA: 363 sq ft LOADED WEIGHT: (Land) 4,737lb; (Sea) 5,100lb MAX SPEED: (Land) 169 mph; (Sea) 156 mph CLIMB RATE: (Land) 5,000ft in 4.2 min; (Sea) 5,000ft in 5.8 min

- >>> NOV 1926
 Specification issued
- **MAY 16, 1929** First flight of N235
- >>> JUL 18, 1932 Aircraft wrecked

A two-seat spotter for the fleet

DEVELOPMENT

The FAA had desired a new twin-seat spotter reconnaissance aircraft from 1926. The same year, Specification 0.22/26 was issued for such a machine. Marcel Lobelle responded in typical style by designing a very clean aircraft, not dissimilar in appearance to the attractive Fox.

DESIGN

The Fleetwing followed the same construction technique as the Fox IIM with an all-metal, steel-tube fuselage and steel-strip wing spars. Power was initially provided by a Kestrel I and later by a moderately supercharged Kestrel IIMS. The Fleetwing first flew with wooden wings but, following trials with the A&AEE, these were replaced by metal ones, the fin area was increased and a horn balance was fitted to the rudder.

Armament was a single Vickers machine-gun which was fired through a blast trough located on the port side of the engine cowling by the pilot, while the observer/gunner was equipped with a Lewis machine-gun mounted on a Fairey high-speed mounting. Four 20lb bombs could be carried on racks under the port wing.

Specification O.22/26 was a hotly contested requirement and Blackburn, Short Brothers and Hawker also entered. The all-dominant Hawker won the day yet again, this time with a modified version of the Hart which would emerge as the Osprey. However, the Fleetwing achieved a commendable second place during the

evaluation period, which was carried out at the MAEE during May and June 1930.

SERVICE

The one and only Fleetwing (c/n F.1132) was serialled N235 and flew for the first time in the hands of Norman Macmillan on May 16, 1929 from Northolt. Following preliminary trials at Martlesham Heath, the aircraft performed deck trials on board HMS *Furious* in June. Following a return to Fairey for the early modifications, the Fleetwing was back in the air by September and, in October 1929, carried out final service trials at Martlesham.

As well as the O.22/26 trials, the Fleetwing also operated alongside the prototype Hart and Blackburn Nautilus as part of 405 Flight on HMS *Furious* from January to March 1930. Later, in April 1932, the Fleetwing also served aboard HMS *Norfolk* for various trials, including extensive catapult work. 29 flights were carried out from *Norfolk*, the final one on July 18 which ended in a forced-landing in heavy seas alongside the ship. Unfortunately, the float undercarriage collapsed and sufficient additional damage was caused during the recovery to warrant the Fleetwing being written off.

N235 had proved to be a very useful aircraft during its three-year career its duties included being used as hack, along with the Firefly III, as part of the RAF High Speed Flight, during its build up to the Schneider Trophy race in 1931.

Pictured at the Great West aerodrome, three Hayes-built Fairey Fox IIs await delivery by Belgian pilots, Capt Guillaume, Adjt Caryn and Verboomen in January 1932. *Aeroplane*



DEVELOPMENT

The Fox II came about in haste after Fairey failed to be invited to tender for Specification 12/26 for a two-seat high-performance day bomber, despite the fact that the Fox I was already in RAF service. The Fox IIM, an all-metal successor of the Fox I and Mk IA, failed to win the specification which was won by the Hawker Hart but, along with the Firefly, proved to be a success in Belgium where the aircraft and its many derivatives were built in healthy numbers.

DESIGN

Designed by Marcel Lobelle and P A Ralli, the structure, aerodynamics and arrangement were very different from the Fox I. Built from steel with a tubular structured fuselage, the aircraft was fabric-covered and the main wing spars were made of drawn high-tensile strip. The Fox II also introduced a horn-balanced rudder and fin which would continue to be used on the Seal, Gordon II and several other Fairey biplanes. Armament was the same as the Fox I while power, for the prototype at least, was provided by 480hp supercharged F.XIB (later Kestrel IB).

SERVICE

The prototype, designated as the Fox IIM and serialled J9834, first took to the air from Northolt on October 25, 1929 flown by Norman Macmillan. Later registered as G-ABFG, the

prototype was flown to Belgium in late 1930 by Chris Staniland for a demonstration. Subsequent negotiations were successful on the condition that the type was maintained, repaired and later assembled in Belgium.

A contract was signed for a dozen Hayes-built Foxes, the first of which flew in December 1931 and was delivered to the Belgian Air Force a few weeks later. Simply designated the Fox II, these were powered by a supercharged Kestrel IIS engine and, following the establishment of Avions Fairey, a further 31 were built under licence including a pair of Fox IISs which were dual-controlled.

The Fox III (F.1842) was a Hayes-built demonstrator which first flew on June 22, 1933, appearing at the SBAC display at Hendon just four days later. Registered as G-ABBY, the aircraft was modified into a reconnaissance fighter with a pair of forward firing machine-guns and spatted undercarriage. It was later heavily demonstrated in the Far East but failed to secure any orders.

The follow-up Mk IIIs were built in Belgium in three variants, the first powered by a Kestrel IIS with twin forward-firing machine-guns. The second variant was a trainer, designated as Fox IIIS with a Kestrel IIMS and finally the most prolific mark was the Fox IIIC, the 'C' being for combat. Approximately 48 IIICs are believed to have been built with twin machine guns and an enclosed cockpit. The later production IIICs were powered by a 600hp Kestrel V engine, these were designated as the Fox V.

TECHNICAL DATA FOX II, III, IV AND V

ENGINE: (IIM) One 480hp Rolls-Royce F.XIB (later Kestrel IB); (II) One 480hp Kestrel IIS; (III) One 525hp Kestrel IIMS; (Trainer) One 360hp Armstrong Siddeley Serval

WING SPAN: 38ft LENGTH: 29ft 10in HEIGHT: 10ft LOADED WEIGHT: 4,665lb MAX SPEED: 189 mph CEILING: 28,860ft

>>> OCT 29, 1929 J9834 first flight

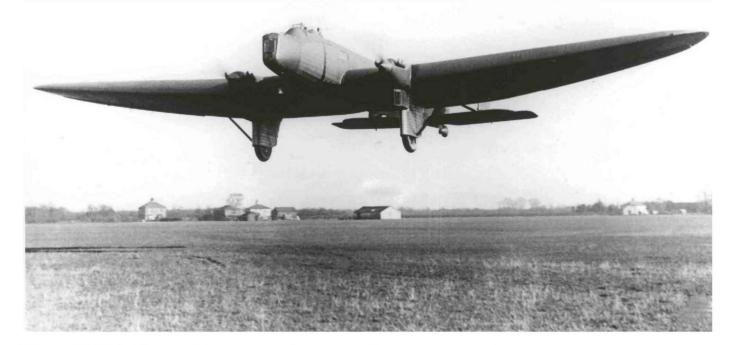
>>> JUL 1932
Fox enters service

with BAF

>>> JUL 1934
Fox IV demonstrated in China

HENDON (FAIREY NIGHT BOMBER)

With Flt Lt Chris S Staniland at the controls, the prototype Fairey Hendon, K1695 takes off from Harmondsworth aerodrome which is today better known as Heathrow. *Via Martyn Chorlton*



The RAF's first all-metal low-wing monoplane

- >>> NOV 25, 1930 Prototype makes maiden flight
- >>> NOV 13, 1931
 Revised prototype
 made first flight
- >>> JUN 25, 1932 Public debut at RAF Display, Hendon
- >>> NOV 1936
 Hendon enters
 service with 38 Sqn
- >>> JUN 15, 1937 Hendon helps reform 115 Sqn
- >>> JAN 1939
 Hendon replaced
 by Wellington

DEVELOPMENT

The Hendon broke the company mould in more ways than one, being Fairey's only deviation towards a heavy bomber design and the first twin-engined aircraft since the F.2 of 1917. Originally named the Fairey Night Bomber (the name Hendon was not attached until October 1934), the aircraft was designed to the 1927 Specification B/19/27. This specification, of which all other entrants were biplanes, was eventually won by the stately Handley Page Heyford leaving the advanced design of Hendon adrift.

DESIGN

Designed by D L Hollis Williams and P A Rallis, who had produced the unique Long-range Monoplane, the Hendon was heavily influenced by the duo's knowledge of stressing and aerodynamics. The all-metal, low-wing deep-wing cantilever 'Fairey Night-bomber', when first penned in 1927 had a host of novel features including an internal bomb-bay which could accommodate the latest 1,000lb bombs and a fully streamlined undercarriage. There was an internal corrugated walkway which traversed from the nose to the tail gunner's position. This was made possible by offsetting the pilot's cockpit which, unusually for the day, was fitted with a canopy.

OPERATIONAL SERVICE

The prototype, K1695, having been built at Hayes was transported to the aerodrome at Harmondsworth (aka Great West and later Heathrow) where it was reassembled in November 1930. On November 17, Fairey

chief test pilot, Norman Macmillan, carried out taxying trials and, eight days later, with the chief designer, Hollis Williams, on board, the Hendon made its maiden flight.

On March 15, 1931, K1695 over ran the airfield boundary and was temporarily returned to the workshops from where it re-emerged with a pair of Rolls-Royce Kestrel IIs engines, rather than the original Bristol Jupiter Xs.

Considering the Hendon was designed to a 1927 Specification, the Air Ministry still ordered 14 examples as a stop gap prior to the arrival of the Vickers Wellington and the Armstrong Whitworth Whitley. The first production aircraft flew from Barton on September 24, 1936 and entered service with 38 Squadron at Mildenhall in November. The production aircraft were designated as the Hendon II as they were fitted with Kestrel VI engines with more power and an enclosed turret was provided for the front gunner rather than an open cockpit. Fairey three-blade propellers were also fitted in place of the Hendon Is wooden two-blade types.

The Hendon II also served with 115 Squadron when 'B' Flight of 38 Squadron was used to reform the new unit at Marham on June 15, 1937. By August, 115 Squadron had re-equipped with the Harrow while 38 Squadron continued to operate the Hendon II until January 1939 when it was replaced by the Wellington.

PRODUCTION

15 Hendons built including one prototype, K1695 plus K5085-K5098. A further order for 62 Hendons was later cancelled.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA - HENDON

ENGINE: (Proto) Two 525hp Bristol Jupiter XF; (II) Two 695hp Rolls-Royce Kestrel VI fully-supercharged

WING SPAN: 101ft 9in

LENGTH: 60ft 9in

HEIGHT: 18ft 9in

WING AREA: 1,447 sq ft

EMPTY WEIGHT: 12,773lb

LOADED WEIGHT:

20,000lb

MAX SPEED: 155 mph

at 15,000ft

CRUISING SPEED: 133 mph at 15,000ft

RATE OF CLIMB:

940 ft/min

CLIMB: 10,000ft in 15 min

SERVICE CEILING:

21,400ft

RANGE: 1,360 miles



The prototype Hendon at the A&AEE, Martlesham Heath, in May 1925, complete with canopy, enclosed front turret (mock-up in this view) and a pair of Kestrel VI engines. *Via Martyn Chorlton*



Delivered to 38 Squadron at Marham on January 19,1937 and coded 'C', K5092 served with the unit until January 1939. By August 1939, the aircraft was reduced to instructional airframe no. 1617M with 1 E&WS (Electrical & Wireless School) at Cranwell, where the bomber is pictured, picketed out in the open. *Via Martyn Chorlton*



K5085 made its maiden flight on September 24, 1936 and, following a brief spell with the A&AEE, was delivered to 38 Squadron and remained on strength until January 1939. *Via Martyn Chorlton*

GORDON I & II (IIIF MK V)



The III-series finally gets a name

- >>> MAR 3, 1931 K1697 flies from Harmondsworth
- >>> APR 1931
 Gordon joins
 40 Sqn
- >>> JAN 1933
 Final aircraft delivered
- >>> MAY 23, 1934
 Gordon II flies
 for first time
- >>> 1938
 RAF operational service ends
- >>> 1941

 RAF second line duties end

DEVELOPMENT

With the arrival of the more-powerful and lighter Panther IIA engine, the opportunity arose to re-engine the IIIF. However, rather than continue the series by making the latest Fairey design, the IIIF Mk V, it was deemed that there were sufficient changes from its predecessors to give the aircraft the name Gordon.

DESIGN

Fairey's own proposal to re-engine the IIIF resulted in Specification 18/30 which was simply described as a IIIF replacement, the Panther-engined Gordon being the only entrant. The new engine saw a marked improvement in performance, thanks to the radial being lighter and more fuel-efficient, the latter resulted in a 10% improvement in air mileage per gallon. The distance covered during take-off was reduced by 15% and the service ceiling rose by a further 10%.

The chosen airframe was IIIF Mk IV M/A K1697, which, while still on the production line was re-engined and given a general tidy up. Designated as the prototype Gordon, K1697 first took to the air from Harmondsworth on March 3, 1931 in the hands of Chris Staniland.

OPERATIONAL SERVICE

By the end of March, performance tests had already been carried out by the A&AEE and, in no time at all, an initial order of 28 Gordons, originally scheduled to be built as

IIIF Mk IVBs, was placed. The first twelve off the line were built as bombers, complete with a prone bomb aimer's position for service with 40 Squadron at Upper Heyford, from April 1931. The remaining 14 Gordons were shipped to the Middle East to serve with 6 Squadron at Ramleh.

A bigger order for 47 Gordons followed, allowing 35 Squadron at Bircham Newton to retire its IIIFs from July 1932. 14 Squadron at Ramleh followed in July 1932 and 4 FTS began receiving dual-control versions by the end of the year as well. The biggest and final order for 87 Gordons saw the type being delivered to the RAF between May 1932 and January 1933.

An improved version, the Gordon II, began arriving in the Middle East in 1934. The Mk II had a new fin and rudder with horn balance, a modified rear fuselage and Frise ailerons. The RNZAF would later receive 13 Mk IIs and a few Mk Is.

By September 1939, the Gordon was withdrawn from operational service but at least 40 examples remained on RAF strength, all serving in a secondary role until the middle of 1941.

PRODUCTION

162 Gordon Is were built in the serial ranges K1721-K1748, K2603-K2649 and K2683-K2769. 24 Gordon Ils were built, K3986-K4009 and 79 IIIFs were also converted in the serial range J9062-J9828, K1159-K1778 and S1178-S1203.



TECHNICAL DATA - GORDON I

ENGINE: One 525hp Armstrong Siddeley Panther IIA fourteencylinder two-row radial

WING SPAN: 45ft 9in

LENGTH: 36ft 9in

HEIGHT: 14ft 2in

WING AREA: 438 sq ft

EMPTY WEIGHT:

3,500lb

LOADED WEIGHT:

5,906lb

MAX SPEED: 145 mph

at 3,000ft

CRUISING SPEED:

110 mph

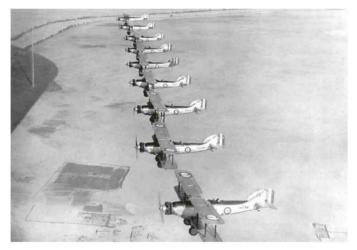
INITIAL CLIMB: 1,000

ft/min

SERVICE CEILING:

22,000ft

RANGE: 600 miles



An impressive nine-ship formation of 6 Squadron Fairey Gordons, operating out of Ismailia. Aeroplane



20 Gordons were sold to Brazilian Government, all but five of them were landplanes making this photo of 1-EB-4 a rare catch. *Via Martyn Chorlton*



6 Squadron flew the Gordon from June 1931 to October 1935 although K2617 pictured here served the unit for two years before being transferred to the Ismailia TT (Target-Tug) Flight. *Via Martyn Chorlton*

SEAL (IIIF Mk IV)



Patrolling the high seas

- >>> NOV 27, 1931
 Prototype flies
- >>> SEP 29, 1932
 Prototype flies with floats fitted
- >>> NOV 1932
 First production
 Seal delivered
 to FAA
- APR 1933
 444 Flt receives
 Seal
- >>> MAR 1935
 Last Seal delivered to FAA
- >>> APR 1937
 Seal withdrawn from service

DEVELOPMENT

Originally designated as the IIIF MK IV, the Seal was effectively the three-seat predecessor of the Gordon II. The Seal came about because of Specification 12/29 which covered the IIIF Mk VI prior to the Mk V, which, in turn became the Gordon.

DESIGN

The prototype, converted IIIF, S1325 did not take to the air until November 27, 1931. At this time, the Gordon I was already on the production line and further work was required before the Seal actually met 12/29. Once this was achieved, the Seal entered production in front of the Gordon II and it was a production Seal, K3577, which was later converted to become the prototype Mk II.

By early 1932, the first production order for eleven Seals was placed, the type featuring Frise ailerons, a fin with a bigger aspect ratio, an 'A' frame arrestor hook below the rear fuselage and a tail wheel in place of the long-serving skid.

OPERATIONAL SERVICE

By the end of November 1932, the first four production aircraft were ready for delivery to 444 (Fleet

Reconnaissance Catapult) and, by the end of the year, all were on board ships of the 2nd Battle Squadron in home seas.

821 Squadron was the first operational unit to receive the Seal at Gosport in April 1933. The unit embarked on HMS *Courageous* the following month for a two year tour of duty in home waters. However, by August 1935, the carrier was serving with the Mediterranean Fleet during the Abyssinian crisis.

In 1934, a second Seal unit, 820 Squadron, embarked on *Courageous* for a six-month tour of duty which saw the Seal replaced by the Blackburn Shark torpedo bomber by the end of it. The Fairey Seal went on to serve with eight FAA squadrons, although only four ever served at one time. Only 824 Squadron saw service east of the Suez when HMS *Hermes* arrived in Hong Kong in January 1935. The ship served as part of the China station for the next three years but the Seal did not; its replacement, the Swordfish took over in April 1937 while *Hermes* was visiting Seletar.

PRODUCTION

91 Seals built for the FAA in the serial ranges, K3477-K3487, K3514-K3545, K3575-K3579, K4201-K4225 and K4779-K4796. Six Seals were sold to Peru, four to Latvia, two to Chile and one to Argentina.



TECHNICAL DATA - SEAL LANDPLANE AND SEAPLANE

ENGINE: One 525hp Armstrong Siddeley Panther IIA fourteencylinder two-row radial

WING SPAN: 45ft 9in

LENGTH: (land) 36ft 9in; (sea) 35ft 4in

HEIGHT: (land) 12ft 9in;

(sea) 14ft 4in

WING AREA: 443.5 sq ft

LOADED WEIGHT:

(land) 6,000lb; (sea) 6,400lb

MAX SPEED: (land) 138 mph; (sea) 129 mph

CLIMB: 5,000 ft in 5.35 min

SERVICE CEILING:

(land) 17,000ft; (sea) 13,900ft

ENDURANCE: 4 to 5 hr



823 Squadron replaced its IIIFs with Gordons from December 1934 but only operated them until the arrival of the Swordfish in November 1936. Displaying the 823 Squadron code '802', K3543 only served the unit for a few months. *Via Martyn Chorlton*



Seal K3577 pictured at MAEE, Felixstowe, for trials in early 1934. The aircraft was also experimentally fitted with a Panther VI engine in a long-chord cowling and, later, converted to landplane configuration to become the prototype Gordon II. *Via Martyn Chorlton*



Seal K3522 pictured at Gosport in July 1933, prior to being allocated to the strength of C Flight at Lee-on-Solent. *Via Martyn Chorlton*

1934 FOX VI

The prototype Fairey Fox IV complete with canopy, twin cockpits and wheel spats and plenty of power from its Hispano-Suiza engine. This photograph was taken in September 1934. *Aeroplane*



TECHNICAL DATA FOX IV

ENGINE: (Proto) One 650hp Hispano-Suiza 12 Ybrs; (Production) One 860hp 12 Ydrs WING SPAN: 38ft LENGTH: 30ft 1in HEIGHT: 11ft LOADED WEIGHT: 4,950lb MAX SPEED: 227 mph CLIMB: 16,400ft in 6 min 30 sec

>>> JAN 31, 1934

First flight of prototype Mk VI

Demo by S H G Trower

>>> MAY 1940

Saw action during German invasion

The most successful Fox derivative

DEVELOPMENT

Basically a re-engined Fox IIIC, the Fox VI was created as a two-seat reconnaissance fighter which was destined to be licence-built in Belgium.

DESIGN

The Fox VI story only began as an experiment when Fox II (F.1753, AF.6031) was fitted with a 650hp Hispano-Suiza 12 Ybrs engine. The aircraft was first flown by Chris Staniland from Gosselies on January 31, 1934 and flight testing continued in Britain while the Fox VI was loaned to Fairey.

The aircraft had a fully enclosed canopy which covered both cockpits and a spatted undercarriage which, combined with the power of the Hispano-Suiza, helped the Fox VI to achieve a maximum speed of 227mph.

Once the Fox VI returned to Gosseilies, the engine was replaced by an 860hp 12 Ydrs engine and its wheel-spats were removed. By now, the experimental aircraft had turned into an official prototype. In August 1934, the aircraft was demonstrated by S H G Trower in front of a Belgian military mission at Gosselies and, not long after, was ordered into production.

SERVICE

The Fox VI served the Belgian Air Force in two variants, the Mk VIC and the Mk VIR, the final letter relating to combat and reconnaissance. On top of these two duties, the Fox VI was also employed as a day and night fighter and trainer, at least four aircraft being converted with dual-controls.

The Fox VI served with the following Belgian units from 1935 to 1940; 7/IV/1 'Méphisto' at Goetsenhoven and Lonzee, 5/III/2 'Aigle Bleu' and 6/III/2 'Aigle Rouge' at Nivelles and Vissenaken.

The only other operator of the Fox VI was the Swiss Government who purchased F.2246 and F.2247 for evaluation. They were subsequently given the military serial numbers 871 and 872 and finally the civil registrations of HB-HAF and HB-HAK.

PRODUCTION

Approximately 85 Fox VIs were built by Avions Fairey in the following serial ranges; F.2039 to F.2092 (54); F.2232 to F.2241 (10); F.2242 to F.2245 (4) all dual-controlled; F.2246 and F.2247 to Switzerland and F.2238 to F2263 (16).

G.4/31 1934



Replacing the Gordon and the Wapiti

DEVELOPMENT

When Specification G.4/31 was first issued in July 1931, virtually the entire British aircraft industry began work to design a new General-Purpose/Torpedo bomber for the RAF. The specification was aimed at a replacement for the Gordon and the Westland Wapiti, but the document's requirements even outweighed what these two capable aircraft had been achieving. The new aircraft had to be capable of bombing day or night, dive-bombing by day, army co-operation, aerial photography, general reconnaissance, casualty evacuation not to mention the ability to operate from make shift airfields in tropical conditions. It was clearly a tall order but this did not stop Blackburn, Bristol, Handley Page, Hawker, Parnall, Vickers, Westland and, of course, Fairey submitting designs and proposals for G.4/31.

DESIGN

The Fairey G.4/31, just like the majority of others bidding for the specification, was not an attractive aircraft. When it was first rolled out, the Pegasus II.M3 engine was without a cowl, the tail surfaces were untraditionally angular and a pair of unsightly longitudinal strakes was attached to the rear fuselage, in response to the loss of the TSR.I during a flat spin.

The G.4/31 had a portly appearance, allowing for a spacious cabin inside the forward fuselage. This was accessed via the rear cockpit or a large door on the starboard side. The pilot's cockpit was offset to port and each of the main planes had large cut-outs in their trailing edges, giving a good field of vision. Depending on the role the aircraft was carrying out, a second crew

member could take up a prone position inside the cabin for bomb aiming.

SERVICE

The Fairey G.4/31 was first flown by Chris Staniland on March 29, 1934. It was immediately apparent that the 635hp Pegasus engines were not delivering enough power to achieve the performance figures set out in the specification and, with a heavy heart, it was replaced by a 750hp Armstrong Siddeley Tiger IV. The heavier Tiger was still being developed and it was widely known that the engine was far from reliable.

Whilst the G.4/31 was re-engined, Fairey also took the opportunity to clean the aircraft up. The work including fitting wheel spats, re-designing the tail surfaces into a more rounded shape, the dispensing of a ventral fin, cowl around the engine and a Fairey metal propeller in place of the original wooden one. These modifications certainly made the aircraft more purposeful looking and the face lift saw the aircraft being unofficially referred to as the G.4/31 Mk II.

The Mk II carried out its first flight, once again with Staniland at the controls, on June 22, 1934. Almost immediately, the Tiger engine began to play and virtually every flight had to be brought to a premature conclusion because of overheating problems. Regardless, the Air Ministry were informed about the engine problems before the aircraft was serialled as K3905 and delivered to the A&AEE in January 1935. Sadly for Fairey, the engine let the aircraft down and not a single performance test was achieved. By February, K3905 was returned and later withdrawn from the battle to win Specification G.4/31.

TECHNICAL DATA G.4/31

ENGINE: On 635hp Bristol Pegasus II.M3 nine-cylinder air-cooled radial and later, one 750hp Armstrong Siddeley Tiger IV fourteen-cylinder two row radial WING SPAN: 53ft LENGTH: 40ft 10in HEIGHT: 15ft 8in WING AREA: 658 sq ft EMPTY WEIGHT: 6,987lb LOADED WEIGHT: 8,790lb MAX SPEED: (Bomber) 157 mph at 6,500ft; (Torpedo-bomber) 144 mph at 6,500ft CLIMB: (Bomber) 5,000ft in 7.2 min SERVICE CEILING: (Bomber) 23,200ft

MAR 29, 1934First flight of

Fairey G.4/31

>>> JUN 22, 1934
First flight of
Tiger-powered Mk II

MAR 23, 1939
K3905 adopted as
Tiger test-bed

S.9/30 & TSR.I



Where the story of the Swordfish began

- >>> JUN 1930 Specification S.9/30 issued
- >>> AUG 3, 1931
 Contract to
 proceed received
- >>> MAR 21, 1933 First flight of TSR.I
- >>> SEP11,1933 TSR.I crashes
- >>> FEB 22,1934
 First flight
 of \$1706
- >>> NOV 30, 1936 S1706 SOC

DEVELOPMENT

Its seems remarkable to think that an aircraft that was designed to one specification, led to another that was designed two specifications later, both machines making their maiden flights a mere eight weeks apart. That is what occurred with the S.9/30 which in turn evolved into the TSR.I and then the TSR.II, the latter also becoming known as the Swordfish.

What actually occurred was that the Fairey designers were looking way beyond the original Specification S.9/30 which was issued in June 1930. By the time the aircraft flew in 1934, the requirement had predictably passed in favour of S.15/33 but Fairey had already been working on their TSR.I, which was intended for service with the Greek Navy. The TSR.I was then put forward as a contender for S.15/33, but despite being destroyed in a crash, the design was carried on with the TSR.II and the FAA went on to receive its greatest ever torpedo-bomber.

DESIGN

The Fairey S.9/30, on the surface, was a two-bay biplane but was actually a single; the inboard inter-plane struts were only there to provide rigidity for when the wings were folded. A de-rated Kestrel IIMS engine powered the S.9/30, an engine that used evaporative cooling via surface-type steam condensers attached to the lower side of the upper wing. The fuselage was made of stainless-steel strip and tube and was constructed in four sections.

The TSR.I was a privately financed project powered by a 625hp Panther radial engine and, despite being designed specifically for the Greek Navy, no orders were

forthcoming. Prior to its bland designation name, the aircraft was referred to within the company walls as the 'Greek Machine'. Once the order fell through, the aircraft was re-engined with a 635hp Pegasus IIM engine cowled off with a Townend ring. The fixed undercarriage was trialled with and without spats, the rear fuselage was strengthened for deck operations and an arrestor hook was fitted.

SERVICE

The Fairey S.9/30 was first flown from Harmondsworth by Chris Staniland on February 22, 1934. Following its initial visit to the A&AEE, which was academic as the specification criteria had already been passed, the aircraft, serially S1706 was converted into a seaplane. Fitted with a large central float and smaller floats mounted on struts under the outer wings, S1706 was trialled at the MAEE, Felixstowe alongside the Hawker Osprey. The S.9/30 performed well, but the derivate of the all-conquering Hart family won the day and Osprey was chosen instead. S1706 was struck off charge on November 30, 1936 and ended its days being used for crash barrier trials.

The TSR.1, which remained un-serialled, was first flown by Staniland on March 21, 1933 in its original Greek Navy form. Accepted as a contender for Specification S.15/33, trials were progressing well until September 11, 1933 when Staniland failed to recover from a flat spin and was forced to abandon the aircraft after twelve rotations. Regardless, flight reports showed that the aircraft was more than capable to meet the specification and all the knowledge and experienced gained was ploughed into the TSR.II



TECHNICAL DATA - S.9/30 LAND AND FLOATPLANE AND TSR.I

ENGINE: (S.9/30) One 525hp Rolls-Royce Kestrel IIMS; (TSR.I) One 625hp Armstrong Siddeley Panther VI and later one 635hp Bristol Pegasus IIM

WING SPAN: (S.9/30 Land) 46ft; (folded) 17ft 10in

LENGTH: (land) 34ft 1in; (float) 39ft 3in

HEIGHT: (land) 14ft (float) 16ft 6in

WING AREA: 442 sq ft

LOADED WEIGHT:

(land) 5,740lb (float) 6,500lb

MAX SPEED:

(land) 147 mph at 2,000ft (float) 136 mph at

2,000ft

CLIMB: (land) 5,000ft in 5 min 30 sec; (float) 5,000ft in 6.3 min



The Marcel Lobelle designed S.9/30, was first flown as a landplane with a divided undercarriage and low-pressure tyres. *Via Martyn Chorlton*

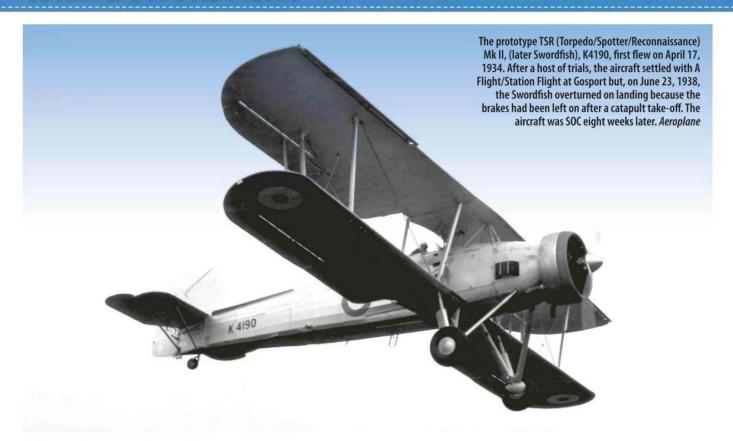


The TSR.I, also referred to by its construction number, F.1875, powered by a Armstrong Siddeley Panther engine. *Via Martyn Chorlton*



The privately funded TSR.I fitted with a Bristol Pegasus engine inside a Townend ring, driving a Watts propeller, wheel spats and an arrestor hook. *Via Martyn Chorlton*

TSR.II & SWORDFISH I



The 'Stringbag'

- » APR 17, 1934
 TSR.II first flight
- >>> DEC31, 1935
 First production
 aircraft flies
- >>> JUL 1936
 Swordfish joins the FAA
- >>> APR 1940
 First active
 service in Norway
- >>> AUG 1940 Swordfish joins the RAF
- >>> MAY 21, 1945
 Final operational sortie

DEVELOPMENT

As mentioned earlier, the roots of the TSR.II which would be renamed in the Swordfish are firmly planted in the story of the S.9/30 and the TSR.I. Following the loss of the latter in September 1933, Marcel Lobelle immediately set to work re-designing the TSR.I to a new more advanced specification.

DESIGN

Air Ministry Specification S.15/33 was for a Naval carrier-borne torpedo/spotter/reconnaissance (TSR) aircraft, which was a more advanced version of S.9/30. On the surface, the second aircraft, the TSR.II, was very similar to the TSR.I, only differing in having an extra bay in the fuselage and spin recovery strakes ahead of the tailplane (a lesson learned from the loss of the TSR.I).

The upper wing had a sweepback of 4° to compensate for the longer fuselage while other differences included a greater chord fin and rudder. Construction was generally similar to the TSR.I, the TSR.II having a pair of built-up steel-strip spars, duralumin ribs in the wings, steel drag struts and a steel-tube fuselage. Power was provided by a Pegasus IIM3 engine, cowled by substantial Townend-ring, driving a two-blade wooden Watts propeller, although later, a three-blade metal type became standard.

The TSR.II, serialled K4190, was first flown on April 17, 1934 by Chris Staniland and, two months later, was being trialled at Martlesham Heath, followed by the RAE for catapult trials, then on board HMS *Courageous* for deck landing tests.

By April 1935, the TSR.II had been renamed the Swordfish and a pre-production order for three aircraft, then a production batch of 86 aircraft followed. The first pre-production aircraft, K5660, was first flown on December 31, 1935.

OPERATIONAL SERVICE

The Swordfish entered service in July 1936 when it joined 825 Squadron on board HMS *Glorious*, the unit having previously operated the IIIF. By the end of the year, three more FAA squadrons had re-equipped with the Swordfish and, by late 1938, three more had re-equipped, the majority of them having replaced Blackburn Sharks and Fairey Seals. This left the Swordfish as the FAA's only torpedo-bomber until the arrival of the Fairey Albacore in March 1940.

By the beginning of the Second World War, 13 operational FAA squadrons were equipped with the Swordfish and a further twelve would be formed during it. On top of these, 25 operational squadrons and a further 22 second-line squadrons also operated the Swordfish and eleven catapult flights.

The Swordfish I served with great distinction in all theatres of the war but special mention should be made regarding the aircraft's involvement in the successful attack against Italian warships in Taranto harbour on November 11, 1940 and the demise of the *Bismarck* in May 1941.

The Swordfish I also served with RAF units, 8 and 202 Squadrons, between August and December 1940 and October 1940 and June 1941 respectively; the latter operating the seaplane variant.

The Swordfish outlived its operational replacement, the Albacore, by many years, mainly because of the larger aircraft's unsuitability to operate from escort carriers. It was while carrying out these duties aboard MAC ships that the Swordfish carried out its last operational duty on May 21, 1945 with 836 Squadron.

PRODUCTION

992 Swordfish Mk Is were built, 692 (201 of this number were delivered in 1937 alone) of them by Fairey at Hayes and 300 of them by Blackburn at Sherburn-in-Elmet.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA TSR.II & SWORDFISH I

ENGINE: One 690hp Bristol Pegasus IIIM3 nine-cylinder radial

WING SPAN: (TSR) 45ft 5in; (I) 45ft 6in

LENGTH: (TSR) 36ft 6in (I) 35ft 8in

HEIGHT: (I) 12ft 4in

WING AREA: (TSR) 542 sq ft; (I) 607 sq ft

EMPTY WEIGHT:

(I) 4,195lb

LOADED WEIGHT:

(I) 7,720lb

MAX SPEED: (I) 154 mph

CRUISING SPEED:

(I) 131 mph

SERVICE CEILING:

(I) 19,250ft

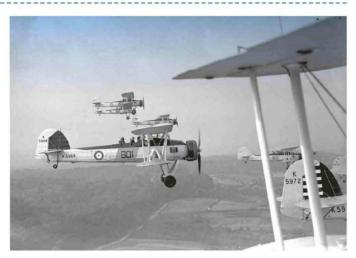
DURATION: 5.7 hr



K8440 takes the cable aboard HMS *Courageous* in early 1939. The Swordfish I first joined 811 Squadron on HMS *Furious* in March 1937 before joining 811 Squadron and was coded '609', as depicted here. *Aeroplane*



A rating tries his hand at being an air gunner in a Swordfish I at HMS *Kestrel* (aka Worthy Down) in 1940. It is impossible to identify the individual aircraft but it most likely belongs to 815 Squadron which was formed there from 811 and 822 Squadrons in October 1939 and left in May 1940. *Aeroplane*



823 Squadron airborne from HMS *Glorious* in early 1937, including K5968 '801', which joined the unit in November 1936. On April 18, 1938, the engine cut with Sub Lt D C Garton-Stone at the controls forcing the Swordfish to ditch into the sea. Garton-Stone was unhurt, but the aircraft was a total loss. *Aeroplane*



ENGINE: One 925hp Hispano-Suiza 12Yers twelve-cylinder vee liquid-cooled WING SPAN: 34ft 6in LENGTH: 27ft 7in HEIGHT: 11ft 4in WING AREA: 293 sq ft EMPTY WEIGHT: 2,500lb LOADED WEIGHT: 4,120lb MAX SPEED: 224 mph at sea level; 270 mph at 13,120ft LANDING SPEED: 60 mph CLIMB: 13,120ft in 5 min **CEILING: 36.080ft** DURATION: 2 hr at a

>>> JUN 6, 1935 First flight of Fantôme

cruising speed of 217 mph

>>> JUL 17, 1935 Test pilot S H G Trower killed

>>> NOV 1936 Two aircraft sold to

A still-born fighting aeroplane

DEVELOPMENT

The Fantôme, the Belgian-built version of which was known as the Féroce, came about because of a Belgian Government requirement for a new fighter to take over from the Firefly II. The criteria for this new aircraft were tough; a maximum speed of no less than 248 mph at 16,404ft, a duration of two hours at 80% power, a climb rate to 16,404ft in less than six minutes and a landing speed of 75 mph. It was a tough call but again Marcel Lobelle had produced an aircraft that was more than capable.

DESIGN

The Fantôme was an all-metal, single-bay biplane fighter with unequal-span staggered wings which were fabric covered. The undercarriage was semi-cantilever with spatted wheels and the ailerons were only fitted to the upper wing. Armament was a 20mm Oerlikon cannon, which could fire through the propeller hub. A pair of Browning machine-guns could also be fitted in the forward upper fuselage or wings if the 20mm was not installed.

SERVICE

The Fantôme, coded 'F-6', was first flown from the Great West aerodrome by Chris Staniland on June 6, 1935. Registered as G-ADIF, the aircraft was demonstrated at both the RAF and the SBAC display but while it was taking part in the competition at Evère for a new Belgian fighter, the aircraft inexplicably crashed in the landing circuit, taking the life of test pilot S H G Trower.

Spare parts and various assemblies were despatched

to Belgium by Fairey, who were planning on building three extra aircraft. By this time, the fighter, re-named Féroce, left the Avions Fairey factory at Gosselies, the Belgians had changed their fighter requirements. In the meantime though, the Soviets expressed an interest and, after being demonstrated by Staniland at Gosselies in November 1936, two were shipped, via Antwerp, to the Soviet Union.

A fourth aircraft was ordered by the Air Ministry on May 11, 1937 and, serialled L7045, was flown for the first time by F H Dixon from Gosselies on November 4, 1937 before being ferried back to Britain six days later. During December 1937 and January 1938, L7045 was flown from the Great West aerodrome to Farnborough where the main point of interest was the fighter's novel armament. All of the guns were remotely controlled, loading and cocking were performed pneumatically and compressedair was used for the trigger motors on the fuselage guns and the wing guns.

PRODUCTION

Four aircraft were built, the first, G-ADIF (F.2118), was built at Hayes; F.2264 and F.2265 were sold to the Soviet Union but were later sold to Spain to serve in the Republican Air Force where at least one of them was shot down. L7045 (F.3451) was built and delivered to Contract No. 613518/37 and following trials with the A&AEE and RAE, the aircraft was transferred to AGS (Air Gunnery School) Rollestone on December 12, 1940 and was not SOC until March 19, 1943.

Fairey Fox VII AF.6134, one of only two of this mark ever built was also known as the Mono-Fox and the Kangourou; a reference to the large pouch-like radiator under the fuselage. Via Martyn Chorlton



The Fox's final fling

DEVELOPMENT

The last two marks of the successful, in Belgian service at least, Fairey Fox family, were the Mk VII and VIII. While the former was little more than an experimental conversion, the latter, alongside its predecessors, briefly saw action during Germany's onslaught in May 1940.

DESIGN

The Fox VII evolved from the Mk VI but was designed as a single-seater which was merely a temporary conversion as it could be put back to its original two-seater arrangement within hours. Armament was up to a maximum of six guns, four in the upper wings and two in the fuselage.

As with the Fantôme, the Fox VII could be fitted with a 20mm Oerlikon *moteur cannon* if a Hispano-Suiza 12 Ycrs engine was installed instead of the two fuselagemounted machine-guns.

The Fox VII was also referred to as the Mono-Fox and, unofficially, as the Kangourou; the latter coming about because of the way the large ventral radiator resembled a Kangaroo's pouch.

The Fox VIII was also a modified and revised Mk VI, fitted with a Ratier three-blade propeller instead of a wooden two-blade one.

SERVICE

The first of only two Fox VIIs built first flew on December 14, 1935 from Gosselies with Belgian test pilot A J Eyskens at the controls.

The Fox VIII, of which twelve were ordered at the time of the Munich crisis in 1938, entered service in mid-1939, the last of them leaving the Gosselies production line on May 25.

The Fox VIII only served with 7/III/3, 7e Escadrille 'Flèche Ailée' at Evère during peacetime and Belcele when the Second World War began. It can only be presumed that the type did not serve beyond May 1940.

PRODUCTION

Two Fox VIIs were built and only referred to as AF.6134 and AF.6142 and twelve Fox VIIIs were built without Avions' Fairey construction numbers applied but they were serialled as O-182 to O-193 in Belgian Air Force service.

TECHNICAL DATA FOX VII & VIII

ENGINE: (VII) One 860hp Hispano-Suiza 12 Ydrs or Ycrs WING SPAN: 38ft LENGTH: 29ft 8in HEIGHT: 11ft MAX SPEED: 208 mph at

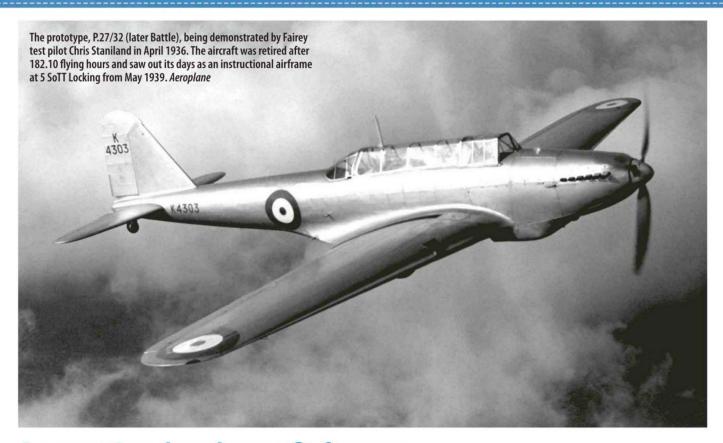
sea level

CLIMB: 16,400ft in 7 min CEILING: 37,720ft

>>> DEC 14, 1935 First flight of Fox VII

>>> SEP 1938
Fox VIII ordered

MAY 25, 1939
Last Fox built
leaves Gosselies



A peacetime bomber unfit for war

- >>> AUG 1932 Specification P.27/32 issued
- >>> SEP 1935
 First order for 155
 aircraft placed
- **MAR 10, 1936**First flight of P.27/32
- >>> MAY 1937
 Enters service
 with 63 Squadron
- WAY 12, 1940
 VC won by
 Fg Off D E Garland
 & Sgt T Grey
- >>> JUL 1941
 Withdrawn from
 RAF operational
 service

DEVELOPMENT

When the specification for a new light bomber was issued in 1932, the world was a comparatively peaceful place. But as the 1930s progressed, it became clear that aircraft development and tactics were both rapidly advancing and improving at a pace much quicker than the Fairey Battle. Adequate when it entered service in 1937, it was hopelessly outclassed by the beginning of the Second World War and was equally let down by being employed in the wrong theatre of action and with outdated tactics.

The aircraft was created in response to Air Ministry Specification P.27/32, originally issued in August 1932 but not confirmed until April 1933. The remit was that the aircraft should in service by 1936, be able to a carry a bomb load of no less than 1,000lb over a range of 1,000miles and have a speed of not less than 200mph.

DESIGN

Designed by Marcel Lobelle, the P.27/32 was designed to accommodate the PV.12 engine (the prototype Rolls-Royce Merlin) which was not available until the spring of 1934. The wing, a cantilever design built in five sections, housed the retractable undercarriage and a pair of bomb traps in each side for up to four 250lb bombs. The centre section of the wing was integral to the fuselage while aft of the pilot's cockpit construction was semi-monocoque. Lobelle presented his design to the Air Ministry on June 11, 1934 and, seemingly impressed with what they saw, a contract for a single prototype was awarded.

It was not until November 1935 that Rolls-Royce delivered a Merlin C engine to Fairey although by the time Chris Staniland flew the prototype P.27/32, K4303 on March 10, 1936, the powerplant was a 970hp Merlin G. However, senior staff was so impressed with the aircraft, even before its first flight, an order for 155 aircraft was placed in September 1935. Now named the Battle, it was

with this 'G' engine that K4303 was delivered to the A&AEE in October 1936 where all the performance figures as per Specification were achieved.

OPERATIONAL SERVICE

Because of Roll-Royce's pre-occupation with refining the Merlin engine for the Hurricane and Spitfire, further delays meant that the Battle did not enter RAF service until May 1937. The first recipient was 63 Squadron at Upwood followed by 105 Squadron at Harwell and, by the end of the year, five squadrons had converted to the Battle.

These early arrivals were all powered by the Merlin I which proved a troublesome engine and, after the 136th Battle was built, the unit was changed for the Merlin II which caused a great deal conflict with Hurricane production at the time.

By the beginning of the Second World War, 15 squadrons were equipped with the Battle and, as part of the AASF in support of the BEF, 12, 15, 40, 88, 103, 142, 150, 218 and 226 Squadrons were all sent to France. The type's baptism of fire came during May 1940 when the light bomber was employed on a host of near suicidal low-level daylight attacks. One of these famous raids was on May 12 against the Meuse Bridge, Maastricht, when four out of five Battles from 12 Squadron failed to return, one of the crews being posthumously awarded the RAF's first Victoria Crosses for the determined attack.

Virtually relegated to second line duties on their return to England, the Battle saw out its operational days with 98 Squadron in July 1941.

PRODUCTION

2,200 Battles were built in total, 1,171 of them by Fairey at Hayes and Heaton Chapel and 1,029 by Austin. 16 were supplied to Belgium and, out of the grand total, 739 were shipped to Canada and 364 to Australia.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA BATTLE

ENGINE: One 1,030hp Rolls-Royce Merlin I twelve-cylinder liquidcooled in-line and later Merlin II, III and V

WING SPAN: 54ft

LENGTH: 42ft 4in

HEIGHT: 15ft 6in

WING AREA: 422 sq ft

EMPTY WEIGHT: 6,647lb

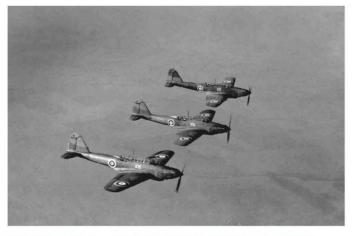
LOADED WEIGHT: 10,792lb

MAX SPEED: 210 mph at sea level; 257 mph at 15,000ft

CLIMB: 5,000ft in 4 min 6 sec

CEILING: 25,000ft

RANGE: 1,000 miles at 200 mph and 16,000ft



105 Squadron replaced its Hawker Audaxes with Battles from August 1937. The squadron, like so many Battle squadrons, suffered terribly in France and, in May 1940, was re-equipped with the Blenheim IV. *Aeroplane*



A large number of Battles were converted to target tugs, including ex-Battle I, L5664, pictured here during trials with the A&AEE on December 8, 1941. The aircraft was later shipped to South Africa in June 1942. *Via Martyn Chorlton*



Crews from 226 Squadron, stationed at Harwell, prepare for a peacetime navigational exercise while Battle K7620 warms its Merlin through beyond. Transferred to 35 Squadron, K7620 never saw action and was wrecked at Jurby while serving with 5 AOS in November 1939. *Aeroplane*

SEAFOX



A restricted specification

- >>> JAN 1936
 Production
 order placed
- **MAY 27,1936**First flight
 of K4304
- >>> MAR 1937
 Trials with
 HMS Neptune
- >>> APR 23,1937
 First production
 Seafox flies
- DEC 1939

 Battle of the River Plate
- >>> 1943 Seafox retired

DEVELOPMENT

The Fairey Seafox was specifically designed to be launched from Royal Navy light cruisers and, as such, was designed to a very stringent specification. The specification, S.11/32, called for a two-seat reconnaissance floatplane which would inevitably result in a lightly-loaded, conventional aircraft.

DESIGN

Of all-metal construction with fabric covered wings (which could be folded), the Seafox had a monocoque fuselage which was made up of 'Z' sections covered in Alclad panelling. The pilot's cockpit was open while the observer/gunner's station was covered in a transparent hood. The reason for the open cockpit was that the pilot had a better view during a catapult launch and, more importantly, a good view during recovery back to the ship. The enclosed canopy could be raised from the rear, allowing sufficient room for a Lewis machine-gun which was fitted on a Fairey high-speed mount.

Originally designed to take a 500hp Bristol Aquila radial which would have given the Seafox spirited performance, the unique 393hp Napier Rapier was selected instead. This one-off engine in military service initially suffered from excessive oil consumption and high cylinder head temperatures. These faults were eventually cured but performance of the Seafox suffered as a result.

The first of two prototypes, K4304, made its maiden flight from Hamble on May 27, 1936 and the second, K4305, which was built as a landplane, first flew on November 5, 1936, also by Hamble.

OPERATIONAL SERVICE

After the Seafox had completed many trials and catapult tests with the RAE, and further trials with HMS *Neptune* off Gibraltar in March 1937, the Seafox entered service. The type served with 702, 713, 714, 716 and 718 Catapult Flights (all of these were later pooled into 700 Squadron from January 1940) and 703, 754, 764, 765 and 773 Squadrons.

The Seafox had an active service career up to its retirement from the FAA in 1943, the most distinguished action being its involvement in the Battle of the River Plate in December 1939. After an engagement with the Admiral Graf Spee, the Walruses on board HMS Exeter were damaged, leaving only one Seafox available on board HMS Ajax; a second Seafox had been destroyed. Flown by Lt E D G Lewin, with observer Lt R E N Kearney, the Seafox acted as a spotter for the guns of HMS Ajax, Achilles and Exeter throughout the action and, on December 17, were the first to report that Graf Spee had been scuttled off Montevideo. Lewin was later awarded the DFC for his part in the sinking of the Graf Spee; the first FAA officer to be decorated during the Second World War.

PRODUCTION

66 Seafoxes were built in 1936 and 1937, including two prototypes, K4304 and K4305, the first production batch of 49 aircraft was placed in January 1936 serialled K8569 to K8617 and a final repeat order of 15 aircraft ordered in September 1936, serialled L4519 to L4533.



TECHNICAL DATA SEAFOX

ENGINE: One 395hp Napier Rapier VI sixteen-cylinder–H air-cooled

WING SPAN: 40ft

LENGTH: 33ft 5in

HEIGHT: 12ft 2in

WING AREA: 434 sq ft

EMPTY WEIGHT:

3,805lb

LOADED WEIGHT:

5,420lb

MAX SPEED:

124 mph at 5,860ft

CRUISING SPEED:

106 mph

CLIMB: 5,000ft in

15 min 30 sec

CEILING: 9,700ft

ENDURANCE: 4.25 hr

RANGE: 440 miles



With its Napier Rapier engine warming through, the prototype Seafox, K4304, is being manhandled down the slipway at Felixstowe. Note the Short Mayo Composite in the background. *Aeroplane*



A Fairey Seafox of 765 Squadron pictured being recovered at Sandbanks, Poole Harbour in 1942. The Seafox served the unit from May 1939 to June 1942. Via Martyn Chorlton



Seafox K8577 of 716 Flight on board the Leander-class light cruiser, HMS *Neptune*, pictured before the ship left for South Africa in September 1937. The Seafox, which was being flown by Lt D H Burke and Lt J Roe, was credited with spotting the German merchant ship *Inn* which was sunk by gunfire on September 5, 1939. *Aeroplane*



Turning a day bomber into an all-weather fighter

- »» NOV 12, 1934 P.4/34 issued
- **JAN 13,1937** First P.4/34 flies
- **APR 19, 1937**Second P.4/34 flies
- >>> 1939
 Licence acquired
 by Denmark
- >>> 1941 K7555 tests Fairey-Youngman flaps
- **MAR 18, 1948** K7555 SOC

DEVELOPMENT

The aircraft that would later be known as the Fulmar was originally designed as a day bomber to Specification P.4/34, from which the two prototypes were named. Marcel Lobelle designed an aircraft that was very clean and very fast for a day bomber the machine inadvertently became the main contender for new Naval Specification O.8/38 for a carrier-borne fighter and observation aircraft. The caveat to this specification was that it would have to be a two-seater because navigational aids were still lacking in the late 1930s which were efficient enough to get a lone pilot back to an aircraft carrier in open seas.

DESIGN

Issued on November 12, 1934, Specification P.4/34 was also contended by Hawker's Henley, while Lobelle based Fairey's entrant on the Battle. However, the Fairey P.4/34 was lighter, smaller and improvements included an undercarriage which folded inwards and, unlike the Battle, retracted completely into the wing. The P.4/34 was also sufficiently stressed for dive-bombing duties, a level of strength that would stand it in good stead as a fighter in the future. Compared to the Henley, the Fairey P.3/34 carried its bomb-load externally which was one of the main reasons why the Hawker aircraft won the original specification. However, the day-bomber requirement was later abandoned, leaving the Henley to serve out its days as a target-tug while the P.4/34, in the guise of the Fulmar, was destined for a more actionpacked career.

SERVICE

The first of two prototypes, K5099 (c/n F.2231), was first flown on January 13, 1937, followed by the second P.4/24, K7555, on April 19, both aircraft by flown Chris Staniland. The latter would basically evolve as a flying mock-up of the Fulmar fleet-fighter from March 1938. Some of the modifications carried out to meet the specification included a reduction of eight inches of the wing span and repositioning of the tailplane eight inches higher.

K7555 was then delivered to Martlesham Heath, where it was tested during September and October 1937. The report was generally good but the P.4/34 was let down by its stalling characteristics and heavy rudder control above 80 mph. Following further modifications, the aircraft's behaviour during the stall was described as benign and the rudder control was rectified by a bias control.

The P.4/34 proved a useful test-bed for other Fairey types as well, including the Firefly. K7555 was fitted with Fairey-Youngman flaps which were tested by the RAE during 1941 and the future Fireflies' four-cannon installation was also trialled by the P.4/34. K7555 was also used to test pneumatically-operated bellow-type air brakes.

The first aircraft, K5099, also carried out its fair share of performance trials with the A&AEE and the RAE. It was with the latter that the aircraft was used for 'survival' test in the event of a collision with barrage-balloon cables during mid-1937, followed by a variety of 'defence experiments'. The aircraft saw out its days as instructional airframe No.3665M. K7555 continued to serve Fairey and the RAE throughout the war and was not SOC until March 18, 1945.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA P.4/34

ENGINE: One 1,030hp Rolls-Royce Merlin I vee-twelve liquid

cooled

WING SPAN: 46ft 4½in

LENGTH: 40ft

HEIGHT: 14ft 1in

WING AREA: 342 sq ft

EMPTY WEIGHT:

6,405lb

LOADED WEIGHT:

8,787lb

MAX SPEED:

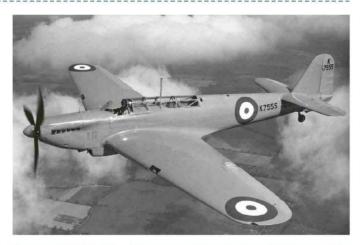
Merlin II) 283mph at 15,000ft

MAX RATE OF CLIMB:

1,200 ft/min

CEILING: 26,600ft

RANGE: 1,000 miles



The second P.4/34, K7555, which gave valuable service, not only in support of the Fulmar, but also the Firefly; the aircraft served from April 1937 through to early 1945. *Aeroplane*

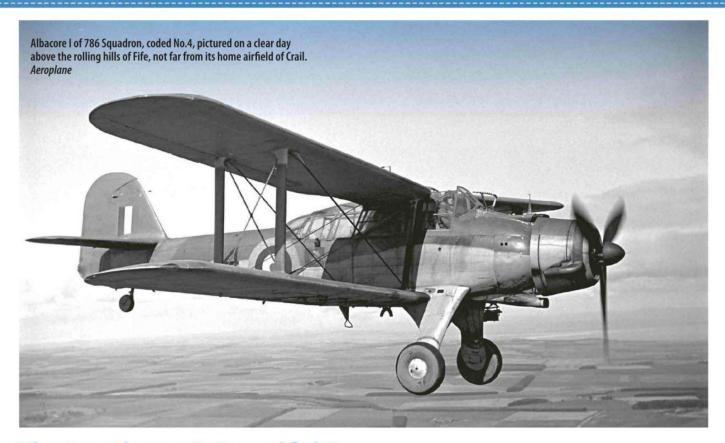


One of the more obvious differences between the Battle and the P.4/34 was the wide-track undercarriage which retracted inwards instead of backwards and was completely concealed in a neat recess in the underside of the wing. *Aeroplane*



Fairey ground-crew use a hand crank to the start the Rolls-Royce Merlin I engine of P.4/34, K5099, with Chris Staniland in the cockpit. *Aeroplane*

ALBACORE



The 'gentleman's Swordfish'

- >>> SEP 8, 1936
 Original specification issued
- >>> DEC12,1938
 Prototype flies
 from Great West
- >> 1940
 Tested by the A&AEE
- >>> JUL 9, 1942
 Tobruk convoy
 attacks
- >>> 1942 Replaced by Barracuda
- >>> 1949
 Retired from CAF

DEVELOPMENT

It is common knowledge that the Swordfish, which the Albacore was meant to replace, continued in service much longer than planned. The more refined Albacore was a classic example of an obsolete idea being over-developed. Refinements over its predecessor included a fully enclosed centrally-heated cockpit, a windscreen wiper, a laboratory system, an automatic dinghy launching system, variable pitch propeller and hydraulically-operated flaps. While this did make the Albacore a considerably more comfortable and efficient aircraft compared to the Swordfish, the concept had already been passed and had been successfully achieved by the older open-cockpit Fairey design.

DESIGN

The Albacore was created from Specification M.7/36, dated September, 1936, which called for a torpedo/bomber/reconnaissance aircraft with a speed range of between 58 and 183kts and the ability to carry an 18in Mk XIIA torpedo. Other criteria were dual-controls, a power-operated rear turret, full navigation and observation facilities plus sufficient heating and sound-proofing.

M.7/36 was actually abandoned by the Air Ministry but Fairey had already thrown themselves into the specification and a new one was created, S.41/36, specifically for a Swordfish replacement and the Albacore was the only competitor.

Designer Marcel Lobelle presented the ministry with two layouts to S.41/36, one a monoplane and the other a biplane. The monoplane was dismissed, so the Albacore became an equal-span, single-bay, folding wing all-metal biplane complete with a monocoque fuselage, fabric-covered wings, an enclosed twin cockpit and a heavily faired split undercarriage. Hydraulically-operated flaps doubled as air-brakes when the aircraft

was used for dive-bombing.

A Taurus II engine was initially fitted, later to be replaced by the more powerful Taurus XII. Armament was a single forward-firing machine gun in the upper starboard wing and one or two Vickers K machine guns in the rear cockpit. As per the original specification, the Albacore could carry a single 18in 1,600lb torpedo or up to four 500lb bombs carried on racks under each lower wing.

OPERATIONAL SERVICE

The prototype, L7074, was first flown from the Great West aerodrome by F H Dixon on December 18, 1939. This aircraft, and the second prototype, L7075, were not individually ordered but were the first pair from an initial production order of 100 aircraft.

826 Squadron, which was formed at RNAS Ford on March 15, 1940, was the first unit to receive the Albacore. This unit was also the first to take the type into action during the German invasion of the Low Countries, the Albacores attacking E-boats off Zeebrugge and communication lines at Westende.

826 and 829 Squadrons were the first Albacore units to embark on an aircraft carrier when both joined HMS *Formidable* on November 26, 1940. Both of these units went on to serve with distinction in the Battle of Cape Matapan in March 1941.

By 1942 and into mid-1943, the Albacore was already being superseded by another Fairey dive-bomber, the Barracuda, and, by November 1943, only 820 and 841 Squadrons were still operating the type.

PRODUCTION

All 800 Albacores were built at Hayes between 1938 and 1942. The type not only served with 46 FAA squadrons (16 of these were operational units), but also the RAF, the RCAF and RCN.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA ALBACORE TORPEDO-BOMBER & RECONNAISSANCE

ENGINE: One 1,065hp Bristol Taurus II fourteencylinder two-row sleevevalve radial and later one 1,130hp Taurus XII

WING SPAN: 50ft

LENGTH: (landplane) 39ft 10in; (seaplane) 42ft 5½in

HEIGHT: (landplane) 14ft 2in; (seaplane) 17ft 9in

WING AREA: 623 sq ft

EMPTY WEIGHT: (TB) 7,250lb; (Recce) 7,200lb

LOADED WEIGHT:

(TB) 10,460lb (Recce) 9,615lb

MAX SPEED: (TB) 161 mph at 4,500ft; (Recce) 169 mph at 4,500ft

CRUISING SPEED: 116 mph

CLIMB: 6,000ft in 8 min

CEILING: 20,700ft

RANGE: 930 miles with a 1,610lb war load



An Albacore being loaded with 250lb bombs on board HMS Formidable during Operation Torch in November 1942. Via Martyn Chorlton

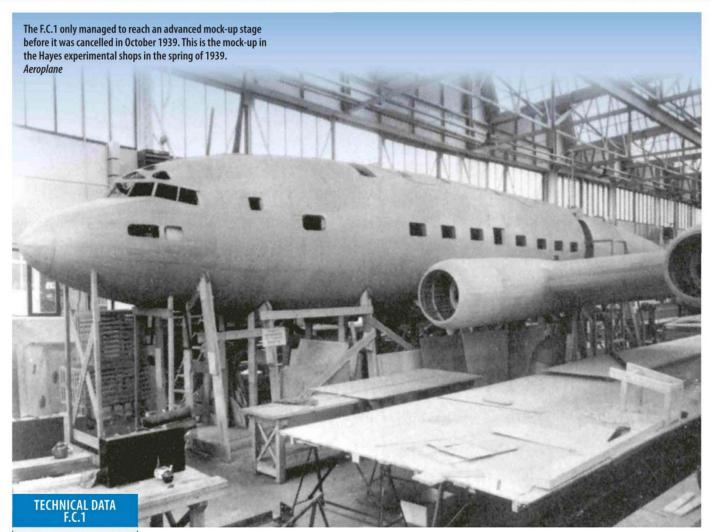


A fine display of torpedoes, complete with their trolleys and hydraulic jacks, which made light work of raising the weapon up to the crutches on the underside of the fuselage. This is Crail and the Albacores in the background belong to 785 and 786 Squadrons. *Via the late R C Sturtivant*



A Crail-based Albacore drops a practice torpedo in the Firth of Forth in 1942. Note the freed wire, which can be seen swinging between the undercarriage legs. *Via Ian Templar*

1938 F.C.1



ENGINE: (prototype) Four 1,000hp Bristol Taurus fourteen-cylinder sleeve-valve radial; (production) Four 1,200hp Rolls-Royce Exe twenty-four cylinder X sleeve-valve pressure air-cooled WING SPAN: 105ft LENGTH: 82ft WING AREA: 1,300 sq ft LOADED WEIGHT: (short-haul) 42,000lb MAX SPEED: 275 mph at 13,000ft STALLING SPEED: 70 mph MAX RANGE IN STILL AIR: 1,700 miles at 50% rated power

- >>> JUL 4, 1938
 Fairey submit proposal
- >>> NOV 12, 1938
 Fairey win contract
- >>> OCT 17, 1939
 The F.C.1 was
 cancelled

Keeping pace with the Americans

DEVELOPMENT

Concern was raised in 1938 by the Directorate of Civil Aviation about how far Britain was lagging behind in the development and production of civilian landplanes. The Douglas DC-2 and DC-3 were already beginning to dominate the world market and, after many meetings, discussions and study groups, it was decided that financial aid was the only way to kick-start manufacturers into producing such aircraft.

At the same time, a pair of specifications was issued; 15/28 for a short to medium-haul aircraft in May 1938 and 14/38 for a long-haul transport, the latter being made public in July. Short Brothers won the long-haul contract while 14/38 saw a more hotly contested aircraft with Armstrong Whitworth, Bristol, Fairey, General Aircraft and Vickers all receiving requests to tender. Later Folland also became involved but, after much negotiation, only Fairey remained - the only manufacturer who had not been seriously involved in civilian aircraft before.

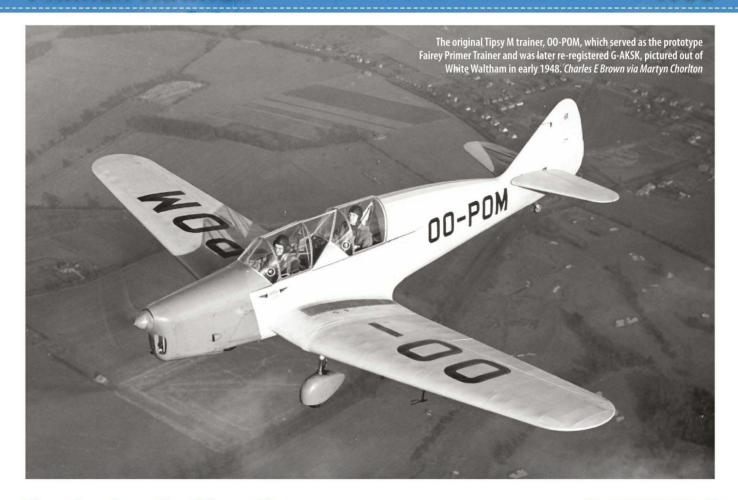
DESIGN

By October 1938, Fairey estimated that the cost of a pair of prototypes, minus engines, would be £385,000 and if twelve aircraft were ordered into production an estimated maximum cost of £80,000 each was quoted. Once treasury approval was received, Fairey formally announced that it would start work, the F.C.1 (Fairey

Commercial Number One) which, following an agreement of November 12, 1938, would serve with British Airways.

The F.C.1 was designed as a four-engine aircraft, mainly because the 15/38 requirements for take-off distance meant that there were not two engines available at the time with sufficient power should one fail. The aircraft was to have a pressurised cabin capable of carrying 26 passengers (plus four more occasional) with a crew of five. The aircraft had a tricycle undercarriage and, to give it excellent short-field landing and take-off performance, the F.C.1 was fitted with Fairey-Youngman flaps. Power was to be provided by four 1,000hp Taurus on the prototype and the untried 1,200hp Exe (Boreas) for the production.

By the spring of 1939, a very complex mock-up had been built and an equally detailed model was being wind tunnel tested, by which time, an official production order had been placed. Unfortunately, the Second World War got in the way and, on October 17, 1939, the F.C.1 was cancelled and one month later Specification 14/38 was also cancelled. Following the end of the Second World War it is alleged that Fairey seriously considered reviving the F.C.1, taking advantage of their wartime production lines and a large number of employees. Planned with Bristol Hercules engines, up to three prototypes were planned with sufficient tooling to produce up to 200 aircraft.



Continuing the Tipsy line

DEVELOPMENT

Prior to the beginning of the Second World War, the Avion's Fairey designer, Ernest O Tips, had created the moderately successful range of Tipsy aircraft. This included the Tipsy S single-seater and the Tipsy B two-seater, the latter being developed during the post-war period.

Another Tipsy machine was the Primary Trainer, or Tipsy M, later renamed the Primer Trainer, which was taken over by Fairey in 1948 with the intention of beginning limited production at Hamble. The origins of this aircraft were in the prototype, OO-POM, originally designated as the Tipsy M in 1938. This aircraft was dismantled and secretly shipped to England prior to the German invasion of Belgium, although the aircraft had previously been tested by Fairey pilots in June 1939. Once re-assembled, the Tipsy M was recorded as flying from the Great West aerodrome between November 1940 and May 1941 and again until September as a company communications aircraft. Dismantled and stored again, the aircraft would not appear again until the war was over.

DESIGN

A low-wing monoplane, the Fairey Primer Trainer, had a fully enclosed tandem cockpit and was powered by a Gipsy Major 10 engine. The aircraft had manually operated flaps and a fixed undercarriage which, on OO-POM (by now re-registered as G-AKSK but flown displaying 'G-6-1') was spatted and fitted with wheel brakes. The fuselage and sections of the wing were constructed from bronze-welded metal tubes backed up by the subsidiary components made of wood, such as the stringers and ribs. The Primer was fabric covered.

SERVICE

As soon as the war was over, OO-POM was returned to Avions Fairey, where some minor modifications were carried out before the aircraft was taken over by Fairey in 1947. Ex-OO-POM was registered as G-AKSK in February 1948 and, not long after, was delivered to the A&AEE at Boscombe Down for flight testing and various assessments. The aircraft was then dismantled again at Hayes so that production drawings could be produced and a set of jigs were designed and assembled. The original drawings and technical information had been deliberately destroyed by staff of Avions Fairey at Gosselies in 1940.

The first production aircraft, registered as G-ALBL 'G-6-4', was made up of parts of G-AKSK, including the Major engine. G-ALBL was certificated on October 22, 1948 and, during its brief flying career, the Major was replaced by a Cirrus Major only for the aircraft to be dismantled in 1949. The second production aircraft, G-ALEW 'G-6-5', was fitted with the 155hp Cirrus Major engine and was put forward as a primary trainer for the RAF in competition with the de Havilland Canada Chipmunk. Having failed to secure this potentially huge military contract, the last Primer was dismantled in 1951.

G-AKSK is recorded as having been sold abroad in August 1948, but to where, and whether the Primer was in an airworthy condition, is not known.

PRODUCTION

Three aircraft, ex OO-POM, re-registered as G-AKSK; G-ALBL (c/n F.8455) and G-ALEW (c/n F.8456).

TECHNICAL DATA PRIMER

ENGINE: One 145hp Gipsy Major 10 four-cylinder inline air-cooled; or a 155hp Blackburn Cirrus Major 3 WING SPAN: 32ft 10in LENGTH: 27ft 6in HEIGHT: 6ft 10in WING AREA: 154.5 sq ft EMPTY: 1,360lb LOADED WEIGHT: 1,960lb MAX SPEED: 134 mph at sea level CRUISING SPEED: 122 mph STALLING SPEED: (flaps down) 51 mph **CEILING: 19,500ft CRUISING RANGE: 383 miles**

» 1938

Tipsy M flies for the first time

>> 1940

OO-POM shipped to England

>> 1951

Last Primer dismantled

FULMAR I & II



The FAA's big eight-gunned fleet-fighter

- >>> JAN 4, 1940

 First production

 Mk I aircraft flies
- >>> JAN 20, 1941
 First flight of the Mk II
- >>> MAY 1941 Shadowed the Bismarck
- FEB 8, 1945
 Fulmar retired
 from FAA
- SEP 8, 1962
 G-AIBE at the
 SBAC
- DEC 18, 1962
 G-AIBE finally
 retired

DEVELOPMENT

The Fulmar obviously hit the ground running because of all the valuable work already achieved by the P.4/34 and as no further prototype was needed. The first Fulmar to fly was a production aircraft from an initial batch of 40 aircraft of which two of this group were planned as seaplanes. However, by March 1939, a decision to use the Fulmar in this role was dropped. The actual name Fulmar was not applied to the aircraft until August 1938 and the type was not officially made public until after it had joined the FAA in June 1940.

The very first production Fulmar, N1854, was flown for the first time by Duncan Menzies from Ringway on January 4, 1940. After being sent to the A&AEE, along with N1855 and N1858, and then to HMS *Illustrious* for further trials, N1854 was retained by Fairey for development work. Later converted to Mk II standard with a Merlin 30 engine, the aircraft remained airworthy into the post-war period and was re-registered as G-AIBE. The Fulmar served as a hack and general communications aircraft until it was finally retired on December 18, 1962 and today is preserved in the FAA Museum at Yeovilton.

The Fulmar II was fitted with the Merlin 30 engine and, as a result, several modifications had to be made. These included a Rotol propeller, new radiator, oil cooler, a revised sump for the fuel tank, bigger pipes for the fuel system, trimmer stops and a rudder mass-balance. The first production Fulmar II, N4021, made its maiden

flight from Ringway on January 20, 1941.

OPERATIONAL SERVICE

The Fulmar first went into action in September 1940 during the protection of the Malta convoys. The opposition at the time was the Italian Air Force and the combined efforts of 806, 807 and 808 Squadrons sent many enemy aircraft plunging into the Mediterranean. The Fulmars of 806 Squadron were particularly successful and, during November alone, shot down six more enemy aircraft whilst providing fighter cover for the Swordfish operation at Taranto.

The Fulmar also played a vital role during the Petsamo attack in July 1941, and convoy protection for the Russian convoys, the siege of Malta and both the North African and subsequent Italian invasions. The type also helped to shadow the *Bismarck* in May 1941 and was used for night-intruder operations as the NF.II , night-fighter training and even target towing as the TT.II.

The Fulmar served with a total of 49 FAA squadrons (and one RAF unit, 273 Squadron), 18 of these were operational units on board eight fleet aircraft carriers and five escort carriers until February 8, 1945.

PRODUCTION

A total of 600 Fulmar Is and IIs were built; 250 were Mk Is and 350 were Mk IIs. The Fulmar was produced between 1940 and 1943; peak production was 319 aircraft in 1941.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA FULMAR I & II

ENGINE: One 1,080hp Merlin VIII twelve-cylinder liquid-cooled and later one 1,300hp Merlin 30

WING SPAN: 46ft 41/2in

LENGTH: 40ft 3in

HEIGHT: 14ft

WING AREA: 342 sq ft

EMPTY WEIGHT: (I) 6,915lb; (II) 7,015lb

LOADED WEIGHT:

(I) 9,800lb; (II) 9,672lb

MAX SPEED: (I) 280 mph (II) 272 mph

CRUISING SPEED:

235 mph

MAX RATE OF CLIMB:

1,200 ft/min

CEILING: (I) 26,000ft

(II) 27,200ft

RANGE: (I) 800 miles

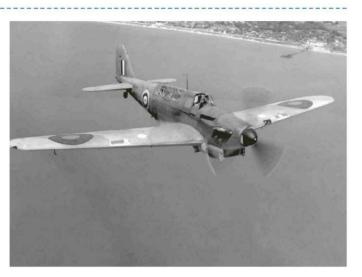
(II) 780 miles



Fulmar I, N1858, the fifth production aircraft (c/n F.3711), like the second P.4/34 before, was used by Fairey to test Youngman and double-split flaps. *Aeroplane*



The Fairey Fulmar entered FAA service in 1940 and, despite being overtaken by the better equipped and more powerful single-seat machines from 1943 onwards, the type remained operational until early 1945. *Charles E Brown via Martyn Chorlton*



The first production Merlin 30-powered Fulmar II, N4021, which first flew in January 1941. This photograph was taken on June 2, 1942 whilst undergoing testing with the A&AEE at Boscombe Down. *Aeroplane*

BARRACUDA I



Re-introducing dive-bombing to the FAA

- >>> NOV 9, 1937 Specification S.24/27 is issued
- >>> JAN 30, 1939
 Contract for
 two prototypes
 placed
- >>> FEB 2, 1939
 Ministry approval of mock-up
- >>> JUN 1940 Undercarriage tested by Battle K9370
- >>> DEC 7, 1940
 First prototype
- » AUG 1944

 Barracuda I is retired from FAA service

DEVELOPMENT

Another creation from the drawing board of Marcel Lobelle, the Barracuda was designed to replace the Albacore in the torpedo carrying role but by the time the aircraft entered service, the 'mighty metal monster' (as dubbed by the FAA) also brought the art of dive bombing back into FAA operations.

A continuation of the Battle, Fulmar and Firefly train of thought, the Fairey Barracuda was designed, built and flown in the space of 20 months but was stalled when demand for its powerplant, the Rolls-Royce Merlin, was being diverted to the Spitfire and Hurricane.

Designed to Specification S.24/27 dating from November 1937, a contract for two prototypes was not placed until January 1939 and, by August, a production contract was in place.

DESIGN

The Barracuda had a high set shoulder wing, all-metal stressed skin fuselage and a long canopy which enclosed the pilots, observer/navigators and wireless operator air/gunners positions. Because of the shoulder wing, the undercarriage was hinged at the base of the fuselage, retracting outboard into the wing roots.

Another novel feature of the aircraft was the fitment of Fairey Youngman flaps mounted above and below the trailing edge of the wing. These provided lift on take-off, a larger area wing in flight and drag during landing as well as providing stability during a dive bombing attack. The first prototype had a conventional tailplane but, during early test flights, it was found that when the Youngman flaps were set at a negative angle (during a 30° dive bombing attack), the wake from them caused severe tailplane buffet. The solution was to mount the tailplane high on the fin in a 'T-tail' arrangement which was strut braced for added strength.

Power was originally intended to be the Rolls-Royce 1,200hp Exe, but when development was abandoned, the 1,300hp Merlin 30 was chosen instead.

OPERATIONAL SERVICE

The first aircraft, P1767, was flown on its maiden flight by Chris Staniland from the Great West aerodrome on December 7, 1940 followed by the second Barracuda, P1770 on June 29, 1941.

The Barracuda I entered service with 785, 786 Squadrons at Crail and 831 Squadron at Lee-on-Solent in December 1942. The Mk I also served with 747, 767, 768, 778, 787 and 827 Squadrons but with regard to the two operational units, the Barracuda I never saw any action. It remained in FAA service until August 1944, when it was withdrawn from the strength of 767 Squadron at East Haven.

PRODUCTION

32 Barracuda Is were built including the two prototypes, P1767 (c/n F.4468) and P1770 (c/n F.4469) Both prototypes and P9642 to P9666 (25) built at Hayes and five aircraft, DN625 to DN629 built by Westlands.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA BARRACUDA I

ENGINE: One 1,300hp Rolls-Royce Merlin 30 twelve-cylinder vee liquid-cooled

WING SPAN: 49ft 2in; (folded) 17ft 9in

LENGTH: 39ft 9in **HEIGHT:** 15ft 2in

WING AREA: 405 sq ft

EMPTY WEIGHT: 8,700lb

MAX LOADED WEIGHT: 13,500lb

13,30010

MAX SPEED: 235 mph at 11,000ft

MAX CRUISING SPEED:

191 mph at 6,000ft

CLIMB: 6,000ft in 6.7 min

CEILING: 18,400ft

RANGE: 853 miles with

a 1,610lb torpedo



Pictured prior to its maiden flight in early December 1940, P1767 is displayed with its original and traditional tailplane arrangement. *Aeroplane*

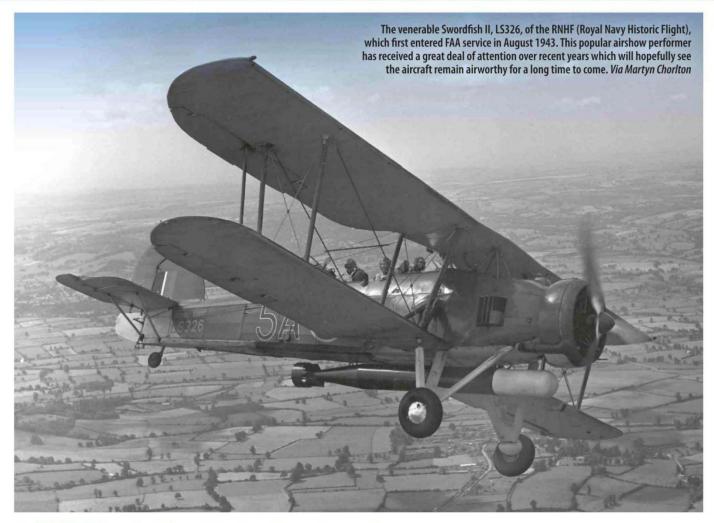


An usual pose for a Barracuda carrying a torpedo which the type as whole was never destined to drop in anger. However, the aircraft did re-introduce the art of dive-bombing back to FAA operations, a role in which the type was generally successful. *Aeroplane*



Barracuda I, P9659, one of just 25 built by Fairey at Hayes pictured carrying a single 1,572lb 18in torpedo. *Charles E Brown via Aeroplane*

SWORDFISH II & III (INCLUDING IV)



>>> FEB 1942

Swordfish enters service with 786 Squadron

>>> MAY 23,1943

First RP success against U-752 from Archer

>>> OCTOBER 1943

Swordfish III enters FAA service

» SEP 1944

Four enemy subs sunk in one voyage from *Vindex*

» DEC13,1944

U-365 sunk by Swordfish from Campania

>> 1946

Swordfish II retired from second FAA duties

The 'Stringbag' evolves

DESIGN

The design of the Swordfish II, III and IV differed little from the Swordfish I (covered on pages 58-59). Introduced into production by Blackburn in 1941, the Mk II was the most prolifically built of all Swordfish. The most significant difference from the earlier mark was that, from the Mk II onwards, the underside of the lower wing was strengthened with a metal skin, enabling the aircraft to carry up to eight 60lb RPs. It was this very effective, yet simple weapon that was tested by a Swordfish at the A&AEE during 1941 for the first time.

The original 690hp Pegasus powered many early Swordfish Mk IIs but this was later replaced by the 750hp Pegasus 30 which was installed in all later aircraft, including all Mk IIIs and Mk IVs.

The Swordfish III also had a strengthened lower wing but was also modified to carry an ASV (Air-to-Surface-Vessel) Mk X radar inside a large radome mounted between the undercarriage legs. The Swordfish IV was basically a Mk II fitted with a fully enclosed canopy specifically for training use in Canada.

Basic defensive armament was the same as the Swordfish I, all later marks could also carry a single 18in torpedo, a 1,500lb sea mine or the same weight in bombs and depth charges. Because of the strengthened wing, as well as the eight RPs, the Mk II and III could alternatively carry eight 25lb armour-piercing RPs on the same under-wing rails.

The Mk III also had the option of being fitted with

rocket-assisted take-off gear (RATOG) which was designed for Swordfish operations with heavy loads from short decks, such as escort carriers and MAC (Merchant Air Carriers) ships. Several Swordfish were also modified to accept a Leigh light under the port wing for nocturnal submarine hunting.

OPERATIONAL SERVICE

The Swordfish II entered FAA service in 1941 and the Mk III in 1943; both continuing to serve until mid-1946. The Swordfish II served with a remarkable 64 FAA squadrons, 23 of them operational. The Swordfish Mk III served with 20 different FAA squadrons and, between the two marks, the type served at sea aboard, HMS Argus, Ark Royal, Courageous, Eagle, Furious, Glorious, Hermes, Illustrious, Indefatigable and Victorious. The MAC ships and escort carriers included HMS Activity, Archer, Attacker, Avenger, Battler, Biter, Campania, Chaser, Dasher, Fencer, Hunter, Nairana, Rapana, Stalker, Tracker and Vindex were also frequented by Swordfish IIs and IIIs.

PRODUCTION

All production of the Swordfish II, III and the Mk IV conversions was carried out by the Blackburn Aircraft Company at Sherburn in Elmet, North Yorkshire. This equated to 1,080 Mk IIs and 320 Mk IIIs. 110 Mk IIs were converted to Mk IVs and Blackburn also supplied the Royal Canadian Navy with 99 Mk IIs and six Mk IIIs.



TECHNICAL DATA SWORDFISH II & III LANDPLANE

ENGINE: One 690hp Bristol Pegasus IIIM3 nine-cylinder radial and later one 750hp Pegasus 30

WING SPAN: 45ft 6in

LENGTH: 35ft 8in

HEIGHT: 12ft 4in

WING AREA: 607 sq ft

EMPTY WEIGHT: 4,700lb

LOADED WEIGHT: 6,750lb

MAX SPEED:

(Torpedo bomber) 139 mph at 4,750ft

CRUISING SPEED:

104-129 mph at 5,000ft

CLIMB RATE:

(III) 5,000ft in 10 mins

RANGE: (Normal fuel & 1,610lb torpedo)

546 miles



All Blackburn-built Swordfish were nicknamed as 'Blackfish' by Fairey staff and Hayes including this Mk II LS268 pictured at Boscombe Down. Built to contract B31192/39, this aircraft was from a batch of 250 which were delivered to the FAA from May 1943 onwards. *Via Martyn Chorlton*



110 Swordfish IIs were modified with canopies over the cockpits for training in Canada. Referred to as the Mk IV, there is no actual official evidence from the period when this designation was used. *Via Martyn Chorlton*



There is no hiding the large radome which houses the ASV Mk X of this Swordfish III pictured at Boscombe Down in July 1944. 320 Mk IIIs were built by Blackburn, many of them going on to serve in support of the Atlantic and Russian conveys achieving several U-boat kills to their credit. *Via Martyn Chorlton*

FIREFLY F.1 (INCLUDING FR & NF.1/2)



The new generation takes over

- >>> DEC 22,1941 Maiden flight of Z1826
- >>> JUN 26, 1942 Chris Staniland killed in Z1827
- >>> AUG 26,1942 Third prototype, Z1828, flies
- >>> OCT 1943
 Firefly F.1 enters
 FAA service
- >>> JUL 1944
 First operational
- >>> JAN 1946
 F.1 enters service
 with the
 Dutch Navy

DEVELOPMENT

Created from a requirement that had its seeds sown in the 1920s, the Firefly was the latest in a long line of two-seat spotter-reconnaissance aircraft, whose performance was good enough for it to be classed as a fighter. By the time the Firefly arrived, more emphasis had been placed on the fighter role but the aircraft would evolve into many specialist variants.

DESIGN

Designed by H E Chaplin, the Firefly made full use of the Fairey-Youngman flap arrangement, although unlike the untidy arrangement of the Barracuda, these could be fully and neatly retracted into the wing.

The Firefly F.1 was a two-seat observer-navigated day fighter, armed with four 20mm cannon and powered by a Griffon IIB engine which gave the aircraft a top speed of 316mph at 14,000ft, which was more than 40mph faster than the Fulmar. The Mk I had a number of derivatives, beginning with the FR.1 which was fitted with ASH submarine and ship-detected radar mounted in a pod directly under the forward fuselage. Next was the NF.1, a night-fighter version of the FR.1, and the F.1A which was a standard F.1 converted to an FR.1. Others were the T.1 dual trainer, the more radical converted post-war T.1 and the TT.1 target-tug (all covered in separate chapters).

An anomaly was the NF.2, which was a derivative of the Mk I but was modified to such an extent that it warranted a different variant. The NF.2 was much heavier than previous aircraft because of the additional equipment needed for an Al operator and because the centre of gravity was moved aft. To compensate for the latter, an 18in long fuselage-lengthening bay was installed behind the engine firewall. Later, NF.2s were converted back to Mk I standard.

OPERATIONAL SERVICE

The first of four prototypes/development aircraft, Z1826 was first flown by Chris Staniland on December 22, 1941 and then by the second, Z1827, on June 4, 1942. Unfortunately it was the latter aircraft that claimed Staniland's life on June 26 when the Firefly suffered an elevator over-balance which caused the tail-unit to fail at low level.

The first production aircraft was delivered to Yeovilton on March 4, 1943 and it was here that the first unit, 1770 Squadron, was formed with the type in October 1943. By the end of the Second World War, nine FAA squadrons were operating the Mk I or one of its sub-variants. The F.1/FR.1 served with a total of 48 FAA squadrons, the NF.1 with twelve squadrons and the NF.2 with two.

Firefly Mk Is flew their first operational sortie in July 1944 when 1770 Squadron from HMS *Indefatigable* led a dive bombing raid against the *Tirpitz*. The same unit also took part in the FAA's first major action against the Japanese in January 1945 when oil refineries in Sumatra were destroyed by an RP attack.

PRODUCTION

By late 1946, 872 Mk Is had been built Fairey constructed the following at Hayes, 327 F.1s, 376 FR.1/NF.1s and 37 NF.2s while General Aircraft built a further 132 Mk Is.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA FIREFLY F.I

ENGINE: One 1,730hp Rolls-Royce Griffon IIB twelve-cylinder vee liquid-cooled

WING SPAN: 44ft 6in (folded) 13ft 6in

LENGTH: 37ft 7in

HEIGHT: 13ft 7in

WING AREA: 328 sq ft

EMPTY WEIGHT: 9,750lb

LOADED WEIGHT:

14,020lb

MAX SPEED: 316 mph at 14,000ft

CLIMB: 5,000ft in 2 min 30 sec

CEILING: 28,000ft

RANGE: 1,300 miles



The prototype Firefly, Z1826, which was first flown from Great West aerodrome by Chris Staniland on December 22, 1941. *Aeroplane*

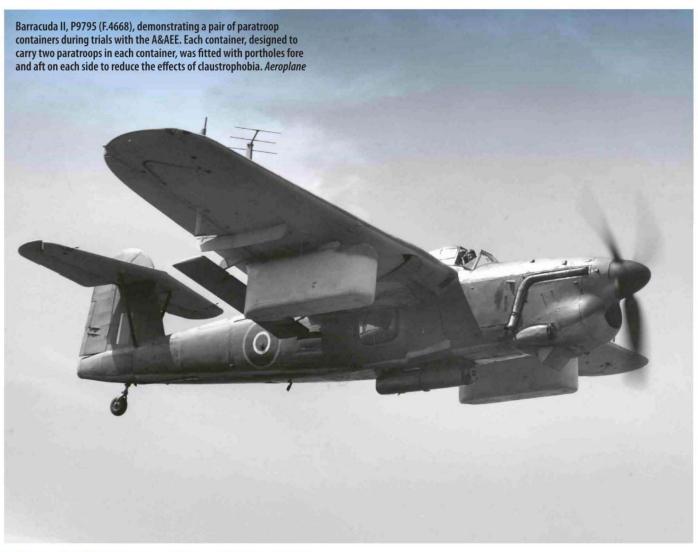


The third prototype Firefly, Z1828, which first flew on August 26, 1942, pictured during trials aboard HMS *Illustrious* later in the year. *Via Martyn Chorlton*



Firefly F.1, DT985, sporting a pair of 500lb bombs. The aircraft was first delivered to RDU Cudham in February 1946 and enjoyed a busy career until June 13, 1955 when it was SOC. *Via Martyn Chorlton*

BARRACUDA II



The definitive dive-bomber

- >>> AUG 17, 1942 First Mk II makes maiden flight
- >>> JAN 10, 1943 Enters service with 827 Squadron
- >>> SEP 1943
 First action
 over Salerno
- >>> APR 3, 1944 Successful attack on the *Tirpitz*
- >>> AUG 24, 1944
 Final attack on
 Tirpitz by
 Barracudas
- >>> 1945 Mk II retired from service

DESIGN

The Barracuda II differed from the Mk I in being fitted with a 1,640hp Rolls-Royce Merlin 32 driving a four-blade propeller and the ability to carry an ASV radar. The latter was a Mk IIN radar with Yagi type antennae which was fitted to each outer upper wing.

By far the most prolific mark built, the Barracuda II, thanks to the extra horsepower of the Merlin 32, could carry a wide range of external stores. While the torpedo was never carried in anger, various bombs, depth-charges and mines were and at least one Mk II was fitted with an air sea rescue (ASR) lifeboat below the fuselage. Another aircraft had containers fitted under each wing, large enough to carry two paratroopers in each. Tested by the AFEE (Airborne Forces Experimental Establishment), several successful 'live' drops were made but the idea was abandoned when it was realised that the psychological effect on the paratroopers, because of being enclosed in such a small space for a long time, was too much to bear.

OPERATIONAL SERVICE

The Barracuda II first entered FAA service when twelve aircraft were delivered to 827 Squadron at Stretton on January 10, 1943. However, the aircraft did not see any action until September 1943 while serving with 810 Squadron from HMS *Illustrious* during the Salerno landings. Two months later, *Illustrious* was joined by 847

Squadron before setting sail for Ceylon where the Barracuda squadrons became part of the 21st TBR Wing. Employed generally for anti-shipping patrols, the aircraft were also used to attack the occasional shore target including oil storage tanks and docks at Sabang in Sumatra and a Japanese submarine base during 1944.

The Mk II hit the headlines on April 3, 1944 when 831 Squadron aboard HMS *Furious* and 827, 829 and 830 Squadrons aboard HMS *Victorious*, escorted by Wildcats, Seafires, Corsairs and Hellcats attacked the battleship *Tirpitz* at Kaafjord in Northern Norway. Attacking in two waves, the Barracudas managed to score up to 15 direct hits with 500lb SAP and 1,000lb HE, causing sufficient damage to keep the battleship in Kaafjord for repairs into the summer of 1944. Only three Barracudas and a single fighter were lost in this brave action which was repeated in May and August 1944 but not with the same level of success.

The Mk II served with 53 FAA squadrons, 20 of them operational, and also with the RAF's 567, 667, 679 and 691 Squadrons; the type being withdrawn from service by the summer of 1945.

PRODUCTION

1,688 Barracuda IIs were built; 675 by Fairey at their Stockport and Ringway factories, 700 by Blackburn, 300 by Boulton-Paul and 13 by Westland.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA BARRACUDA II

ENGINE: One 1,640hp Rolls-Royce Merlin 32 twelve-cylinder liquid-cooled

WING SPAN: 49ft 2in (folded) 17ft 9in

LENGTH: 39ft 9in **HEIGHT:** 15ft 2in

WING AREA: 405 sq ft

EMPTY WEIGHT: 9,350lb

MAX LOADED WEIGHT: 14,100lb

MAX SPEED: 228 mph at 1,750ft

MAX CRUISING SPEED: 193 mph at 5,000ft

CLIMB: 5,000ft in 6 min

CEILING: 16,600ft

RANGE: 686 miles with a 1,610lb torpedo



Fairey-built Mk II, P9926, presents an excellent view of the Yagi antenna for the aircraft's ASV Mk IIN radar. The aircraft is also carrying a 1,620lb 18in torpedo, a weapon the Barracuda II never dropped in anger. *Aeroplane*

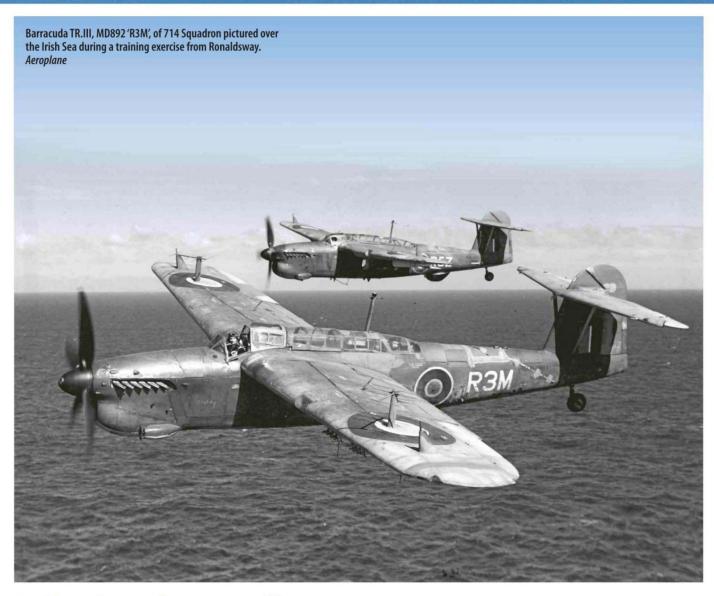


Fairey Barracuda IIs of 815 Squadron lead the way from HMS *Indomitable* in 1944. The unit operated the mark on two occasions, firstly October 1943 to October 1944, and then between November 1944 and April 1945. *Via Martyn Chorlton*



Blackburn-built Mk II, MX613, which was modified to carry a substantial ASR lifeboat which apparently worked well but was only ever used for experiments and demonstrations. *Aeroplane*

BARRACUDA III (AKA TR.III)



Anti-submarine operations

- >>> 1943 First flight
- >>> MAR 1944
 First delivery to FAA
- >>> JAN 1945
 Entered FAA
 operational
 service
- >>> FEB 1946 821 Squadron disbanded
- >>> 1953
 Retired from service

DESIGN

First flown in 1943 following the conversion of Boulton Paul-built Mk II, DP855, the Barracuda III did not enter service until 1944. Specifically introduced for antisubmarine operations, the Mk III was fitted with an ASV Mk X (redesignated Mk XI) radar in an enclosed radome under the rear fuselage. This radar had a range of 38 miles when used against surface ships and, when the conditions were right, a submarine on the surface could be detected from a distance of 12 miles. The weapon of choice for these operations was a single 1,500lb mine which was carried on crutches underneath the fuselage.

OPERATIONAL SERVICE

The Barracuda III, redesignated the TR.III, first entered operational service with 815, 821 and 822 Squadrons in January 1945 and 810 Squadron in the following month. Only 821 Squadron enjoyed a period of limited action, operating from HMS *Puncher* off the Norwegian coast laying mines. During these final months of the war, the squadron came and went across the North Sea, also serving with HMS *Campania*, *Puncher* and *Searcher* before embarking on HMS *Trumpeter*. In July 1945, it was bound for the Far East to join the Eastern Fleet. VJ Day

arrived before they could make a difference and, in early September, the squadron returned home aboard HMS Fencer without aircraft, only to re-equip with new TR.Ills at Rattray. By February 1946, 821 Squadron was disbanded, leaving 815 Squadron which re-equipped with the TR.Ill for a second time in December 1947. Mainly operating to and from Eglinton during this period, the unit did not relinquish its TR.Ills until May 1953. In the second line, 750 Squadron kept the TR.Ill on strength until July 1953.

860 Squadron was the only other operational unit with the TR.III between January 1945 and January 1946, operating from Maydown, Ayr and Fearn. In the second line, the TR.III served 700, 703, 707, 710, 713, 714, 719, 735, 737, 744, 747, 750, 756, 769, 778, 783, 785, 796, 798 and 799 Squadrons.

PRODUCTION

852 Barracuda IIIs were built by Fairey and Boulton Paul. Fairey produced 406 in the serial ranges PM682 to PM999, PN115 to PN164 and RK328 to RK574. Boulton Paul built 392 in the serial ranges, MD811 to MD992, ME104 to ME293 and RJ759 to RJ966. A further 208 aircraft contracted to Boulton Paul were cancelled.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA BARRACUDA III

ENGINE: One 1,640hp Rolls-Royce Merlin 30 twelve-cylinder vee liquid-cooled

WING SPAN: 49ft 2in (folded) 17ft 9in

LENGTH: 39ft 9in

HEIGHT: 15ft 2in

WING AREA: 405 sq ft

EMPTY WEIGHT: 9,407lb

MAX LOADED WEIGHT:

14,100lb

MAX SPEED: 239 mph

at 1,750ft

MAX CRUISING SPEED:

205 mph at 5,000ft

CLIMB: 5,000ft in 4 min

30 sec

CEILING: 20,000ft

RANGE: 684 miles with

a 1,572lb torpedo



Boulton Paul-built Barracuda TR.III, MD837, outside the company's factory at Pendeford, near Wolverhampton, in mid-1944. *Aeroplane*



Devoid of squadron markings and its ASV Mk X radar, TR.III, RJ905, of 750 Squadron warms its Merlin engine through at St Merryn in 1953. Via Martyn Chorlton



TR.III, RJ921, '305/GN' of 815 Squadron pictured at Eglinton (now Derry Airport) in 1950. *Via Martyn Chorlton*

BARRACUDA V



Barracuda TR.V, RK558, '325/LP' of 783 Squadron, operating out of Lee-on-Solent in the summer of 1948. Aeroplane

Griffon power comes too late

- >>> NOV 16, 1944 Mk V maiden flight
- >>> NOV 22, 1945

 First production
 aircraft flies
- >>> SEP 1946
 Enters service with
 778 Squadron
- >>> OCT 27, 1947
 The last TR.V is delivered to the FAA
- >>> DEC 1947
 Enters service with
 783 Squadron
- >>> OCT 1948
 Retired from the FAA

DESIGN

Nothing more than a modified TR.III with a Griffon engine on the surface, the Barracuda V was a very different animal compared to its predecessors. The aircraft was redesigned to serve in the war in the Pacific as an interim, pending the arrival of the Spearfish but was ultimately too late, never entered operational service and was only produced in small numbers.

The prototype aircraft, which was designated the Mk IV, was converted Mk II, P9976, powered by a 1,850hp Griffon VII (later an VIII) engine. First flown on November 16, 1944, from Ringway by Flt Lt S M Moseley, the aircraft was built for a crew of two, namely the pilot and a navigator/radar-operator.

The production aircraft were powered by a 2,020hp Griffon 37 and, amongst the many modifications were a redesigned wing with squared off tips and a span four feet longer than the earlier aircraft. The aircraft was structurally strengthened, the electrical systems were heavily modified and brought up to date and a new radar scanner was fitted to leading edge of the port inside a radome which was easily removable as a single unit. A single forward firing machine gun was fitted and the fuel

capacity of the Mk V was increased.

The early production Mk Vs, later designated as the TR.V, were Mk II and Mk III conversions which used the same fin and rudder but, from RK530 onwards, (the first proper production aircraft) a large dorsal fin was fitted complete with an increased area rudder which rose to a point above the fin. The first production aircraft took to the air on November 22, 1945 and the last was delivered on October 27, 1947.

SERVICE

The Barracuda TR.V only served with two FAA units, both of which were second line. The first was 778 Squadron at Ford, the type remaining on strength from September 1946 to July 1947 and the second was 783 Squadron, based at Lee-on-Solent, from December 1947 to October 1948.

PRODUCTION

Only 35 Barracuda TR.Vs were built from an original order for 140 aircraft, including ex-Mk IIs P9976, DT845, PM940, PM941 and PM944, all by Fairey in the serial ranges, RK530 to RK542 and RK558 to RK574.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA BARRACUDA V

ENGINE: (Prototype) One 1,850hp Rolls-Royce Griffon VII twelve-cylinder vee liquid-cooled; (Production) One 2,020hp Griffon 37

WING SPAN: 53ft; (folded) 18ft 5½in

LENGTH: 41ft 7in **HEIGHT:** 17ft 3in

WING AREA: 435 sq ft

EMPTY WEIGHT: 11,430lb

NORMAL LOADED WEIGHT: 15,250lb

MAX SPEED: 253 mph at 10,000ft

CLIMB: 10,000ft in 8.6 min

CEILING: 24,000ft

RANGE: 600 miles at 163 mph with 2,000lb

of bombs



The Rolls-Royce Griffon 37 engine fitted in the TR.V did nothing to improve the appearance of the Barracuda. This is RK532 pictured in October 1946. *Aeroplane*



The final look of the Fairey Barracuda after almost six years of development. The potential of the TR.V was cut short by the end of the Second World War. *Aeroplane*



Originally built by Fairey as a Mk II, P9976 was the original TR.V prototype powered the Griffon VII engine. *Via Martyn Chorlton*

SPEARFISH



The Barracuda replacement

- >>> JAN 1943 Specification 0.5/43 issued
- >>> JUL 5, 1945 RA356 first flight
- >>> DEC 29, 1946 RN241 first flight from Ringway
- **APR 11, 1947**RA360 first flight
- >>> SEP 23, 1947 TJ175 first flight
- >>> MID-1952 Final Spearfish test flight

DEVELOPMENT

As early as January 1943, the Specification O.5/43 was issued for the replacement of the Barracuda and, ultimately, the Grumman Avenger which enter FAA at a similar time. The American-built Avenger performed better than the Barracuda, mainly because it's weapons load was carried internally and the aircraft was lighter and the engine more powerful. Fairey attempted to design all of these factors into its latest aircraft, the Spearfish.

DESIGN

Designed by Herbert Eugene Chapman, the Spearfish was heavily influenced by the fact that it was intended for the Royal Navy's new 45,000 ton *Gibraltar*-Class fleet carriers which were planned to be commissioned by 1945. These huge carriers gave Eugene the freedom to design an aircraft much larger than the Barracuda and with the option of the new, powerful Bristol Centaurus engine, the Spearfish was destined to be a brute of an aircraft.

The aircraft's size was partly dictated by a large weapons bay, capable of holding four 500lb bombs, a single 1,800lb, 18 inch torpedo and/or a combination of mines and depth charges. Defensive armament employed the .50in Browning machine-gun; two forward-firing, and two in a remotely controlled Frazer Nash Type 95s in a barbette behind the rear cockpit.

As with the Barracuda before it, Fairey Youngman flaps were employed although, on the Spearfish, they were fully retractable. The Spearfish had hydraulically-folding wings, a hydraulically powered sliding canopy, space for a Mk XV ASV radar inside a retractable radome below the rear fuselage and a substantial, outwardly retracting undercarriage.

The 2,585hp Centaurus 57 engine was flushly fitted into the nose of the Spearfish, for example, the exhaust manifolds were recessed and the oil cooler was neatly installed into the port leading edge of the wing. The aircraft was fitted with a Rotol five-blade constant-speed propeller but later, production aircraft were to be fitted with a reverse pitch propeller which could double as airbrake during a diving attack.

SERVICE

The prototype, RA356, built at Hayes, was first flown on July 5, 1945 by Fairey's chief test pilot Flt Lt Foster Hickman Dixon. Production was planned to be carried out Fairey's Heaton Chapel factory in Stockport and the first aircraft, RN241, left there on December 29, 1945. Eight prototypes were ordered and an order for 152 production aircraft was already in place when news came through that the *Gibraltar* fleet carriers had been cancelled and the entire FAA's requirements for torpedocarrying aircraft was abandoned.

Regardless, work continued at a much slower rate on the two airframes which were still under construction at Hayes, RA360 and RA363, while RN244 and TJ179 were eventually completed, neither was destined to fly. TJ175 was the last Spearfish to take to the air in September 1947 and is believed to have been the last one flying; the aircraft serving with the Royal Navy's Carrier Trials Unit at Ford until the summer of 1952.

PRODUCTION

Eight prototypes ordered, RA356, RA360, RA363, RN241, RN244, TJ175, TJ179 and TJ184. 152 production aircraft, cancelled in 1945.



TECHNICAL DATA SPEARFISH

ENGINE: One 2,585hp Bristol Centaurus 57 14-cylinder air-cooled sleeve-valved radial

WING SPAN: 60ft 3in (folded) 20ft

LENGTH: 44ft 7in

HEIGHT: 13ft 6in

EQUIPPED EMPTY WEIGHT: 15,200lb

OVERLOAD WEIGHT:

(bomber) 22,083lb; (long-range recce) 21,882lb

MAX SPEED: 292 mph

at 14,000ft

CLIMB: 10,000ft in 7 min 45 sec

CEILING: 25,000ft

MAX RANGE: 1,036 miles at 196 mph

at 15,000ft



The only Stockport-built Spearfish to fly was RN241 which first took to the air from Ringway (now Manchester) on December 29, 1945. *Via Martyn Chorlton*



Hayes-built RA360 which first flew on September 23, 1947 presents a good view of how neatly the powerful Centaurus engine was fitted. *Via Martyn Chorlton*



RN241 was fitted with dummy examples of the Fraser Nash dorsal barbette and the radome for the ASV Mk XV radar. *Via Martyn Chorlton*

FIREFLY 4 & FR.4 (INCLUDING F.3)



A redesign and more power

- FEB 1942
 Fairey raise the concept of Griffon 61-power
- >>> MAY 25, 1945
 Prototype FR.4
 flies for first time
- >>> AUG 1947
 Enters service
 with the RCN
- >>> OCT 1947
 Enters service
 with the FAA
- >>> JAN 1948

 Last FR.4 delivered
- >>> 1949
 Retired from the FAA

DEVELOPMENT

The story of the next variant of the Fairey Firefly came about when F.1, Z1835, was experimentally installed with a two-speed, two-stage supercharged Griffon 61 engine. The idea was first mooted by Fairey in early 1942, who proposed a single-seat version, a concept that the Admiralty were still not comfortable with. The Royal Navy was interested in a two-seat night fighter version and, in October 1942, boldly gave Fairey a contract for 200 aircraft long before Z1835, by now designated as an Mk 3, was even tested.

DESIGN

Z1835 was not an attractive looking aircraft because the Griffon 61 needed a large 'beard-type' radiator surrounded by a large cowling in an effort to keep the powerful engine cool. Despatched to Boscombe Down for trials in July 1943, performance was found to be poor due to the lack of aerodynamics around the forward fuselage; handling was also affected. The aircraft was still 35mph faster than the F.1, but a considerably bigger increase in performance had been hoped for. Trials were abandoned, although an initial order for 100 aircraft from the Admiralty still stood.

By 1944, the idea was looked at again and this time Firefly F.1, Z2118, was modified with leading edge radiators but retained the original aircraft's elliptical wing. By the following year, Z2118 was joined by three other prototypes, including Z1835, which were all modified to take a Griffon 72 powerplant and would effectively become the new Mk 4.

The production aircraft were fitted with a Griffon 74 driving a four-blade propeller. The leading edge radiators were installed within a pair of forward extensions at each

wing root. The wing span was reduced and had squared off tips which increased the rate of roll. Auxiliary fuel was carried in a 55 gallon nacelle under the port wing which was counter-balanced by a second nacelle carrying a radar scanner and equipment. These nacelles could be both used as fuel tanks, raising the aircraft's total capacity to 256 gallons.

At least one aircraft, MB649, was produced as the NF.4 but, despite a healthy initial order, the contract for the type was cancelled in 1946.

OPERATIONAL SERVICE

The first true prototype FR.4 flew on May 25, 1945 and was followed by a second on February 21, 1946. One of the original four prototypes, PP482 was employed to carry out operational trials with HMS *Illustrious* and, on May 25, 1946 the first production aircraft, TW687, made its maiden flight.

The FAA received its first FR.4 in July 1946 but it was not until February 12, 1947 that the FR.4 was cleared for 'restricted' shore-based operations. It was with the Royal Canadian Navy that the aircraft first entered operational service in August 1947 and was followed by the FAA's 810 Squadron on October 1, 1947 followed by 812 and 81 Squadrons.

The FR.4 only served with 810 Squadron for two years and by the middle of 1948, 812 Squadron had converted to the FR.5 and 814 Squadron followed suit in early 1949.

PRODUCTION

67 FR.4s were built, all by Fairey at Hayes, in the serial ranges, VG957 to VG999 and VH121-VH144. This was from an original contract placed in June 1945 for 200 aircraft and, from VH203 onwards, the aircraft were to have been built as NF.4s.



TECHNICAL DATA FIREFLY FR.4

ENGINE: One 2,100hp Rolls-Royce Griffon 74

WING SPAN: 41ft 2in (folded) 13ft 6in

LENGTH: 38ft

HEIGHT: 13ft 11in

EMPTY WEIGHT:

9,674lb

MAX LOADED WEIGHT: 13,479lb

13,1,210

MAX SPEED: 367 mph at 14,000ft

CLIMB: 5,000ft in 3 min 36 sec

CEILING: 31,900ft

MAX RANGE:

(normal tankage) 760 miles at 209 mph



First delivered to RDU Cudham in October 1947, FR.4 VG985 later served with 767 and 787 Squadrons before being converted to a TT.4 and joining the Indian Navy as INS-117 in September 1958. *Via Martyn Chorlton*



Good view of the FR.4's re-designed frontal view which did away with the original 'beard-type' radiator in favour of radiators buried in the leading edge wing roots. This is Z2119, with original elliptical wings and standard tail unit, pictured on May 25, 1945. *Via Martyn Chorlton*



Factory-fresh FR.4, '11-42', one of 40 aircraft (F.8227 to F8266) supplied to the Royal Netherlands Naval Air Service (Kon Marine) between February and September 1947. *Via Martyn Chorlton*

FIREFLY T.1, T.2, T.3, T.5 & T.7



A selection of dual-controlled pilot and observer trainers

- >>> JULY 1946
 Fairey Trainer
 maiden flight
- >>> SEP 1, 1947
 T.1, MB473 first
 production aircraft
- >>> JULY 1948
 T.1 enters
 FAA service
- >>> APR 1949
 T.2 enters
 FAA service
- 33 1951 T.3 makes maiden flight
- >>> AUG 1953
 T.7 enters
 FAA service

DEVELOPMENT

Due to the large number of Fireflies in both operational and second-line service it was only natural that a dual-controlled training variant be produced during the post-war period. Spread across five different marks, the original T.1 was a private venture by Fairey which was accepted by the FAA.

DESIGN

The Firefly T.1, 2 and 3 were developed as pilot and observer trainers and all were conversions of F.1s and FR.1s. All were converted at Fairey's Heaton Chapel factory and flown for the first time from Ringway. The T.1 and T.2 were designed as pilot trainers with the instructor accommodated in the rear cockpit, where there was a second set of flying controls, instruments and a sliding canopy. The instructor's cockpit, in similar fashion to the Battle trainer, was raised to a so that he could see over the nose in level flight.

The first of the Firefly trainers was converted F.1, MB750 (later registered as G-AHYA), first flown from Ringway in July 1946 by Duncan Menzies. Originally known as the 'Firefly Trainer', the aircraft was extensively trialled to test whether it was suitable for deck-landing and operational training. As G-AHYA, the aircraft demonstrated across Europe and, in the hands of Gp Capt R G Slade, took third place in the Lympne High-Speed Handicap on August 31, 1947 with an average speed of 290mph.

The first of 34 production T.1s, MB473, made its maiden flight on September 1, 1947. Out of this batch, nine out of the 34 T.1s were armed with two 20mm cannon. The T.2 was an operational pilot trainer which

was fitted with a pair of 20mm cannon and gyrogunsights which were synchronised for both the pupil and instructor.

The T.3 was a conversion of the FR.1, it made its debut in 1951 and was lacking the raised rear cockpit. This was an observer-trainer which was unarmed but did carry the special equipment needed to train for anti-submarine operations. The T.5 was a dual-control version of twelve Royal Australian Navy AS.5s; the work being carried out by Fairey Aviation in Australia.

The final training Firefly was the T.7 which was the AS.7 converted to an observer/radar training aircraft. The T.7 only operated from shore bases as it was not fitted with deck-arrestor gear. The last T.7 was delivered to the FAA in December 1953.

OPERATIONAL SERVICE

The T.1 first entered service with 736 Squadron at St Merryn in July 1948, later joining 14 other FAA squadrons before being replaced by trainer version of the Sea Fury. The T.2 first entered service with 737 Squadron in April 1949 and the T.3 with 796 Squadron in July 1950. The T.5 entered service with 851 (RAN) Squadron in August 1954 followed by 724 (RAN) Squadron in June 1955. The T.7 served five FAA squadrons, the first being 719 Squadron in August 1953, until the aircraft was retired in December 1957 from 796 Squadron.

PRODUCTION

34 T.1s were built/converted by Fairey at Heaton Chapel; 57 T.2s, an unknown quantity of T.3s; twelve T.5s and approximately 140 T.7s.



TECHNICAL DATA FIREFLY T.1

ENGINE: One 1,735hp Rolls-Royce Griffon IIB or XII

WING SPAN: 44ft 6in (folded) 13ft 6in

LENGTH: 37ft

HEIGHT: 12ft 4in

EMPTY WEIGHT: 9,647lb

LOADED WEIGHT:

12,485lb

MAX SPEED: 305 mph

at 16,500ft

CLIMB: 5,000ft in 2 min

42 sec

CEILING: 28,400ft

MAX RANGE:

(normal tankage)

805 miles



Royal Australian Navy T.5, VX375, which was taken on RAN charge on April 7, 1949 to serve on board HMAS *Sydney*. Only a dozen T.5s were converted from FR.5s. *Aeroplane*

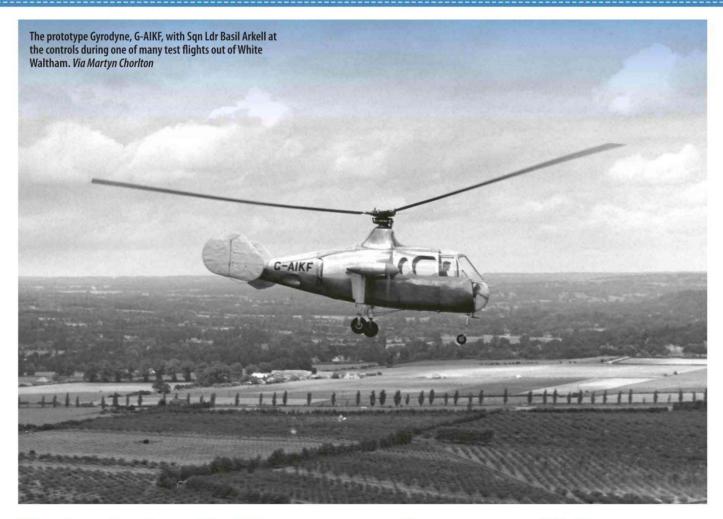


The prototype 'Fairey Trainer' pictured as MB750 in early 1946 but, by the end of the year, was re-registered as G-AHYA. After being SOC from the Royal Navy in late 1948, the aircraft was sold to Thailand as SF9. *Via Martyn Chorlton*



Ex-Firefly F.1 which served briefly with 790 Squadron at Dale before being converted in late 1946 to a T.1. The aircraft was SOC on March 16, 1954 after a wheels up landing at Hooton Park eight days earlier. *Via Martyn Chorlton*

FB.1 GYRODYNE (AKA FAIREY-BENNETT ONE)



The beginning of a 15-year saga of success and failure

- » APR 3, 1946 E.4/46, Gyrodyne announced
- >>> SEP 1947
 Aircraft exhibited at SBAC
- >>> DEC 7, 1947
 First untethered
 flight by Basil Arkell
- >>> MAR 1948
 Second prototype,
 G-AJJP completed
- >>> JUN 28, 1948 'Class G' speed record captured
- APR 17, 1949 G-AIKF crashed at Ufton, nr Reading

DEVELOPMENT

Designed to take advantage of the best features of the autogyro and the helicopter; the Gyrodyne dates back to 1945. By the spring of the following year, Fairey made public its intentions to produce a new rotary wing aircraft. The idea had been brought to the company by Dr J A J Bennett who, prior to joining Fairey in 1945, had taken over Cierva following the death of Juan de la Cierva in December 1936.

DESIGN

Powered by a single Alvis Leonides radial engine, the attractive looking Gyrodyne weighed in at 3,600lb empty, half of this figure was made up of the engine and the complex transmission system. The latter comprised four units; firstly the engine, secondly the main gearbox with reduction gear for the propeller and main rotor drives, thirdly an upper gearbox with double-epicyclic reduction gear and finally, another gearbox in the starboard wing stub for reduction and pitch-changing control of the propeller.

SERVICE

Fairey were awarded a contract for two prototypes under Specification E.4/46, the first of these, G-AIKF (aka VX591), was displayed as a static exhibit at the SBAC, Radlett, in September 1947. Prior to its first flight, the Gyrodyne carried out 85 hours of engine ground tests and 56 hours of rotor testing. On December 7, 1947, Gyrodyne G-AIKF, in the hands of project test pilot, Sqn Ldr B H Arkell, made

its first untethered fight from White Waltham. Flight testing continued into 1948 until March when the aircraft was grounded and dismantled so that the transmission system could be internally checked for signs of wear and tear. In the meantime, the second prototype, registered as G-AJJP, had been completed.

On re-assembly, G-AIKF was prepared for an attempt on the International Helicopter (Class G) speed record in a straight line. This record had stood, 'unofficially', since June 1938 when the Focke Achgelis Fa 61 captured it, although a Sikorsky R-5 had also 'unofficially' been recorded reaching a speed of 114.6mph. On June 28, 1948, Basil Arkell flew a pair of eastbound, and two westbound flights along a 3km course over White Waltham. An average speed of 124.3mph was recorded, capturing the record for Britain.

A few months later, the 100km closed-circuit record was the next target. Unfortunately, only two days before the record attempt, on April 17, 1949, Gyrodyne G-AIKF crashed at Ufton near Reading. The crash killed test pilot, F H Dixon, who had been Fairey's chief test pilot since 1936, and observer, Derek Garroway. Dixon had been involved in a great deal of the demonstration and development flying of the Gyrodyne, alongside Basil Arkell.

The resulting investigation into the accident reported that the cause of the crash was a fatigue failure of the main rotor head. Development was brought to a halt and the second aircraft was grounded, destined to re-appear four years later as the Jet Gyrodyne.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA GYRODYNE

ENGINE: One 520hp Alvis Leonides nine-cylinder radial

ROTOR DIAMETER: 51ft 9in

STUB WING-SPAN: 16ft 8in

FUSELAGE LENGTH: 25ft

HEIGHT: 10ft 2in

EMPTY WEIGHT: 3,600lb

LOADED WEIGHT: 4,800lb

MAX SPEED: Approx 140 mph



Although it is difficult to see in this view, a propeller was only fitted to the starboard wing stub to counteract the torque of the main rotor. Aeroplane



G-AJJP, the second prototype, only differed from the first aircraft in having a more passenger friendly interior and was used extensively for demonstration flying. *Aeroplane*



G-AIKF only existed for approximately 18 months. The aircraft was destroyed following rotor head failure on April 17, 1949. *Aeroplane*

FIREFLY AS/FR & NF.5



Formidable in the Far East

- >>> DEC 12, 1947
 First production
 Firefly Mk 5 flies
- >>> JAN 24, 1949
 First Firefly with power-folding wings
- >>> MAY 19, 1950 Last Firefly Mk 5 delivered
- » APR 25, 1952 825 Sqn in action over Malaya
- >>> 1952
 Fireflies carry out
 123 attacks in a
 single day
- >>> OCT 2, 1972 WB217 restored back to flight for the RNHF

DEVELOPMENT

Very similar, externally at least, to the Firefly Mk 4, the Mk 5 represented the peak of development for the family. It also arrived in service with the FAA during a period of intensive operations, which culminated in the Korean War; a conflict in which the Firefly would serve with distinction.

DESIGN

There were three variants of the Firefly Mk 5; the FR.5, the NF.5 and the AS.5. The latter was the anti-submarine variant which carried detection equipment under each wing. The AS.5 was fitted with an ARI 5284 radio altimeter and an ARI 5286 sonobouy controlled from within the observer's cockpit. The aircraft could carry up to twelve Americantype sonobouys and a pair of 250lb depth charges.

The night-fighter variant, the NF.5, was fitted with flame damping manifolds and an ARI 5284 radio altimeter, this time coupled with an ARI rearward facing radar. This simple, but effective piece of equipment warned the pilot if an aircraft was approaching from the rear by ringing a bell in the cockpit.

The FR.5 was very similar to the FR.4 but was fitted with the latest internal equipment. It was an FR.5 that finally introduced power folding wings. The first aircraft to have this luxury fitted was VX414 in January 1949. The FR.5 also had radome vibration dampers fitted to improve the radar reception.

OPERATIONAL SERVICE

The first production Firefly 5, VT362, flew on December 12, 1947 and, by May 1948, the first FR.5s had joined 778 and

782 Squadrons for trials, followed by the AS.5 which entered FAA service in July, 1949 with 810 Squadron at St Merryn.

The FR.5 was by far the most useful of the three variants (the NF.5 only served with the Dutch Navy) and later several squadrons, especially during the Korean conflict, would operate this variant rather than the later AS.6.

812 Squadron was the first operational unit to receive the FR.5 in July 1948 but it was 825 Squadron operating from HMS *Ocean* that first took the type into action over Malaya on April 25, 1952 whilst en route to Korea. 812 Squadron returned to Malaya in May 1954, its FR.5s carrying out 16 successful attacks on targets in Central Johore.

The Korean conflict saw the FR.5 extensively employed, beginning with 810 Squadron who operated from HMS *Theseus* between October 1950 and April 1951. 812 Squadron's FR.5s, now on board HMS *Glory*, took over until May 1952 when it was the turn of 825 Squadron on board HMS *Ocean*. By October 1952, 825 Squadron took on board HMS *Glory* which remained in theatre until May 1953. It was HMS *Ocean* with 810 Squadron which saw the armistice signed on July 27, 1953.

PRODUCTION

170 aircraft were initially ordered on November 16, 1946 but this was reduced to 117 aircraft in the serial ranges, VT362 to VT381, VT392 to VT441 and VT458 to VT504. Events unfolding in the Far East prompted a second batch of 169 Firefly Mk 5s to be ordered on December 20, 1948 in the serial ranges, WB243to WB272, WB281 to WB316, WB330 to WB382 and WB391 to WB440.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA FIREFLY AS.5

ENGINE: One 2,245hp Rolls-Royce Griffon 74

WING SPAN: 41ft 7in

LENGTH: 37ft 11in

HEIGHT: 13ft 11in **EMPTY WEIGHT:**

9,674lb

LOADED WEIGHT:

16,096lb

MAX SPEED: 386mph

at 14,000ft

CEILING: 31,900ft

MAX RANGE: 760 miles



Firefly AS.5, WB281, which joined 810 Squadron at St Merryn on August 12, 1949. After floating over the wires on HMS *Theseus* on August 8, 1950, the aircraft crashed into WB376 and WB369 which were parked on the deck. The Firefly was SOC and dumped over the side the same day. *Aeroplane*

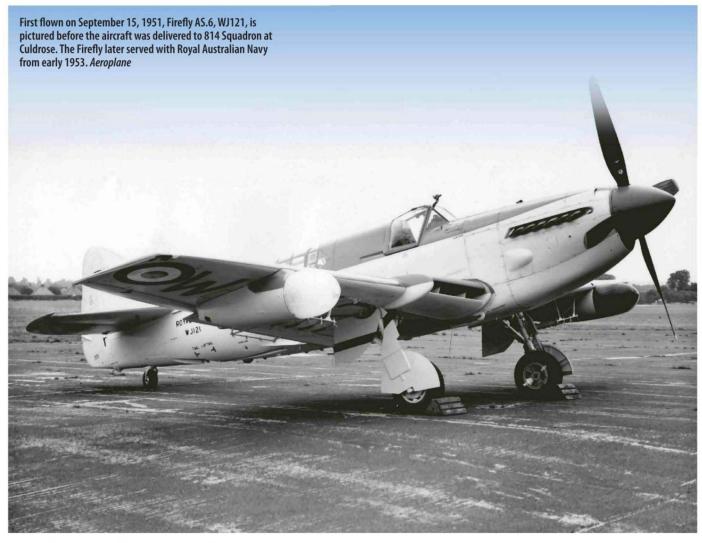


Firefly AS.5, WB404, enjoying a Home Fleet cruise across the Mediterranean on board HMS *Ocean* in the summer of 1951. *Aeroplane*



It took a small army of groundcrew to manually unfold the wings of a Firefly prior to the arrival of power folding wings from VX414 onwards. This is FR.5, VT413, which was later converted to a U.9. Its fate was sealed when a Sea Vixen shot the drone down off Malta November 23, 1961. *Aeroplane*

FIREFLY AS.6



Dedicated anti-submarine

- MAR 23, 1949
 Maiden flight
 of WB505
- >>> MAY 26, 1950

 First production
 aircraft delivered
- >>> DEC 1950 AS.6 joined RAN
- >>> JAN 1951
 First operational
 AS.6 joins FAA
- MAY 1951
 814 Squadron wins
 the Boyd Trophy
- >>> 1958
 Retired from the RAN

DESIGN

To improve the Firefly's anti-submarine capability, a sacrifice had to be made and, in the case of the Firefly AS.6, the defensive 20mm cannon were removed. To compromise, the AS.6 had hard points fitted to the underside of the wings, able to carry 16 3in RPs, carried in a group of eight in two rows of four under each wing. These hard points could also carry depth charges, mines or sonobouys.

The first production Firefly AS.6, WB505, made its maiden flight on March 23, 1949 and was delivered to the FAA on May 26, 1950.

OPERATIONAL SERVICE

The first unit to receive the Firefly AS.6 was the Royal Australian Navy's 817 Squadron, under the command of Lt Cdr R B Lunberg, RN. After reforming at St Merryn with AS.5s the unit, as part of the 21st CAG, set sail on board HMS *Sydney* for Australia. When the unit arrived at Nowra on December 6, 1950, it also took charge of several AS.6s.

For the FAA, it was 814 Squadron, which had reformed at Culdrose on November 22, 1950 under the command of Lt Cdr A C Lindsay DFC, RN, which first received the AS.6 in January 1951. Along with 809 Squadron (operating the Sea Hornet NF.21), the unit formed the 7th NAG (Night Air Group), the first all-weather group. After embarking on HMS *Vengeance* to work-up for its part in

the 7th NAG during May 1951, the unit managed to carry out 927 hours of night training, for which it received the annual Boyd Trophy for its efforts.

The AS.6 served with 13 squadrons in the antisubmarine role, the majority of its work being carried out over the North Sea and Mediterranean, keeping tabs on Soviet movements. By 1955, the AS.6 was being withdrawn from FAA service, making way for the Gannet, but the RAN continued to operate the type until 1958. One of the RAN units, 816 Squadron, used their AS.6s to take part in atomic bomb testing on the Monte Bello Islands in October 1952. The last bastion for the AS.6 was 851 (RAN) Squadron which was reformed at Nowra on August 3, 1954, specifically to carry out anti-submarine training. The unit was disbanded on January 13, 1958 taking the AS.6 with it.

The AS.6 also served extensively in many second-line FAA squadrons, including 703, 703A, 719, 723, 724, 737, 737X, 744, 751, 767, 771, 782 and 796 Squadrons.

PRODUCTION

The first production batch, for just 14 aircraft, WB505 to WB510 & WB516 to WB553, was ordered Feb 1949 although these serials were preceded by WB422 to WB440 (16) which were AS.5s converted to AS.6 standard. A total of 133 'new-build' AS.6s were constructed between January 1947 and September 1951.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA FIREFLY AS.6

ENGINE: One 2,245hp Rolls-Royce Griffon 74

WING SPAN: 41ft 7in

LENGTH: 37ft 11in

HEIGHT: 13ft 11in **EMPTY WEIGHT:**

9,674lb

LOADED WEIGHT:

16,096lb

MAX SPEED: 386mph

at 14,000ft

CEILING: 31,900ft

MAX RANGE: 760 miles



Firefly AS.6, WH632, '211/FD' of 1840 Squadron based at Hal Far. The Malta-based unit flew the As.6 from July 1951 to May 1956. Aeroplane



Originally laid down as an AS.5, WB438 was one of 19 converted from a production batch of 169 AS.5s ordered in December 1948. The aircraft is pictured during its service with 826 Squadron (274/A) in 1953, which was operating from HMS Indomitable. Via Aeroplane



WJ120, caught on final approach to Ford whilst serving with 1840/1842 Squadrons in early 1955. Via Martyn Chorlton

FIREFLY TT.1, 4, 5 & 6



The international target tug

>> 1949

TT.1 enters service with Svensk Flygjänst

- >>> NOV 1951
 TT.4 enters service
- >>> NOV 1954 TT.5 joins the RAN
- >>> DEC 1958
 Airwork FRU
 retired the TT.4
- >>> 1963 Svensk Flygjänst flies last sortie
- >>> MAR 1966

 RAN retired the TT.6

DEVELOPMENT

Fairey did not miss a trick when it came to broadening the appeal of the Fairey Firefly during the post-war period. This even included presenting the aircraft as a potential target-tug, a role that it was particularly well suited to.

DESIGN

The first of four different marks of Firefly target tug was the TT.1 which actually came about from a proposal from Svensk Flygjänst who had won a contract to support Swedish anti-aircraft units. Modifications for this latest mark included the fitment of a windmill-operated (windlass arm) RFD Type 'B'; Mk 2B cable winch, mounted on the port side of the fuselage between the pilot's and cable operator's positions.

Next was the TT.4, based on the Firefly 4, which was fitted with an air-driven Mk Type G winch mounted under the centre-line of the fuselage directly below the wing. The bomb shaped pod made it much easier to convert any Firefly 4, 5 or 6 into a target-tug without major modifications.

Two TT.5s were created for the Royal Australian Navy from a conversion kit supplied by Fairey and four RAN AS.6s were converted into TT.5s in the same way.

OPERATIONAL SERVICE

Out of the 14 TT.1s built, Svensk Flygjänst ordered twelve of them, the first of which arrived in early 1949. The Swedish company later ordered four more TT.1s which remained in service until late 1963. Denmark also ordered a pair of TT.1s and later converted four ex-Royal Canadian Navy Mk 1s as well. The Indian Navy is also believed to

have also operated at least four TT.1s.

The most successful of the TT family was the TT.4, the first of which entered service with 771 Squadron at Lee-on-Solent in November 1951. When 771 Squadron retired the type in August 1955, the type was transferred to the strength of 700 Squadron which had just reformed at Ford. The squadron continued to fly the TT.4 until February 1957 when target towing duties were placed in civilian hands with Airwork FRU. The FRU operated the TT.4 until December 1958. The TT.4 also served with the Indian Navy from 1955 to 1958.

The two TT.5s first entered service with 723 (RAN) Squadron in November 1954 until October 1956. 851 (RAN) Squadron is also credited with operating at least one of the TT.5s while 725 (RAN) Squadron is 'officially' recorded as using the type from January 1958 to May 1959. One of these was WB271 which returned to the UK to serve with RNHF.

All four TT.6s were also operated in Australia by the RAN, initially with 725 (RAN) Squadron at Nowra from January 1958 to May 1959 and finally with 724 (RAN) Squadron from November 1962 to March 1966, also based at Nowra.

PRODUCTION

14 Firefly TT.1s were initially ordered, twelve of them for Sweden and two for Denmark. Later, Sweden ordered four additional aircraft while five were delivered to the Indian Navy. 28 TT.4s were ordered/converted including five for the Indian Navy. VX388 and WB271 were converted to TT.5s and WB518, WD826, WD828 and WJ109 were converted to TT.6 standard.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA FIREFLY TT.1

ENGINE: One 1,730hp Rolls-Royce Griffon IIB

WING SPAN: 44ft 6in

LENGTH: 37ft 7in

HEIGHT: 12ft 4in

LOADED WEIGHT: 14,020lb

MAX SPEED: 316mph at 14,000ft

MAX RANGE:

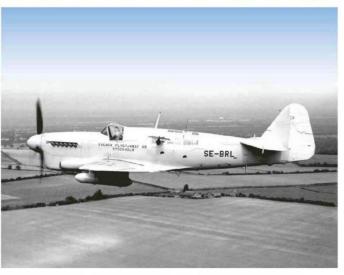
1,300 miles



Converted from a Firefly FR.4 in 1951, TT.4, TW722, '513/FD' is pictured during its service with 771 Squadron at Ford. *Via Martyn Chorlton*



Originally delivered to RDU Culham in 1947 as an FR.4, VH127 was converted to a TT.4 in 1952. Ten years later, the aircraft was retained for 'historical purposes', first arriving at the FAA Museum, Yeovilton, in 1972. Placed in storage at Wroughton in 1988, VH127 returned to Yeovilton four years later and today is on display in Hall 1. *Aeroplane*



Svensk Flygtjänst AB operated 16 Firefly TT.1s, including SE-BRL between January 1949 and October 1963. *Aeroplane*

GANNET (AKA THE FAIREY 'Q', GR.17 & FAIREY 17)



The first FAA machine to combine search and strike

- >>> AUG 12, 1946
 Contract awarded
 for two prototypes
- >>> JUL 19, 1946
 Third prototype ordered
- >>> SEP 19, 1949

 First flight of first prototype, VR456
- >>> JUL 6, 1950 First flight of second prototype, VR557
- >>> JUN 19, 1950 VR546 lands on HMS Illustrious
- >>> MAY 10, 1951
 Definitive third
 prototype, WE488
 flies for first time

DEVELOPMENT

As the range and capability of enemy submarines increased during the latter stages of the Second World War, the necessity to develop an anti-submarine aircraft increased in priority. The Swordfish found itself carrying out this vital role from fleet and smaller aircraft carriers but it was obvious that a newer, more dedicated machine was needed.

The remit was covered in Specification GR.17/45 which called for a two-seat carrier-borne aircraft fitted with a powerful search radar. This radar was to be capable of detecting very small targets such as a snorkel or a conning tower at long range. As well as finding the submarine, the aircraft would also have to have the capability of destroying it, either on the surface or submerged with a variety of weapons, including an airborne homing torpedo.

DESIGN

Designs for GR.17/45 were presented by Blackburn, Fairey and Short and two prototypes from each company were ordered. H E Chaplin led the Fairey design team which initially designed its latest aircraft around the Rolls-Royce Tweed double-propeller-turbine engine. The Tweed was dropped in 1947 in favour of a pair of Armstrong Siddeley Mamba turbines which were coupled together. The prototype was fitted with a 2,950 ehp Double Mamba A.S.M.D.1 which drove a pair of co-axial, counter-rotating propellers through a single gearbox which had a clutch gear that enabled the pilot to shut one engine down. By doing this, the range could be dramatically extended while the aircraft had the inbuilt safety and load-carrying ability of two engines but remained small enough to operate from an aircraft carrier.

Initially known as the Fairey'Q' or '17', the aircraft had a large weapons bay which was later made even bigger to accommodate a pair of homing torpedoes. The first two prototypes were designed as two-seaters and it was not until the arrival of the third, that the aircraft, which would become known as the Gannet, was fitted with a third crew position at the rear.

SERVICE

The first prototype, VR546, was first flown by Gp Capt R G Slade from Aldermaston on September 19, 1949. The flight testing phase that followed was not an easy one for the Gannet which was hampered by a host of handling problems which were caused by the wide flight envelope demanded. The problem was not the layout of the engines but the way the power was delivered through a large range of settings, and this was further complicated by various flight conditions when the aircraft's Fairey-Youngman flaps were employed.

By the time that the third prototype, WE488, was ordered in July 1949, the majority of the handling problems had been illuminated. However, further trials would be needed as the aircraft now had a third cockpit and a larger weapons bay.

Trials on board HMS *Illustrious* and *Albion* in 1951 exposed further problems but, by then, a super-priority order had been placed for 100 production aircraft.

PRODUCTION

Two Gannet prototypes, VR546 and VR557, ordered on August 12, 1946 to contract 6/Acft/494/CB.9(b) to Specification GR.17/45. One Gannet prototype, WE488, ordered July 19, 1949 to contract 6/Acft/3546 and the same specification.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA GANNET AS.1

ENGINE:

One Armstrong Siddeley Double Mamba (A.S.M.D.1)

WING SPAN:

54ft 4in

LENGTH:

43ft

HEIGHT:

13ft 81/2in

EMPTY WEIGHT:

15,069lb

LOADED WEIGHT:

19,600lb

MAX SPEED:

310 mph



The second prototype Gannet, VR557, which first flew from Heathrow (ex-Great West aerodrome) on July 6, 1950. The aircraft flew a host of trials, including its final testing of the aircraft's performance when entering a safety barrier which caused sufficient damage to render the aircraft as a ground instructional airframe by May 1956. Aeroplane



The prototype Gannet, VR546, was used for an intensive period of tests and trials from September 1949 through to August 1956. This important aircraft came to an undignified end on the dump at Donibristle. Via Martyn Chorlton



Ordered in July 1949 to a separate contract, WE488 was the three-seater prototype although the rear cockpit was a mock-up at this stage. The aircraft first flew from White Waltham on May 10, 1951. Via Martyn Chorlton

FIREFLY AS.7 & T.7



A stop-gap for the Gannet

- » MAY 22, 1951
 AS.7 makes
 maiden flight
- >>> OCT 16, 1951
 First production
 AS.7 flies
- **MAR 1953**T.7 joins 719
 Squadron
- DEC 1953
 Last T.7 delivered to FAA
- >>> DEC 1957
 Retired from
 FAA service

DEVELOPMENT

The Firefly AS.7 came about because of an urgent need for an anti-submarine aircraft pending the arrival of the Gannet. However, the AS.7 was never designed to actually destroy the offending submarine, merely having the capability to find it. In the end the anti-submarine role was temporarily and effectively filled by the Avenger AS.4.

DESIGN

The AS.7 differed in many areas from the AS.6 but its general appearance resembled the Firefly F.1 because of the installation of the water-methanol injected Griffon 57 (and later Griffon 59) engine which needed the 'beard' type radiator. The latter removed the need for the leading edge radiators which were introduced by the FR.4. The rear observer's cockpit was another key difference, it was increased in size to accommodate a pair of radar operators under a large, bulbous canopy. A powerful ASV Mk 19A radar set was also installed; its scanner being accommodated in a radome under the starboard wing while a similar nacelle was used as an auxiliary fuel tank under the port wing.

The AS.7 retained the original F.1 full-span wings but was also fitted with an increased area fin and rudder. As mentioned, it had not been planned for the aircraft to have any striking power once in service but it did have the capability to carry a wide range of stores under the wings. It was not fitted with any defensive armament.

The first AS.7 made its maiden flight on May 22, 1951, followed by the first production aircraft, WJ146 on October 16. Preliminary trials and flight testing were not good for the AS.7, especially with regard to handling, which proved to be poor especially at the low speeds needed to land on an aircraft carrier. Despite a great deal of effort to try and rectify the problems, it was clear that the AS.7 was not fit for purpose. By this time an order for 337 aircraft had been placed and these were produced as shore-based T.7 dual-control trainers, of which 151 were eventually built. Many of these were later converted into U.8 drones.

OPERATIONAL SERVICE

The Firefly T.7 first entered service with 719 Squadron, the Naval Air Anti-Submarine School at Eglinton, under the command of Lt Cdr R H W Blake in March 1953. 750 Squadron received the type next at Culdrose in April 1953, then 796 Squadron at St Merryn in June 1953, followed by 765 Squadron, also at Culdrose in February 1955 and finally 1840 Squadron at Ford in March 1956. The type was eventually superseded by the Gannet T.2. 796 Squadron were the last to relinquish their Firefly T.7s in December 1957.

PRODUCTION

151 Firefly 7s were built, the vast majority them T.7s, 110 of them at Hayes and the remainder at Heaton Chapel.



TECHNICAL DATA FIREFLY AS.7

ENGINE: One 1,925hp Rolls-Royce Griffon 59

WING SPAN: 44ft 6in

LENGTH: 38ft 3in

HEIGHT: 13ft 3in

WING AREA: 342 sq ft

EMPTY WEIGHT:

11,016lb

LOADED WEIGHT:

13,970lb

MAX SPEED: 300 mph

at 10,750ft

CRUISING SPEED:

257 mph

INITIAL CLIMB:

1,500ft/min

CEILING: 25,500ft

MAX RANGE:

860 miles at 166 mph



Delivered to Handling Squadron, RAF Manby on October 2, 1952, AS.7 WJ154 was transferred to RDU Anthorn in May 1953, then AHU Lossiemouth in October 1956, only to be retired in August 1958 with just 27.20 flying hours on the clock. Via Martyn Chorlton



The first prototype AS.7, WJ215, pictured on the day of its maiden flight on May 22, 1951. Via Martyn Chorlton



Photographs of AS.7s actually flying for an FAA unit are rare. This pair are from 719 Squadron based at Eglinton, Northern Ireland. Nearest to the camera is WK368, which only served the unit from March 1953 to September 1954 following a taxying accident. Via Martyn Chorlton

F.D.1 (E.10/47, TYPE R OR THE FAIREY DELTA ONE)



Britain's first vertical take-off jet delta

- >>> MAY 1, 1949
 First successful model launch at Woomera
- >>> MAY 12, 1950 First high-speed taxy run at Ringway
- MAR 12, 1951 F.D.1 VX350 makes maiden flight from Boscombe Down
- **MAY 2, 1953**VX357 cancelled
- >>> SEP 1954
 VX350 performs
 at the SBAC
- >>> FEB 2, 1956 VX350 swung on landing and written off

DEVELOPMENT

With its long history of producing naval aircraft behind it, Fairey had been considering the design of a delta-wing fighter vertically launched from warships for quite some time. Specification E.10/47 followed, calling for a delta-winged research aircraft which could test the theories that Fairey had already researched. The company's response was the Fairey Delta One (F.D.1), originally referred to as the Fairey Type R.

Prior to the construction of the full sized machine, a large number of models with a span of 10ft were built and at first launched from Aberporth. However, following of couple of potentially dangerous incidents involving the use of hydrogen peroxide and methanol hydrazine which if mixed was highly explosive, test launches were continued from the tank-landing craft HMS *Sulva* out in Cardigan Bay.

These small but potent models were powered by a single Fairey-built Beta 1 bi-fuel engine with twin combustion chambers capable of producing up to 1,800lbs of thrust. Thrust for the first stage of the launch was provided by a pair of 600lb rockets. The first Fairey model was launched from HMS *Sulva* on May 1, 1949 but this situation was not ideal and the remainder of the trials was transferred to the Long-Range Weapons Establishment at Woomera in Australia.

DESIGN

The full size F.D.1 was still not a larger aircraft, its span only being 19ft 6ins. Powered by a single Derwent engine, the aircraft also had the provision to be fitted with a pair of booster rockets as per the models, but these were never installed.

The F.D.1 was a 'tubby' looking mid-wing delta with a

large intake in the nose for the engine to breathe. The aircraft was fitted with a tricycle undercarriage, despite being originally designed with a jettisonable one (the aircraft landing on a skid), which cleverly concealed itself within the circular fuselage through the main gear retracting outwards and then rearwards through 45°.

Control, both fore-and-aft and lateral was by powered elevons fitted to the trailing edge of the wing. A big powerful rudder was also fitted and, as well as conventional brakes, a drogue chute was fitted for landing. If the aircraft got into trouble in flight, a pair of spin-recovery chutes were fitted into the wing tips.

SERVICE

The first of three F.D.1s ordered, VX350, was flown for the first time by Gordon Slade from Boscombe Down for just 17 minutes on March 12, 1951. Prior to this, taxying trials had been carried out at Ringway from May 1950 by Peter Swiss and Slade.

By this time the Air Ministry had lost interest in the potential of a vertical take-off fighter and even before VX350 had flown, a second F.D.1, VX364, had been cancelled. The third aircraft, VX357, was cancelled in May 1953 and like the VX364 had reached an advanced stage of construction. Both were scrapped at Fairey's Stockport factory.

VX350 continued to be used for trials flying and provided Fairey with a huge amount of useful data which proved very useful for the F.D.2, despite the latter having very little in common from an appearance point of view. VX350 was unfortunately written off on February 6, 1956 when it swung on landing, seriously damaging the undercarriage at Boscombe Down. The aircraft was then despatched to P&EE Foulness in October 1956 where it was later scrapped.



TECHNICAL DATA F.D.1

ENGINE: One 3,600lb Rolls-Royce Derwent 8 turbojet

WING SPAN: 19ft 61/2in

LENGTH: 26ft 3in

LOADED WEIGHT:

6,800lb

MAX SPEED:

628 mph at 10,000ft

INITIAL RATE OF CLIMB:

9,300 ft/min

CLIMB:

15,000ft in 1.9 min



Not the most attractive aircraft designed by Fairey, the F.D.1 was, nevertheless, a very useful research vehicle for future projects. Aeroplane



An undignified end for the F.D.1 after the aircraft swung on landing at Boscombe Down on February 6, 1956. Aeroplane



Streaming its drogue parachute, F.D.1, VX350, slows after landing at the 1954 SBAC display at Farnborough. Aeroplane

FIREFLY U.8 & U.9



The Firefly drones on

- >>> 1952
 Heaton Chapel takes over development
- >>> DEC 30, 1954 First U.8 makes maiden flight
- >>> FEB 1955

 First operational U.8 flight
- >>> SEP 29, 1955
 First U.8 shoot
 down with a
 Firestreak
- >>> DEC 13, 1956 U.9 makes maiden flight
- >>> NOV 29, 1961
 The last U.9 is shot down

DEVELOPMENT

The story of the Firefly drones began in the late 1940s at Hayes and White Waltham where development had been progressing well. This work was transferred to the Stockport factory in 1952. The idea was to produce an aircraft which could be used to support guided-missile testing which was being carried out in Britain and at Woomera in Australia. A further requirement was that the aircraft had to be capable of being operated by a normal crew or by a remote control system controlled from the ground.

DESIGN

The first of the two drones was designated as the Firefly U.8 which used the T.7 airframe. A Mk 8 autopilot was fitted, which had already been successfully tested by the RAE in an Avro Lancaster. The autopilot was monitored from the ground and all flying adjustments were controlled by radio. The throttle, flaps, propeller and undercarriage were controlled by actuators. The arrestor hook was not originally fitted to the T.7 but it proved a useful piece of equipment, especially when a drone returned to base under remote control. Only the most basic of flying instruments and controls were left in place for a 'real' crew. A pair of nacelles were attached to each wingtip, containing four cine cameras in each, for recording the effects of a missile strike or near miss.

By 1955, the supply of U.8 drones was running low because of successful missile strikes and various flying accidents. The Ministry of Supply placed an order for 40 additional drones and, to accommodate this, Fairey offered to convert surplus FR.5s. These were designated as the Firefly U.9 which used the same autopilot as the U.8. Conversion work began in 1956.

OPERATIONAL SERVICE

The first U.8 made its maiden flight from Ringway on December 30, 1954. The majority of U.8s served the RAE from Llanbedr and it was from there that the first operational flight took place in February 1955. Flying the U.8 was a complex and expensive operation which involved a second Firefly flying as a 'shepherd' with its own remote control equipment and two further sets of equipment on the ground to take over when landing the drone.

On September 29, 1955, the first U.8 was shot down by a de Havilland Venom launching a Firestreak missile. U.8s also played a key role in the development of the Armstrong Siddeley Seaslug surface-to-air missile and several more were claimed by the fledgling Sidewinder.

The first U.9 took to the air on December 13, 1956 and, by January 1958, had entered service with 728B Squadron, based at Stretton, alongside a few remaining U.8s. On March 1, 1958, the unit relocated to Hal Far from where the first remote control flight was carried out on July 8. Like the U.8 before it, the U.9 played an active role in the development of the Seaslug which was being fired from RFA Girdle Ness.

By 1959, the Fireflies were supplemented by Meteor drones and finally by Canberras in May 1961. U.9s were also used by 728B Squadron for Firestreak testing in later 1958, all of which were fired by Sea Venoms of 893 Squadron from HMS *Victorious*. By late 1961, the final few U.9s were being despatched by anti-aircraft guns from Royal Navy warships; the last being shot down on November 29, 1961 by the guns of HMS *Duchess*.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA FIREFLY U.8 & U.9

ENGINE: (U.8) One 1,925hp Rolls-Royce Griffon 59; (U.9) Óne 2,100hp Griffon 74

WING SPAN: (U.8) 44ft 6in; (U.9) 41ft 2in

LENGTH: (U.8) 38ft 3in (U.9) 38ft

HEIGHT: (U.8) 13ft 3in (U.9) 13ft 11in

EMPTY WEIGHT: (U.8) 11,016lb (U.9) 9,674lb

MAX LOADED WEIGHT:

(U.8) 13,970lb (U.9) 13,479lb

MAX SPEED: 300 mph

MAX RANGE:

Approximately 700 miles



The very first U.8 completed at Stockport was WM810 which first flew from Ringway on November 27, 1953. Despite crashing into the sea off Aberporth in September 1957 the drone was repaired, only to become a total loss while operating from Llanbedr in November 1959. Via Martyn Chorlton



Nice angle of U.8, WM856. The pods on the wing tips contained four cameras. Via Martyn Chorlton



Originally built as an FR.5, WB257, was converted into the U.9 prototype in May 1955. Whilst serving with 728B Squadron at Hal Far, the aircraft was shot down of Malta on May 25, 1960. Via Martyn Chorlton

GANNET AS.1



- >>> JUN 9, 1953
 First production
 aircraft, WN339, flies
- >>> OCT 1953
 WN341 begins
 carrier trials with
 Illustrious and Eagle
- >>> MAR 15, 1954 703X Sqn formed
- APR 5, 1954 First four Gannets 'officially' handed over to the FAA
- >>> OCT 5, 1954
 First Stockport-built
 Gannet demo at
 Ringway
- >>> JAN 17, 1955

 First operational

 Gannet squadron formed

The FAA finally gets its sub-killer

DEVELOPMENT

Within two years of the priority production order for 100 Gannets being placed, the first production aircraft were already in the air but it was still more than year before the type reached front-line service. To fill the Royal Navy's capability gap, 100 Grumman Avenger AS.4s were ordered through the Mutual Defence Aid Programme, the first entering FAA service in May 1953, and many were still in front-line service up to 1955, eventually being replaced by the Gannet.

DESIGN

Compared to the prototypes, the Gannet AS.1 was powered by a Double Mamba 100 engine. It had its main undercarriage re-positioned and its nose wheel revised. The connection between the flaps and the tailplane incidence gear was also improved so that the aircraft's trim was automatically adjusted as the flaps were raised and lowered.

The first production Gannet AS.1, WN339, was first flown by Peter Twiss from Northolt on June 9, 1953. WN339 was later flown to White Waltham for test work and the installation of operational equipment. The third production aircraft, WN341, was used for carrier trials on HMS *Illustrious* and *Eagle* during October 1953, which included night flying.

OPERATIONAL SERVICE

The Gannet first entered FAA service on April 5, 1954, when four aircraft, WN347 to WN350, were handed over to 703X Flight (part of 703 Squadron Service Trials Unit) under the command of Lt Cdr F E Cowtan at Ford.

The first operational unit to receive the Gannet AS.1 was 826 Squadron at Lee-on-Solent under the command of Lt Cdr G F Birch in January 1955. By early June, the unit's eight Gannet AS.1s had embarked aboard the recently updated HMS *Eagle* for a cruise of the Mediterranean.

Fifteen AS.1s also joined the Royal Australian Navy, serving with 816 and 817 Squadrons on board HMAS *Melbourne* and *Sydney* respectively.

Eventually, nine operational and one reserve FAA squadron were re-equipped with the Gannet AS.1 which served until the late 1960s with the Royal Australian Navy but was quickly superseded by the AS.4 with front-line FAA units during the mid to late 1950s. The following units also operated the Gannet AS.1; 700, 703, 719, 724, 725, 737, 744, 796, 812, 815, 820, 824, 825, 831, 847 and 1840 Squadrons.

PRODUCTION

180 Gannet AS.1s were built, 116 of them at Hayes and 64 at Stockport, within the serial ranges WN339 to WN464, XA319 to XA436, XG784 to XG826 and XG898.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA GANNET AS.1

ENGINE: One 2,950 ehp Double Mamba 100 coupled-turboprop driving co-axial counter-rotating four blade propellers; WN340 was fitted with a Double Mamba Mk 3 engine

WING SPAN: 54ft 4in

LENGTH: 43ft

HEIGHT: 13ft 81/2in

EMPTY WEIGHT:

15,069lb

LOADED WEIGHT: 19,600lb

MAX SPEED: 310 mph



AS.1 WN391 being test flown from RDU Anthorn during late 1954 prior to delivery to 824 Squadron at Eglinton. Via Martyn Chorlton



Gannet AS.1, WN346, was delivered to Handling Squadron, RAF Manby, in March 1954 and remained when the unit moved to Boscombe Down the following month. Via Martyn Chorlton



The Royal Australian Navy ordered 36 Gannet AS.1s, including XA326, which served with 817 Squadron aboard HMAS *Melbourne* from August 1955 and then completed two tours of duty with 816 Squadron. Via Martyn Chorlton

JET GYRODYNE



Testing the compound helicopter theory

- >>> JAN 1954
 First tethered flight
- >>> MAR 1, 1955
 First successful transition
- >>> SEP, 1955

 Demonstrated at SBAC
- >>> SEP, 1956
 End of exhaustive test period
- >>> 1961 Aircraft SOC

DEVELOPMENT

By late 1953, the second Gyrodyne prototype re-emerged in a completely different guise. Behind the scenes, a large amount of testing had been carried out at White Waltham on static rigs to prove the principles of a tip-jet rotor-driven system. A contract was already in place from the Ministry of Supply (MoS) to carry out further research into the system and it was time to test the idea on a live aircraft.

This was essential, not only to continue development and testing of the tip-jet, but also to test the aircraft's handling and to begin the lengthy process of creating the procedures for the operation of a convertible or compound helicopter. Named the Jet Gyrodyne in early 1954 and serialled XD759 (duplicated serial shared with a Canadair Sabre F.4), later changed to XJ389, the aircraft made its first tethered flight at White Waltham with John N Dennis at the controls. By late January 1954, the aircraft made a 'free' flight for the first time.

DESIGN

Only the fuselage, stub wings, tail unit and tricycle undercarriage of the original Gyrodyne was retained. The Leonides engine drove a pair of variable-pitch pusher propellers via gearbox and shafts in the stub-wings. The propellers gave propulsion in the cruise and at lower speeds and directional control was achieved via the rudder pedals thanks to a differential pitch-change placed on top of the collective-pitch action. This is where the similarity to the original Gyrodyne comes to an end with a third drive direct from the gearbox to a pair of

Rolls-Royce Merlin centrifugal compressors mounted together under the main rotor pylon. These compressors provided air to the tip burners at the ends of the two-blade rotor.

While the theory seemed easy to convey, in practice flying the aircraft was very difficult, aggravated by the fact that the Jet Gyrodyne weighed in at a hefty 2,720kg. This meant that the Leonides engine had to be operated at full boost and, even then, the aircraft could only momentarily maintain level flight. It was not until March 1,1955 that John Dennis managed to carry out a successful transition from a vertical take-off to a level flight cruise; the first such exercise in any aircraft.

SERVICE

From that first transition in March 1955 through to September 1956, the Jet Gyrodyne carried out 190 transitions and 140 auto-rotative landings. By then, the techniques needed to carry out an inflight tip burner relight were now understood and could be carried out with relative ease. While this stage had taken 18 months of hard work by highly experienced test-pilots, once the correct procedures were in place, the reward came when half a dozen MoS pilots were taught how to fly the Jet Gyrodyne after only one hours of instruction.

The aircraft paved the way for the Rotodyne and, when ground testing of its bigger younger sibling began, the Jet Gyrodyne was retired. Earmarked for the scrap man in 1961, the Jet Gyrodyne was saved and today resides in the Museum of Berkshire Aviation.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA JET GYRODYNE

ENGINE: One Alvis Leonides nine-cylinder radial engine and two wing-tip compress air/ fuel burning jet engines

ROTOR DIAMETER:

51ft 9in

FUSELAGE LENGTH:

HEIGHT: 10ft 2in

EMPTY WEIGHT: 3,600lb

GROSS WEIGHT: 6,000lb

MAX SPEED: 140 mph



By the time the Jet Gyrodyne made its biggest public appearance at the 1955 SBAC Display at Farnborough, the aircraft had been re-serialled as XJ389. Via Martyn Chorlton



XJ689 about to begin another routine at the 1955 SBAC, where the transition cycle was publically demonstrated for the first time. Whilst practising for the display, a total of 65 in-flight tip-jet re-lights were carried out over an eight day period. Aeroplane



With its twin engine-driven propellers in the pusher configuration and jet-driven rotor, the Jet Gyrodyne was an advancement of the basic Gyrodyne theory. Aeroplane



The first big carrier borne trainer

- >>> AUG 16, 1954
 Maiden flight
 of WN365
- FEB 1955
 First production
 aircraft delivered
 to 737 Squadron
- >>> AUG 15, 1967
 T.2 retired from
 RAN service

DESIGN

The prototype Gannet T.2, WN365, was allotted for conversion on March 12, 1954 from an AS.1 on the production line at Hayes. After the work had been carried out, the aircraft was transported to White Waltham by road on June 8, 1954 and was first flown on August 16. Fitted with dual controls in the front two cockpits, the instructor in the rear had access to a periscope which retracted when the forward canopy was opened. The radar and its supporting equipment were removed while the spare rear cockpit could be used by a radio operator or to carry a pair of passengers.

OPERATIONAL SERVICE

The first of 36 production aircraft, XA508, joined 737 Squadron at Eglinton under the command of Lt Cdr D W Pennick, in February 1955 and remained on strength until the unit disbanded in November 1957. The Gannet T.2 also served the following FAA units; 700, 719, 725, 728, 796, 812, 820, 824, 825 and 1840 Squadrons. The majority

of these units had given up the type by the late 1950s, only 725 Squadron kept the T.2 into the next decade, retiring its aircraft in May 1961.

The Gannet T.2 also served the Royal Australian Navy. Three aircraft remained on strength until the late 1960s with 724 and 816 Squadrons. Two T.2s, ex-XA521 and ex-XG574, were refurbished by Fairey for service with the Indonesian Navy as LA-18 and LA-17 respectively.

While some minor incidents did occur during the T.2's decade of service, not one aircraft was destroyed in a flying accident, which is quite remarkable for a trainer.

PRODUCTION

37 Gannet T.2s were built at Hayes and flown from White Waltham. The prototype, WN365, a converted AS.1, and two production batches of 23 and 13 aircraft were built. The first batch, XA508 to XA531, (c/n F.9328 to F9350) was ordered to Contract 6/Acft/8203/CB.9(a) on May 7, 1952 and the second, XG869 to XG881 (c/n F.9398 to F.9410) ordered to Contract 6/Acft/10545/CB.9(A) on June 25, 1954.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA GANNET T.2

ENGINE: One 2,950 ehp Double Mamba 100 (A.S.M.D.1)

WING SPAN: 54ft 4in

LENGTH: 43ft

HEIGHT: 13ft 81/2in

EMPTY WEIGHT: 15,069lb

LOADED WEIGHT: 19,600lb

MAX SPEED: 310 mph



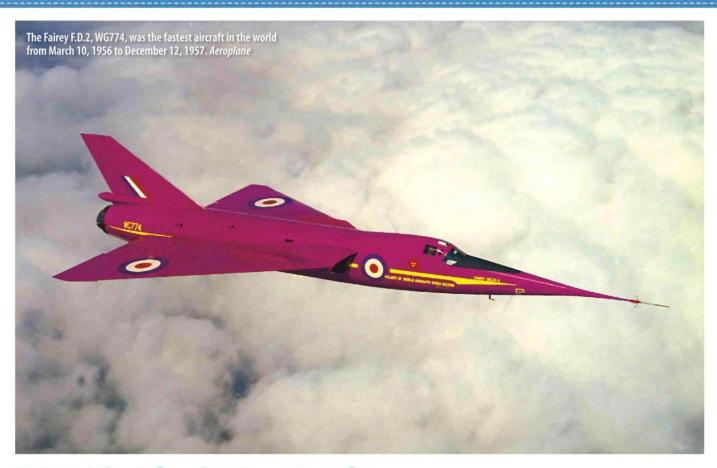
The prototype Gannet T.2, WN365, being demonstrated at the SBAC show at Farnborough in September 1954, only days after its maiden flight. Via Martyn Chorlton



Gannet T.2, XA515, pictured at Andover on September 20, 1958 whilst serving with the ETPS which ended in September 1960 when the aircraft was allocated for fire-fighting practice on Farnborough's dump. Via Martyn Chorlton



The third production T.2, XA510, from the first batch of 24 aircraft which were built at Hayes and first flown from White Waltham. First flown on January 1, 1955, XA510 was retired in 1968 having flown only 356.45 flying hours. Via Martyn Chorlton



Fairey's last fixed-wing aircraft

- >>> OCT 1950

 Contract placed for two prototypes
- >>> 1952
 Design work begins
- >>> OCT 6, 1954
 First flight of
 WG774 from
 Boscombe Down
- >>> FEB 15, 1956 Second F.D.2, WG777 flies for the first time
- >>> MAR 10, 1956 WG774 captures World Absolute Speed Record
- >>> MAY 1, 1964
 FD.2 reincarnated
 as the BAC 221

DEVELOPMENT

During the post-war period, Britain was lacking with regard to supersonic aircraft design. However, the Ministry of Supply attempted to rectify this by issuing Specification ER.103 which called for an experimental aircraft which could reach Mach 1.5 at 36,000ft; English Electric proposed their P.1 and Fairey their Delta Two. While P.1 evolved into the successful Lightning, the F.D.2 would be used to investigate aircraft performance at both transonic and supersonic speeds.

DESIGN

A contract for a pair of aircraft was placed in October 1950 but, because all focus was on Gannet production at the time, design work on the F.D.2 did not begin until mid-1952.

Designed with a wing which had the lowest thickness to chord ratio that had ever been attempted before, the F.D.2, like its predecessor, was of a delta wing arrangement and that's where all similarities end. The aircraft's very narrow fuselage was designed around the Avon powerplant and the outer skin of the F.D.2 was never any more than five inches from the exterior of the engine.

One of the many novel features of this outstanding aircraft was that the entire nose, cockpit and all, could be dropped through 10° to give the pilot a better field of view when the aircraft was being flown at a high angle of attack. This same system was later adopted for Concorde.

SERVICE

The first aircraft, WG774, took to the air in the hands of Peter Twiss from Boscombe Down on October 6, 1954. Twiss rapidly gained confidence in flying the F.D.2 which he described, very early on, as having great promise. By November 17, Twiss was carrying out the aircraft's fourteenth flight when at 30,000ft and approaching Mach 0.9, the Avon engine became starved of fuel because of a system fault. Despite being 30 miles from Boscombe Down, Twiss managed to get WG774 back to the airfield although some damage was caused in the subsequent crash landing which eventually delayed the flight programme for eight months.

Fitted with a new wing and various other modifications, the F.D.2 was back in the air again in March 1955 and, by October, had finally gone supersonic. These early flights beyond the sound barrier were actually achieved without the use of re-heat (still in its infancy) or maximum throttle. By the following month, the F.D.2 was recording speeds of Mach 1.56 (1,028mph) at 36,000ft and beyond with ease so it was decided that the aircraft should make an attempt on the world speed record.

On March 10, 1956, Peter Twiss flew WG774 along a course 9.7 miles in length between Chichester and RNAS Ford. Flying at 38,000ft, eight pairs of runs were flown recording an average speed of 1,132mph, which was 310mph faster than the previous record. This outstanding achievement stood until December 12, 1957 when an F-101A Voodoo recorded a speed of 1,208mph.

The second aircraft, WG777, was first flown, again by Twiss from Boscombe on February 15, 1956 but by this time the Air Ministry was losing interest in the development of high speed fighters. Both aircraft went on to carry out valuable flight testing, especially WG774 which later became the BAC 221 to play a crucial role in the development of Concorde. Both machines survive today, WG774 at the FAA Museum and WG777 at the RAF Museum at Cosford.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA F.D.2

ENGINE: One 10,000lb Rolls-Royce Avon 200 WING SPAN: 26ft 10in

LENGTH: 51ft 71/2in

HEIGHT: 11ft

WING AREA: 360 sq ft

EQUIPPED EMPTY WEIGHT: 11,000lb

MAX TAKE-OFF WEIGHT: 13,884lb MAX SPEED: (low) 748 mph; (above 36,000ft) Mach 1.7 (1,222 mph)

INITIAL CLIMB: (without after-burner) 4,750ft ft/ min; (with after-burner) 15,000 ft/min

TIME TO 40,000ft: Climbing at Mach 0.93, 2.5 min

MAX RANGE: (without after-burner) 830 miles



The Fairey F.D.2, WG774, pictured en route to the 1956 SBAC air show at Farnborough. Aeroplane

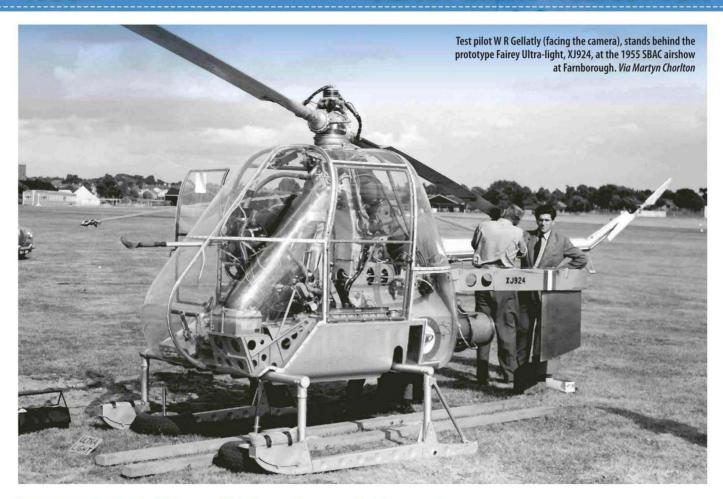


WG774 in the landing configuration at Farnborough in 1957 gives away the aircraft's 'drooped snoot' and the petal-type air brakes, which are partially opened around the jet pipe. Aeroplane



Staged after the event of the F.D.2 breaking the World Absolute Speed Record, Peter Twiss is congratulated by Robert L Lickley, Fairey's chief engineer, with Maurice Childs the chief flight development engineer looking on. Aeroplane

ULTRA-LIGHT HELICOPTER



TECHNICAL DATA ULTRA-LIGHT

ENGINE: One Turboméca Palouste BnPe.2 turbojet **ROTOR DIAMETER:** 28ft 31/2 in **FUSELAGE LENGTH: 15ft** HEIGHT: 8ft 2in LOADED WEIGHT: 1,800lb **VERTICAL RATE OF CLIMB:** 950 ft/min MAX RATE OF CLIMB: 1,350 ft/min at sea level **HOVERING CEILING:** 4,800ft CRUISING SPEED: 95 mph MAX RANGE: 180 miles MAX ENDURANCE: 2 hr 30 min

AUG 14, 1955XJ924 first flight

>>> OCT 1958

Aircraft demonstrated by Peter Twiss

>>> 1959
Project abandoned

Keep it simple and cheap!

DEVELOPMENT

In 1953, the War Office, in collaboration with the Air Ministry and the MoS, began compiling a specification for a straightforward, inexpensive helicopter which the Army could use for reconnaissance, casualty evacuation and training duties. The criteria was demanding and included the ability to be part-dismantled and be transported on a three-ton truck.

Half a dozen manufacturers responded to the original specification, H.144T, which was won by Fairey in July 1954. Named the Ultra-light, Fairey incorporated the same rotor-tip drive system which they had already developed for its Jet Gyrodyne and the forthcoming Rotodyne.

DESIGN

Power for the small helicopter was provided by a Palouste BnPe.2 engine which was being built under licence by Blackburn and General Aircraft. Thrust was diverted to the tip-jets via an oversized centrifugal compressor. The compressed air travelled through a lagged duct to the rotor head and, combined with fuel, continued on to the tip-jets.

The prototype used a direct tilting-head control but this was later replaced by a hydraulically-powered system for the cyclic-pitch which was also fitted to later Ultra-lights. Flying controls were the same as any other helicopter, the collective-pitch lever carried a twist-grip throttle for raising the engine rpm which simultaneously increased the air pressure to the tip jets. A normal stick was used for roll and pitch and rudder pedals were used for directional control via a steel-skinned rudder.

SERVICE

The prototype, XJ924, made its maiden flight from White Waltham in the hands of W R Gellatly on August 14, 1955, a mere 13 months since design work had begun. Unfortunately, by mid-1956 the MoS had lost interest (for economic reasons) but Fairey continued to develop it as a private venture. During 1957, the Piasecki Aircraft Corporation in the USA showed a great deal of interest in what Fairey was achieving which, in turn, led to the US Army evaluating the Ultra-light although nothing came of it.

The second Ultra-light, XJ928, was used by Fairey as a development aircraft with a different cabin which, among other things, was modified to carry a single stretcher. XJ928 was later registered as G-AOUJ and fitted with hydraulic controls. The prototype XJ924, had already appeared at the SBAC in 1955 but it was the turn of XJ936 for the 1956 event, where it performed as the main demonstration aircraft. Later aircraft were trialled by the RAE at Bedford, the College of Aeronautics at Cranfield and the Royal Navy. The latter saw G-AOUJ performing trials with HMS *Grenville* in late 1957. The small helicopter successfully carried out 70 landings and take-off in winds up to 62kts, during which time, the deck pitched up to 3.66 metres and rolled through 14°. Frustratingly for Fairey, the entire project was reluctantly abandoned in 1959.

PRODUCTION

Six Ultra-lights were built; the prototype, XJ924, followed by XJ928 (G-AOUJ). XJ930 was delivered direct to the MoS and XJ936 (later G-AOUK) first flew on August 24, 1956. The fifth aircraft was G-APJJ which was evaluated by the Royal Navy in 1958 and the sixth airframe never flew. **GANNET T.5** 1957



Heavier more powerful T-Bird

DESIGN

Like the Gannet AS.4 from which it evolved, the second trainer version of the big anti-submarine aircraft, the T.5, came about because the rising weight of the AS.1 which resulted in the need for a more powerful engine. Basically, the T.5s were converted from the final eight T.2s on the Hayes production line by fitting a 3,035 ehp Double Mamba 101 engine. Only nine were built and the first production aircraft, XG882, flew for the first time from Northolt on March 1, 1957.

OPERATIONAL SERVICE

The Gannet T.5 officially entered service with 849 HQ Squadron at Culdrose in September 1961 although one aircraft, XG887, was briefly on the strength of 719 Squadron at Eglinton on February/March 1959. The T.5 also served with the Station Flight at Culdrose before transferring to 849 Squadron and Station Flight/MTPS at Abbotsinch, NARIU Lee-on-Solent and NASU Brawdy and Lossiemouth. The T.5 remained in service with 849 Squadron until January 1976.

Only one aircraft was lost in service on July 1, 1963 when XG887, being flown by Mr H Proudlove of the Ferry Flight, Rochester, was returning from Shorts, Belfast to Culdrose following modification work. A leaking fuel pipe caught fire after making contact with a hot exhaust, causing an extinguisher to explode which severed more fuel lines. Whilst near Milton, the aircraft exploded, then crashed at Michael-on-Arrow, 15 miles west of Hereford, killing the pilot.

XG890 was the only aircraft to see foreign service, being despatched direct to the West German Navy to serve as UA+99. Of the remainder, XG882 survives in a sorry state at Errol, XG883 is on loan from the FAA Museum to the Berkshire Air Museum, while parts of XG889 live on in XG882.

PRODUCTION

Nine T.5s ordered on June 25, 1954 under Contract 6/ Acft/10545/CB.9(a), serialled XG882 to XG890. Only eight were built as XG888 was produced as a T.2. XG890 was supplied to the West German Naval Air Arm as UA+99.

WING SPAN: 54ft 4in LENGTH: 43ft HEIGHT: 13ft 81/2in EMPTY WEIGHT: 14,069lb LOADED WEIGHT: 23,446lb MAX SPEED: 299 mph INITIAL CLIMB: 2,000 ft/min CEILING: 25,000ft RANGE: 662 miles ENDURANCE: 4.9 hr at 150 mph

>>> MAR 1, 1957

First flight of XG882 from Northolt

>>> SEP 1961

T.5 enters service with 849 Squadron

>>> JAN 1976

T.5 retired from FAA service

GANNET AS.4 (INCLUDING COD.4 & ECM.6)



The increasing weight of the final anti-submarine variant

- >>> MAY 7, 1954
 First production
 order for AS.4 placed
- >>> MAR 12, 1956 First flight of AS.4 prototype, WN372
- >>> AUG 1956
 Enters service with
 824 Squadron
- >>> 1958
 AS.4s supplied to the Kriegsmarine
- >>> MAY 16, 1966 ECM.6 retires from Royal Navy service
- >>> SEP 1974
 COD.4 retired from the FAA

DESIGN

The Gannet AS.4, its training variant, the T.5 and its sub-variants, the COD (Carrier-on-board).4, the ECM (Electronic Counter Measures). 4 and Mk 6 all came about because of the rise in weight and the demands being placed upon the Gannet AS.1. More power was needed, and this was provided by 3,035 ehp Double Mamba 101 engine which was first installed in ex-AS.1, WN372. After returning from hot weather trials at Idris, the aircraft re-emerged from the White Waltham factory as the prototype AS.4 and first flew in its new guise from Northolt on March 12, 1956.

The first production aircraft, XA412, quickly followed and less than a month later ex-AS.1s, XA410 and XA411. By July 1956, the prototype AS.4 had completed its hot weather trials, again at Idris, only weeks before the type entered operational service.

Six AS.4s were later converted to the COD.4 which was used as a VIP transport and very light cargo aircraft and, more importantly, to the crew of an aircraft carrier far out at sea, the mail. Further modifications saw the introduction of a handful of ECM variants. At least one aircraft became an ECM.4 fitted with NSM3009 ECM equipment, while six others were upgraded with a more powerful radar and updated electronics with large pods under each wing to initially become the AS.6; these were later redesignated as the ECM.6.

OPERATIONAL SERVICE

The Gannet AS.4 first entered service with 824 Squadron at Eglinton in August 1956, the type then going on to serve aboard HMS *Ark Royal* from January the following year. The AS.4 also joined 700, 810, 814, 815, 825, 847 and 849 Squadrons, the latter until May 1966, by which time, the type had virtually been phased out. In the meantime though, 849 Squadron had been operating the Gannet COD.4 since September 1961, a type that served the unit until September 1974, only being surpassed by the T.5.

The rare ECM variants only served in an operational capacity with 831 (Electronic Warfare) Squadron at Culdrose from February 1959 when the ECM.4 first arrived. Phased out by February 1961, the ECM.4 was superseded by the ECM.6 which arrived the same month and served until May 16, 1966 when the unit was disbanded at Watton and its personnel were moved to 360 Squadron.

PRODUCTION

82 Gannet AS.4s were built, 58 of them at Hayes and 24 at Stockport. Six AS.4s were converted to COD.4s and nine to ECM.6 standard which were originally designated as AS.6s (at least one aircraft, WN464, was converted to ECM.4 standard).



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA GANNET AS.4/6

ENGINE: One 3,035 ehp Double Mamba (A.S.M.D.3)

WING SPAN: 54ft 4in

LENGTH: 43ft

HEIGHT: 13ft 81/2in

EMPTY WEIGHT: 14,069lb

LOADED WEIGHT:

23,446lb

MAX SPEED: 299 mph

INITIAL CLIMB:

2,000 ft/min

CEILING: 25,000ft

RANGE: 662 miles

ENDURANCE: 4.9 hr at 150 mph



First flown from Ringway on October 18, 1956, AS.4, XA435 went on to serve with 814 Squadron but was only destined to be flown for 463.45 hours before it was SOC at Belfast on August 28, 1964. Aeroplane

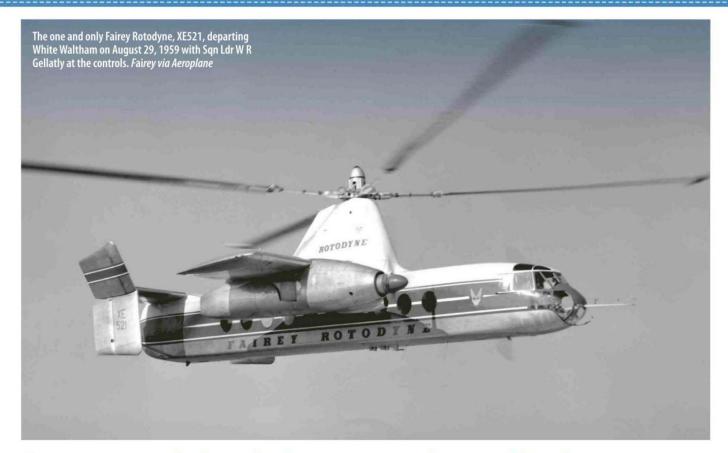


Another Gannet that enjoyed a very busy flying career was XA460. Having served from 1956 as an AS.1, the aircraft was converted to ECM.6 standard in 1962, serving with 831 and finally 849 Squadrons until 1971. The aircraft survives today at Aeroventure in Doncaster. Aeroplane



XA454 was originally built as an AS.1, first flying in October 1956. After service with 814 Squadron aboard HMS Eagle and 849 Squadron at Culdrose, the aircraft was converted to a COD.4 in early 1963. In its new guise, the aircraft gave further excellent service until 1969 when it was SOC but spent the entire 1970s as Yeovilton's gate guard. Via Martyn Chorlton

ROTODYNE



A concept and aircraft that was way beyond its time

- >>> 1947

 First proposal for a large compound helicopter
- >>> NOV 6, 1957
 First untethered flight
- >>> JAN 5, 1959
 Aircraft captures
 absolute straight
 line speed record
- >>> JUN 16, 1959
 Leaves country for the Paris airshow
- >>> FEB 8, 1960
 Fairey Aviation
 and Westland
 Aircraft merge
- >>> FEB, 1962
 Rotodyne
 programme is
 cancelled

DEVELOPMENT

The idea of producing a large commercially viable compound helicopter dated back to a study carried out in 1947 by Dr J A J Bennett and Capt A G Forsyth. The project evolved through several different forms, the aircraft first being referred to as the Rotodyne in 1951. Two years later, the design settled on an aircraft that was to be powered by a pair of Napier Eland engines with tip jet propulsion.

DESIGN

The unique Rotodyne was designed to carry up to 40 passengers with a flight crew of two not including air hostesses/stewards. Beginning with its helicopter roots, the aircraft was fitted with a main rotor of 90ft in diameter which was driven by tip jets fuel by compressed air and fuel via compressors fitted to the rear of the Eland engines. Both air and fuel were fed via the leading edge of the main plane up to the rotor head and then to opposing tip jets. Once a speed of 60 mph was reached, the tip jets were extinguished and the Eland engines took over while the 46ft 6in span main wing took over 50% of the lift away from the rotor blades. This effectively turned the aircraft into a large high-speed autogyro in level flight.

The rotor system was torque-less so there was no need for the corrective tail rotor which is still traditional on helicopters today. To test all of the Rotodyne's systems fully before flight, test rigs were built at White Waltham and Boscombe Down. The latter incorporated a pair of Eland engines with propellers, a shorter version of the main-plane and the entire rotor head with tip jets.

SERVICE

Serialled XE521, the Rotodyne made its first untethered flight at White Waltham on November 6, 1957, in the

hands of W R Gellatly and J G P Morton, who were both destined to see the project through as the main flight test crew from start to finish. The same day, two further flights were carried out around the perimeter of the airfield. For the next five months all flights were carried out as a helicopter but, on April 10, 1958, the first transition from the hover was carried out at 4,000ft, an exercise that benefitted enormously from trials with the Jet Gyrodyne.

By late 1958, the performance of the Rotodyne was clearly good enough for an attempt on a speed record and the most appropriate one, which reflected how the aircraft could be used, was the 100km (62 mile) closed-circuit category. On January 5, 1959, Gellatly and Morton, with Dr D B Leason (flight-test observer) and E J Blackburn (strain-gauge operator) in the back flew a measured circuit between White Waltham and Hungerford. An average speed of 307 km/h (190.9 mph) was recorded, a speed 49 mph faster than the previous straight line record set by a helicopter. This record stood until October 1961 when a Ka-22 Vintokryl convertiplane flew faster.

Interest, by 1960, was growing across the world for the Rotodyne, especially from New York Airways while one of the main intended recipients, BEA, began to go cool on the idea. However, the beginning of the end for the Rotodyne came on February 8, 1960 when Fairey Aviation and Westland Aircraft were merged. Westland was also soaking up Bristol's helicopter division as well, which resulted in the very protracted progress of the Rotodyne from the spring of 1960 onwards.

Sadly, month by month through 1961, all interest in the aircraft began to dwindle and with no sign of further government investment, the Rotodyne was cancelled in February 1962. Yet another great British design had been put out to grass.



Andy Hay/www.flyingart.co.uk

TECHNICAL DATA ROTODYNE

ENGINE: Two 2,800shp Napier Eland N.El.7 turboprops and four 1,000lb rotor tip jets

WING SPAN: 46ft 6in

ROTOR DIAMETER: 90ft

FUSELAGE LENGTH: 58ft 8in

HEIGHT: 22ft 2in

INTERNAL CABIN

LENGTH: 46ft

LOADED WEIGHT:

33,000lb

CRUISING SPEED:

185 mph

MAX RANGE: 450 miles



With the exception of the rotor revolutions, instruments and low-speed-range ASIs, the layout of the Rotodyne flight deck and panels were laid out as if the aircraft was a normal fixed-wing machine. *Flight via Aeroplane*



The Rotodyne in its final configuration with the upper fins of the tail planes now in the fixed vertical position, plus the fitment of ailerons to the 46ft 6in wing. Flight via Aeroplane



Constructed at Hayes, the Rotodyne was dismantled into major components and transported by road to White Waltham on April 8, 1957. Flight via Aeroplane

GANNET AEW.3



Douglas Skyraider replacement

>>> AUG 20, 1958

Prototype, XJ400 makes maiden flight

>>> NOV 1958

Trials aboard HMS Centaur

» MAY 1959

Continued trails aboard HMS Victorious

>>> AUG 1959

Three aircraft delivered to 700G Squadron

>>> DEC 1978

AEW.3 retired from 849 Squadron

DEVELOPMENT

The AEW (Airborne Early Warning) variant of the Gannet had been a long time coming for the Royal Navy and, like the anti-submarine variant before it, an American-built type, the Skyraider AEW.1, had been brought in to fill the capability gap.

DESIGN

While the AS.4 differed very little on the surface from the AS.1, there was no such confusion caused by the Gannet AEW.3. Below the fuselage was mounted a large radome which concealed the aircraft's powerful AN/APS 20. The modification resulted in the aircraft's 3,875 ehp Double Mamba 112 engine having to be moved forward so that the shorter exhausts or jet-pipes could be re-positioned under the leading edge wing root. The undercarriage was lengthened to clear the radome whilst on the ground and the tail surfaces were also modified to a more angular design and increased area, complete with a high-aspect ratio rudder.

Both the second and third cockpits were removed while the interior of the fuselage was re-designed to accommodate a pair of radar operators who accessed the aircraft via a new side entrance door.

The prototype, XJ440, was first flown on August 20, 1958 with a radome fitted but carried no radar equipment, effectively being used as aerodynamic test aircraft. Further trials were carried out aboard HMS *Centaur*, by which time the first production aircraft, XL449, made its maiden flight from Northolt on December 2, 1958.

OPERATIONAL SERVICE

In August 1959, three Gannet AEW.3s were delivered to 700G Squadron, an Intensive Trials Unit based at Culdrose. On February 1, 1960, 700G was absorbed into 849 Squadron, the AEW.3 initially serving with 849A Flight. Even before the 849 Squadron take-over, the unit's AEW.3s had accumulated 1,855 flying hours between them, 1,300 of these between August and December 1959.

849 Squadron, which was divided into four separate flights, was the only unit destined to fly the AEW.3. The flights operated from five Royal Naval carriers over an 18 year period, beginning with 'C' Flight which joined HMS *Hermes* in July 1960.

The main task of the AEW.3 at this time was to simply extend the range of the home aircraft carrier's line-of-sight range protection system. The type was incredibly efficient at the task, many of the AEW.3s in service achieving an average of 30 flying hours per month.

During 1970, four AEW.3s joined HMS *Ark Royal*; the type eventually serving until December 1978 to become the last Fairey aircraft to serve operationally with the Royal Navy.

PRODUCTION

44 AEW.3s were built including the prototype, XJ440, followed by XL449-XL456, XL471-XL482, XL493-XL503, XR431-XR432 and XP197 to XP199, XP224 to XP229 and XR431 to XR433.



TECHNICAL DATA GANNET AEW.3

ENGINE: One 3,875 ehp Double Mamba 112 (A.S.M.D.8)

WING SPAN: 54ft 4in

LENGTH: 44ft

HEIGHT: 16ft 8in

LOADED WEIGHT:

25,000lb

MAX SPEED:

250 mph

CEILING:

25,000ft

RANGE: 700 miles

ENDURANCE:

5 to 6 hr at 130 to

140 mph



The prototype Gannet AEW.3, XJ440, which carried no radar and was only used for handling trials. Compared to earlier variants, the revised fin and rudder are evident. *Via Martyn Chorlton*



XL451, pictured on November 22, 1968, whilst serving with 849A Flight from Brawdy. On July 21, 1968 the aircraft failed to pick up the arrestor cable during a night landing on HMS Hermes off Okinawa. Lt P J Hutchinson was forced to ditch just 50 yards ahead of the carrier but all three crew were quickly rescued. Via Martyn Chorlton



Alongside XL449, Gannet AEW.3, XL451, was used for carrier deck trials on board HMS Victorious in May 1959. Via Martyn Chorlton

A compilation of Fairey construction numbers, their serials (if applied) or registration, aircraft type and variant

Fighter & F.2 Fighter (1) 1, 3702(NB); 2, 3704 (F.2); 3, 3705, Short 827 (12) 4-15, 8550-8561 Campania (10) 16-25, N1000-1009; 26, -, (NB) Sopwith 11/2 Strutter (100) 27-126, A954-1053 N.9/F.127 (1) 127, N9/K-103/G-EAAJ/N-20 III (1) 128, N10/G-EALQ, (P) Hamble Baby (51) 129, 8134, (P); 130-149, N1320-1339; 150-179; N1450-1479 Campania (40) 180-219, N2360-2399 IIIA (50) 220-269, N2850-2899 IIIB (5) 270-274, N2225-2229 Atalanta (1) 275, N118, (NC); 276,

N119 IIIB/C (30) 277-306, N2230-2259 IIIC (30) 307-336, N9230-9259 Titania (1) 337, N129; 338, -, (NB) Pintail I, II, III (3) 339-341, N133-135, (P) N.3 FB (Flying-boat) (0) 342-343, N88-89, (NB)

344-393, N9450-9499; 394-399, ANA.1-6/A10.1-6, (Australia); 400-402, -, (Portugal)

Fawn I/II (3) 403-405, J6907-6909, (P)

part in a Royal Navy/Fleet Air Arm exercise.

Flycatcher (3) 406-408, N163-165,

III (Coastal Defence) (0) 409-410, -, (NB)

IIID (4) 411, -, (Portugal); 412-3, -, 414, -, (NC)

Fawn II (2) 415-6, J6990-1, (P) Flycatcher (3) 417-419, N9611-9613, (D)

Fremantle (1) 420, N173/G-EBLZ IIID (12) 421-432, N9567-9578 Flycatcher (6) 433-438, N9614-

IIID (12) 439-450, N9630-9641 Flycatcher (27) 451-476,

N9655-9680; 477, N9697 Pintail IV (3) 478-480, -, (Japan)

Fawn I (2) 481-2, J7182-3 Fawn II (50) 483, J7184; 484-5, -, (A); 486-532, J7185-7231

Spec 9/23 (Three-engine FB) (0) 533. (NB)

Fawn II (2) 534-5, -, (A)

III (Military Load Carrier) (0) 536, -, (NB)

Fawn II (1) 537, -, (A) Ferret I, II, III (3) 538-540, N190-192, (P)

Spec 26/23 (Single-engined day bomber) (0) 541, (NB) IIID (9) 542,-, (Sweden); 543-549,-,

(Portugal); 550, -, (NC) III (GP Troop Carrier) (0) 551, -,

IIID (20) 552-571, N9730-9749

Firefly I (1) 572, -, (P) Fox I (1) 573, J9515, (P)

IIIF (1) 574, N198, (P)

IIID (2) 575, -; 576 (originally 439) G-EBKE, (British Guiana); 577, -,

Flycatcher (36) 578-613, N9854-9889

IIID (44) 614-655, N9750-9791; 656-7, -, (Mk IA, (NB))

Flycatcher (6) 658-663, N9890-9895

IIID (6) 664-669, -,

Flycatcher (24) 670-693, N9902-9925

IIID (3) 694-5, -, (Amphibian); 696, -, (Sweden)

Flycatcher (40) 697-736, N9926-9965

IIID (45) 737-772, S1000-1035, (Mk II); 773-776, F.1-4, (Netherlands); 777, -, (Seaplane Trainer); 778 (Seaplane Trainer converted to day bomber); 779, -, (Portugal); 780-1, -, (A)

Single/two-seat fighter (0) 782, -, (NB)

Fawn III (12) 783-794, J7768-7779

IIID (3) 795-6, -, (A); 797, -Flycatcher (14) 798-811, \$1060-1073

IIID (35) 812-846, S1074-1108, (Mk III)

Fox I (18) 847-864, J7941-7958 Fawn III (8) 865-872, J7978-7985

Flycatcher II (1) 873, N216

Felixstowe F.2A (Conversion) (1) 874, -, (Portugal)

Fox I (5) 875-879, J8423-8427 IIIF (11) 880-889, S1139-1148; 890, N225, (P)

Felixstowe F.2A (Conversion) (1)

891, -, (Portugal)

IIIF (60) 892-906, S1168-1182, (Mk I); 907-931, S1183-1207, (Mk I); 932-951; S1208-1227, (Mk II)

Fox I/IA (4) 952-955, J9025-9028 IIIF (141) 956-967, S1250-1261, (Mk II); 968, S1262, (Mk II, Irish Free State); 969-975, J9053-9059, (Mk IVC/M); 976-979, 23-26, (Mk I, Chile); 980-997, J9060-9077, (Mk IVC/M); 998-1020, 19132-9154, (Mk IVC/M); 1021-1040, J9155-9174, (Mk IVM); 1041-2, -, (Mk IVM, (D)); 1043-1096, S1303-1356, (Mk

Flycatcher (25) 1097-1121, S1273-1297 IIIF (7) 1122-1127, AP-1-6, (Mk

KEY

IIID (59)

First number in italics from 1-9516 is the aircraft's unique Fairey construction number. Number in brackets after the type name is the amount built. Serial or registration runs from 3702-XR433



Aeroplane

FAIREY NUMBERS

Fantome (1) 3451, L7045, (AF)

IIIM, Argentine); 1128, -, (NB); 1129, G-AABY/VH-UTT, (Mk IIIM) Firefly II (1) 1130, G-ABCN, (P) L.R.M.1 (1) 1131, J9479 Fleetwing (1) 1132, N235 IIIF (4) 1133-4, NZ631-2, (Mk IIIM, New Zealand); 1135-6, -, (Mk IIIM, Irish Free State) Firefly IIIM (1) 1137, S1592/G-ABFH Fox II (1) 1138, J9834/G-ABFG, (P) IIIF (132) 1139-1183, J9637-9681, (Mk IVM/A); 1184-1231, J9784-9831, (Mk IVM/A); 1232-1270, S1370-1408, (Mk IIIM) Hendon (1) 1271, K1695, (P) IIIF (1) 1272, G-AASK, (Mk IIIM); 1273, -, (Mk IIIM, (NB)) Flycatcher (11) 1274-1284, 51409-1418 IIIF (111) 1285-1294, S1454-1463, (Mk IIIM); 1295-1301, K1115-1121, (Mk IIIM/A); 1302-1314, K1158-1170, (Mk IIIM/A); 1315, G-AATT, (Mk IIIM); 1316, S1325, (Mk IIIM/Seal, (P)); 1317-1395, S1474-1552, (Mk IIIB) Gordon I (1) 1396, K1697, (P) IIIF/Gordon I (51) 1397-1427, K1698-1728, (Mk IVB); 1428-1447, K1729-1748 IIIF/Gordon I (41) 1448-1477, K1749-1778, (Mk IVB/Gordon I); 1478, -, (Mk IVM/A, Russia); 1479-1488, -, (Mk IIIB, Greece) Firefly (25) 1489-1513, Y-I to Y-25, (Belgium) IIIF (1) 1514, -, (Mk IIIB (Chile)) Gordon (1) 1515, -, (China) IIIF (66) 1516-1581, S1779-1844, (Mk IIIB); 1582-3, -, (Mk IVB (NB)) Gordon I (47) 1584-1630, K2603-2649; 1631-1638, -, (NB) Fox II (12) 1639-1650, 0-1 to 0-12, (Belgium)

Firefly II (20) 1651-1670, A.F.5001-5020/Y-26 to Y-45, (AF) L.R.M. II (1) 1671, K1991 IIIF (21) 1672-1678, S1845-1851, (Mk IIIB); 1679-1692, S1852-1865, (Mk IIIB) Firefly II (31) 1693-1722; A.F.5021-5051, Y-46 to Y-75, (AF) Fox II (31) 1723-1752; A.F.6001-6030, (AF); 1753, A.F.6031, (Fox VI (P)(AF)) S.9/30 (1) 1754, S1706 Gordon I (87) 1755-1802, K2683-2730; 1803-1812, -, (Brazil); 1813-1826, K2731-2744; 1827-1836, -, (Brazil); 1837-1841, K2745-2749 Fox III/IV (1) 1842, G-ABYY Seal (32) 1843-1853, K3477-3487; 1854-1874, K3514-3534 TSRI(1) 1875, -Firefly II (1) 1876, Among A.F.5022-5030, (Russia) Gordon I (20) 1877-1896. K2750-2769 Seal (16) 1897-1907, K3535-3545; 1908-9, K3575-6; 1910, K3577, (Gordon II (P)); 1911-2, K3578-9 Fox III (13) 1913-1925, A.F.6033-6045, (AF) G.4/31 (2) 1926, K3905; 1927, -, (NC) Firefly II (6) 1928-1933, A.F.5052-5057/Y-76 to Y-81, (AF) Seal (6) 1934-1939, -, (Peru) Gordon II (24) 1940-1963, K3986-4009 Fox Trainer & Fox II Floatplane

(7) 1964, A.F.6032/C-ACKH, (AF);

Seal (25) 1971-1995, K4201-4225

Firefly II (6) 2032-2037, A.F.5058-

1965-1970, -, (Peru)

A.F.6046-6081, (AF)

Fox III (36) 1996-2031,

5063/Y-82 to Y-87, (AF)

Battle (1) 2121, K4303, (P) Seafox (2) 2122-3, K4304-5, (P) Hendon (14) 2124-2137, K5085-5098; 2138-2141, K5768-5771, (NB) Swordfish I (89) 2142-2144, K5660-5662, (D); 2145-2230, K5926-6011 P.4/34 (1) 2231, K5099, (P) Fox VI (32) 2232-2241, -, (AF); 2242-2245, -, (AF); 2246-7, 871-2/ HB-HAF, -HAK (AF) (Switzerland); 2248-2263, -, (AF) Feroce (1) 2264-5, -, (AF) (Russia) P.4/34/Fulmar (1) 2266, K7555, Seafox (49) 2267-2315, K8569-8617 Battle (200) 2316-2470, K7558-7712; 2471-2515, K9176-9220 Fox IIIC (12) 2516-2527, -, (AF) Swordfish I (281) 2528-2631, K8346-8449; 2632-2658, K8860-8886; 2659-2808, L2717-2866 Battle (465) 2809-3074, K9221-9486; 3075-3121, N2020-2066; 3122-3171, N2082-2131; 3172-3215, N2147-2190; 3216-3257, N2211-2252; 3258-3273, F.10-25, (Belgium) Albacore (100) 3274-3373, L7074-7173 Swordfish I (62) 3374-3403, L7632-7661; 3404-3435, L7670-7701

Seafox (15) 3436-3450, L4519-4533

TSR II (1) 2038, K4190

Fox VI (54) 2039-2092, -, (AF)

(Latvia); 2116-7, -, (Chile)

2120, A.F.6134, (AF)

Seal (25) 2093-2110, K4779-4796;

2111, -, (Argentina); 2112-2115, -,

Fantome 2118, F.6/G-ADIF, (P)

Fox VII (2) 2119, A.F.6142, (AF);

Battle (6) 3452-3457, N2253-2258 Swordfish I (60) 3458-3487, L9714-9743; 3488-3517, L9756-9785 Albacore (189) 3518-3566, N4152-4200; 3567-3616, N4219-4268; 3617-3666, N4281-4330; 3667-3706, N4347-4386 Fulmar I (250) 3707-3746, N1854 (G-AIBE)-1893: 3747-3796. N1910-1959; 3797-3833, N1980-2016; 3834-3883, N3994-4043; 3884-3924, N4060-4100; 3925-3956, N4116-4147 Albacore (11) 3957-3961, N4387-4391; 3962-3967, N4420-4425 Battle (250) 3968-3997, P6480-6509; 3998-4047, P6523-6572; 4048-4067, P6596-6615; 4068-4117, P2155-2204; 4118-4163, P2223-2278; 4164-4200, P2300-2336; 4201-4217, P2353-2369 Swordfish I (200) 4218-4266, P3991-4039; 4267-4301, P4061-4095; 4302-4348, P4123-4169; 4349-4390, P4191-4232; 4391-4417, P4253-4279 Battle (152) 4418-4442, P5228-5252; 4443-4467, P5270-5284; 4468-9, P1767, 1770, (P); 4470-4499; P6616-6645, (T); 4500-4529, P6663-6692, (T); 4530-4549, P6718-6737, (T); 4550-4569, P6750-6769, (T) Barracuda I (25) 4570-4594, P9642-9666 Barracuda II (225) 4595-4619, P9667-9691; 4620-4659, P9709-9748;





4660-4709, P9787-9836; 4710-4754, P9847-9891; 4755-4789, P9909-9943; 4790-4819, P9957-9986 Battle T (100) 4820-4849, R7356-7385; 4850-4899, R7399-7448; 4900-4919, R7461-7480

Albacore (100) 4920-4964, T9131-9175; 4965-4989, T9191-9215; 4990-5019, T9231-9260

Fulmar II (200) 5020-5069, X8525-8574; 5070-5114, X8611-8655; 5115-5149, X8680-8714; 5150-5199, X8729-8778; 5200-5219, X8798-8817

Albacore (250) 5220-5264, X8940-8984; 5265-5314, X9010-9059; 5315-5359, X9073-9117; 5360-5409, X9137-9186; 5410-5429, X9214-9233; 5430-5469, X9251-9290

Firefly I (198) 5470-5473, Z1826-1829, (P); 5474-5489, Z1830-1845; 5490-5539, Z1865-1914; 5540-5584, Z1942-1986; 5585-5632, Z2011-2058; 5633-5638,

TW677-682 (three NB); 5639-5663, Z2096-2120; 5664-5667, TW683-686, (NB)

Firefly 4 (2) 5668-9, TW687-688 Albacore (150) 5670-5704, BF584-618; 5705-5754, BF631-680; 5755-5799, BF695-739; 5800-5819, BF758-777 Fulmar II (150) 5820-5841, BP775-796; 5842-5869, BP812-

839; 5870-5919, DR633-682; 5920-5969, DR700-749 Barracuda II (50) 5970-5988,

DT813-831; 5989-6009, DT845-865; 6010-6019, DT878-887 Firefly 4 (5) 6020-6024, TW689-693

Firefly I (19) 6025-6043, DT931-949

Firefly 4 (12) 6044-6049, TW694-699; 6050-6055, TW715-

Firefly I (18) 6056-6073, DT974-991

Firefly 4 (12) 6074-6085, TW721-732 Firefly I (18) 6086-6103,

DV117-134 Firefly 4 (12) 6104-6115, TW733-744

Firefly I (203) 6116-6119,

DV147-150; 6120-6165, PP391-436; 6166, PP437/F-3, (RNNAS); 6167-8, PP456-457; 6169, PP458/F-1, (RNNAS); 6170-6182, PP459-471; 6183, PP472/F-2, (RNNAS); 6184, PP473; 6185, PP474/F-10, (RNNAS); 6186-6194, PP475-483; 6195, PP484/F-6, (RNNAS); 6196, PP485; 6197-8, PP486-7/F-7, F-5, (RNNAS); 6199, PP488; 6200, PP489/F-4 (RNNAS); 6201-2, PP490-491; 6203, PP492/F-8, (RNNAS); 6204-5, PP493-4/F-9, F-14, (RNNAS); 6206-7, PP495-6; 6208, PP497/F-12, (RNNAS); 6209-6211, PP523-525;6213-6214, PP526-528/F-15, -11,-13 (RNNAS); 6215-6253, PP529-567;6254-6256, PP580-582; 6257, PP583/F-16, (RNNAS); 6258-6263, PP584-589; 6264-6267, PP590-593/F-17 to -20, (RNNAS); 6268-6274, PP594-600; 6275-6277, PP601-603/F-21 to -23, (RNNAS): 6278-9, PP604-5; 6280-1, PP606-7/F-24 and 25, (RNNAS);

6282-6287, PP608-613; 6288-6290, PP614-616/F-26 to -28, (RNNAS); 6291-2, PP617-8; 6293,

PP619/F-29, (RNNAS); 6294-6296,

PP620-622; 6297, PP623/F-30, (RNNAS); 6298-6319, PP639-660 Barracuda II (428) 6320-6362, LS464-506; 6363-6400, LS519-556; 6401-6428, LS568-595; 6429-6435, LS596-602; 6436-6474, LS615-653; 6475-6520, LS668-713; 6521-6558, LS726-763; 6559-6601, LS778-820; 6602-6647, LS833-878; 6648-6693,

LS891-936; 6694-6719, LS949-974 Barracuda III (460) 6720-6761, PM682-723; 6762-6804, PM738-780; 6805-6847, PM796-838; 6848-6893, PM852-897; 6894-6939, PM913-958 (Mk V (P)); 6940-6969, PM970-999; 6970-7019, PN115-164; 7020-7061, RK328-369; 7062-7108, RK382-428; 7109-7153, RK441-485; 7154-7179, RK498-523

Barracuda V (30) 7180-7192, RK530-542; 7193-7209, RK558-574; 7210-7369, Within RK575-784, (NB)

Firefly I (59) 7370-7411, MB378-419; 7412-7428, MB433-449

Firefly 4 (10) 7429-7438, TW745-754 Firefly I (430) 7439-7458, MB460-479; 7459-7503, MB492-

FAIREY NUMBERS



536; 7504-7548, MB549-593; 7549-7585, MB613-649; 7586-7627, MB662-703; 7628-7669, MB717-758; 7670-7718; DK414-462, (General Aircraft); 7719-7756; DK476-513, (General Aircraft); 7757-7801, DK526-570 (General Aircraft); 7802-7833, DK588-619, (NB); 7834-7869, DK632-667, (NB) Spearfish (5) 7870, RA356, (P); 7871, RA360, (P); 7872, RA363, (P); 7873, RN241, (P); 7874, TJ175, (NC); 7875-6, TJ179, TJ184, (NB); 7877-7976, -, (NB)

Firefly FR.4 (67) 7977-8019, VG957-999; 8020-8043, VH121-144; 8044-8176, Within VH145-361, (N.F.4 (NB))

Spearfish (0) 8177-8197, TS915-935, (NB); 8198-8225, TS963-990, (NB); 8226, TT110, (NB) Firefly FR.4 (40) 8227-8266, 11-31 to-70/16-31 to -70, (RNNAS)

Spec N.16/45 (Strike Aircraft)

8267-8269, -, (NB) Gannet (2) 8270-1, VR546, VR557, (P)

Firefly FR.4 (70) 8272-8291, VT362-381;8292-8341, VT392-441 Firefly FR.4/5 (47) 8342-8388,

VT458-504 Firefly NF.5 (14) 8389-8402, 11-71 to -84/16-71 to -84, (RNNAS)

Firefly FR.4/5 (52) 8403-8428, VX371-396; 8429-8454, VX413-438 Primer (2) 8455-6, G-ALBL, G-ALEW; 8457-8464, -, (NB) Gyrodyne (1) 8465, VX591/G-AIKF F.D.1 (1) 8466, VX350; 8467-8, VX357, VX364, (NB)

Firefly 5 (169) 8469-8498, WB243-272; 8499-8534, WB281-316; 8535-8587, WB330-382; 8588-8637, WB391-440

Firefly 5/6 (6) 8638-8643, WB505-510

Firefly 6 (105) 8644-8651, WB516-523; 8652-8700; WD824-872; 8701-8748, WD878-925 Gannet (1) 8749, WE488, (P)

Firefly 6 (6) 8750-8755, WH627 632

Firefly AS.7 (52) 8756-8784, WJ146-174; 8785-8807, WJ187-

Firefly AS.6 (18) 8808-8825, WJ104-121

Firefly 7 (2) 8826-7, WJ215-6, (D) Firefly AS.7 (45) 8828-8853, WK348-373; 8854-8872 (19),

WM761-779 Firefly AS.7/U.8 (82) 8873-8909,

WM796-832; 8910-8954, WM855-899

Firefly U.8 (4) 8955-8958, WP351-354

Firefly AS.7 (0) 8959-9059, Within WP355-490, (NB); 9060-9084, WV967-991, (NB); 9085-9110, WW103-128, (NB)

Gannet AS.1 (26) 9111-9136, WN339-364

Gannet T.2 (1) 9137, WN365, (P) Gannet AS.1 (142) 9138-9150, WN366-378; 9151-9190, WN390-429; 9191-9210,

WN445-464; 9211-9256, XA319-364; 9257-9279, XA387-409

Gannet AS.4 (24) 9280-9303, XA410-433

Gannet AS.1 (1) 9304, XA434 Gannet AS.4 (1) 9305, XA435 Gannet AS.1 (1) 9306, XA436

Gannet AS.4 (20) 9307-9326, XA454-473

Gannet AS.1 (1) 9327, XD898 Gannet T.2 (23) 9328-9350, XA508-530

Gannet AS.1/4 (16) 9351-9366, XG783-798

Gannet AS.1 (2) 9367-8, XG825-6 Gannet AS.4 (29) 9369-9397, XG827-855

Gannet T.2 (13) 9398-9410, XG869-881

Gannet T.5 (6) 9411-9416, XG882-887

Gannet T.2 (1) 9417, XG888 Gannet T.5 (2) 9418-9, XG889-890

Gyrodyne/Jet Gyrodyne (1) 9420, G-AJJP/XJ389 F.D.2 (2) 9421-2, WG774, WG777

Ultra-light Helicopter 9423, XJ924; 9424, XJ928/G-AOUJ; 9425, XJ930; 9426, XJ936/G-AOUK; 9427, -, 9428, G-APJJ

Rotodyne (1) 9429, XE521, ('V'); 9430, XH249, ('Z') (NB)

Gannet AEW.3 (41) 9431, XJ440, (P); 9432-9471, XL449-456, XL471-482, 493-503, XP197-199, 224-229

Westland Scout (40) 9472-9483, XP846-857; 9484-9511, XP883-910 Gannet AS.4 (2) 9512-3, -, Gannet AEW.3 (3) 9514-9516, XR431-433

One of the most unlikely heroes of the Second World War was the Fairey Swordfish. Aeroplane

FAIREY















FAIREY — THE SUBCONTRACT WORK



Stockport-built Halifax GR.V, LK688, of 202 Squadron, one of 246 built by Fairey, between October 1942 and January 1944. Via Martyn Chorlton



FROM ITS INCEPTION AND THROUGH TO ITS MERGER WITH WESTLAND AVIATION, FAIREY PRODUCED AIRCRAFT FOR OTHER AIRCRAFT COMPANIES INCLUDING SHORT BROTHERS, BRISTOL, HANDLEY PAGE AND DE HAVILLAND

Short Type 827

8550-8561 (c/n 4-15) Total: 12

Sopwith 11/2 Strutter

A954-1053 (c/n 27-126) Total: 100

Felixstowe F.2A (Conversion for Portugal)

C/n 874 & 891 Total: 2

Bristol Beaufighter IC & IF

337 Beaufighter IFs and ICs were delivered between February 1941 and April 1942 by Fairey, Heaton Chapel to Contract No. B41906/39. T4623-4649 were built as IFs. T4623-4647, T4648-4670, 4700-4734, 4751-4800, 4823-4836, 4862-4899, 4915-4947, 4970-5007, 5027-5055 and 5070-5099

Bristol Beaufighter VI

163 Beaufighter VIs were delivered between April 1942 and May 1943 by Fairey, Heaton Chapel to Contract No. B41906/39. T5100-5114, 5130-5175, 5195-5220, 5250-5299 and 5315-5352

Handley Page Halifax V

150 Halifax Vs were delivered between October 1942 and August 1943 by Fairey Aviation, Stockport. DJ980-999, DK114-151, 165-207 and 223-271

Handley Page Halifax V

96 Halifax Vs were delivered between August 1943 and January 1944 by Fairey Aviation, Stockport.

LK626-667, 680-711 and 725-746

Handley Page Halifax III

104 Halifax IIIs were delivered between January and April 1944 by Fairey Aviation, Stockport. LK747-766, 779-812, 826-850 and 863-887

Handley Page Halifax B.III

180 Halifax IIIs were delivered between April and November 1944 by Fairey Aviation, Ringway. NA492-531, 543-587, 599-644 and 656-704

Handley Page Halifax III

41 Halifax IIIs were delivered between October 1944 and February 1945 by Fairey Aviation, Stockport. PN167-207

Handley Page Halifax A.VII/B.VII

90 Halifax VIIs were delivered between February and October 1945 by Fairey Aviation, Stockport. PN208, PN223-267, 285-327 and 343*

*PN345-362, RS227-497, TM944-TN247, totalling 368 Halifaxes, were all cancelled

De Havilland Vampire FB.9

51 Vampire FB.9s were built by Fairey Aviation, Stockport, to Contract No. 6/Acft/6422. WR205-215 and 230-269

Westland Scout AH.1

40 Scout AH.1s produced for the Army Air Corps to Contract KF/2Q/06. XP846-857 and XP883-910



The final batch of Fairey-built Halifaxes were Mk VIIs, all built at Stockport and delivered between February and October 1945. Many were converted to transport C.VII standard, including PN261, seen here serving with 298 Squadron. Via Martyn Chorlton







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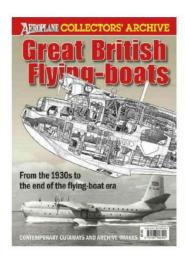
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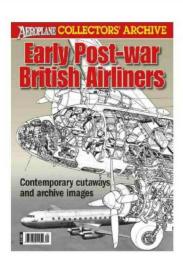


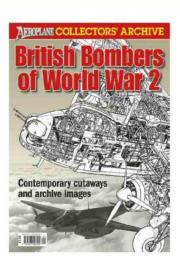


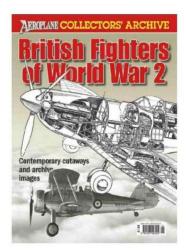


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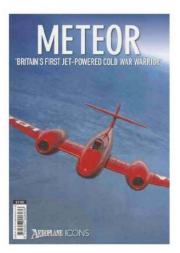
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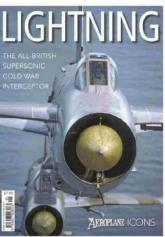
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